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# THE APPLICATION OF BIOCYBERNETIC TECHNIQUES TO ENHANCE PILOT PERFORMANCE DURING TACTICAL MISSIONS

1 OCTOBER 1979

MDC-E2046

FINAL REPORT

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BIOCYBERNETICS AND PILOT PERFORMANCE  1 OCTOBER 1979 MDC E2046		
I OCTOBER 197	TABLE OF CONTENTS	MIDC E2040
LIST OF FIGU	URES	iv
LIST OF SYME	BOLS AND ABBREVIATIONS	x
PREFACE		xvi
SECTION 1.0	ABSTRACT	1
SECTION 2.0	INTRODUCTION	4
SECTION 3.0	AN OVERVIEW OF MISSION TYPES AND PROJECTED TECHNOLOGY ADVANCES FOR TACTICAL AIRCRAFT	8
	3.1 Mission Types	12
	3.2 Technology Advances	20
	3.2.1 Weapons	20
	3.2.2 Avionics	25
	3.2.2.1 Sensors	25
	3.2.2.2 Electronic Displays	27
	3.2.2.3 Multifunction Control Unit	33
	3.2.2.4 Voice Actuated Controls	34
	3.2.2.5 Joint Tactical Information Distribution System	35
	3.2.2.6 Airborne Warning And Control System	36
	3.2.2.7 Global Positioning Satellite	36
SECTION 4.0	MISSION REQUIREMENTS	37
SECTION 5.0	INFORMATION NEEDS	59
SECTION 6.0	PILOT PERFORMANCE AND BIOCYBERNETIC APPLICATIONS	72
	6.1 Time Line Analysis of Pilot Tasks	72
	6.1.1 Launch	79

1 OCTOBER 197	BIOCYBERNETICS AND PILOT PERFORMANCE 9		MDC E2046	
			TABLE OF CONTENTS (Continued)	20 220 10
		6.1.2	Climb	81
			Rendezvous	83
		6.1.4	Ingress	85
			Medium Range Intercept	87
			Surface-To-Air Missile Avoidance	89
			Air Combat Maneuvering	91
			Air-To-Ground Strike	93
		6.1.9	Egress	95
		6.1.10	In-Flight Refuel	97
		6.1.11	Marshal	99
		6.1.12	Prelanding	101
		6.1.13	Landing	103
	6.2		ime Analysis And Interpretation logical Signals	105
		6.2.1	Brain Electrical Activity	109
		6.2.2	Peripheral Activity	120
			6.2.2.1 Psychophysiological Responses	121
			6.2.2.2 Ocular Activity	122
	6.3	Biocybe	ernetic Applications	126
SECTION 7.0	CONC	LUSIONS	AND RECOMMENDATIONS	144
SECTION 8.0	REFE	RENCES		147
			LIST OF PAGES	

Title Page ii through xvi l through 150

#### **1 OCTOBER 1979**

#### **MDC E2046**

# LIST OF FIGURES

FIGURE NUMBER	TITLE
3.1	Integrated Crew Station Concept For A 1990 Tactical Aircraft.
3.2	Descriptions Of The Instruments Depicted In Figure 3.1.
3.3	Pictorial Representation Of The Three TACAIR Missions.
3.4	General Mission Requirements For Pre-Flight, In-Flight, And Post-Flight Phases.
3.5	Mission Profile For Close Air Support.  Accession For
3.6	Mission Profile For Air Interdiction.   NTIS GRADE
3.7	Mission Profile For Counter Air.  DIIC TAB  Unanimod  Instant
3.8	Guidance Systems And Their Characteristics.
3.9	Types Of Air-To-Air Missiles.  By Distribution/
3.10	Types of Air-To-Surface Ballistic Weapons Availability Codes
3.11	Types of Air-To-Surface Missiles.  Dist  Special
3.12	Anticipated Avionics Advances.
3.13	Sensor Characteristics.
3.14	Information Formats And Display Coding Schemes.
3.15	Comparison Of Display Technologies with Respect To Information Formats And Coding Schemes.
3.16	Evaluation Parameters And Corresponding Performances Of Selected Display Devices.
4.1	Pre-Flight Mission Requirements For Close Air Support, Air Interdiction, and Counter Air.
4.2	Pre-Flight Mission Requirements (Continued).
4.3	Pre-Flight Mission Requirements (Continued).

FIGURE NUMBER	TITLE
4.4	Pre-Flight Mission Requirements (Concluded).
4.5	In-Flight Mission Requirements For Close Air Support, Air Interdiction, and Counter Air.
4.6	In-Flight Mission Requirements (Continued).
4.7	In-Flight Mission Requirements (Continued).
4.8	In-Flight Mission Requirements (Continued).
4.9	In-Flight Mission Requirements (Continued).
4.10	In-Flight Mission Requirements (Continued).
4.11	In-Flight Mission Requirements (Continued).
4.12	In-Flight Mission Requirements (Continued).
4.13	In-Flight Mission Requirements (Continued).
4.14	In-Flight Mission Requirements (Continued).
4.15	In-Flight Mission Requirements (Continued).
4.16	In-Flight Mission Requirements (Continued).
4.17	In-Flight Mission Requirements (Continued).
4.18	In-Flight Mission Requirements (Continued).
4.19	In-Flight Mission Requirements (Concluded).
4.20	Post-Flight Mission Requirements For Close Air Support, Air Interdiction, and Counter Air.
4.21	Post-Flight Mission Requirements (Concluded).
5.1	Information Requirements For Close Air Support.
5.2	Information Requirements For Close Air Support (Continued).

FIGURE NUMBER	TITLE
5.3	Information Requirements For Close Air Support (Continued).
5.4	Information Requirements For Close Air Support (Concluded).
5.5	Information Requirements For Air Interdiction.
5.6	Information Requirements For Air Interdiction (Continued).
5.7	Information Requirements For Air Interdiction (Continued).
5.8	Information Requirements For Air Interdiction (Concluded).
5.9	Information Requirements For Counter Air.
5.10	<pre>Information Requirements For Counter Air (Con- tinued).</pre>
5.11	Information Requirements For Counter Air (Continued).
5.12	Information Requirements For Counter Air (Concluded).
6.1	Seyments Of A Carrier Launched Escort Mission.
6.2	Representative Profile For A Carrier Launched Escort Mission.
6.3	Schematic Configuration Of F/A 18 Crew Station.
6.4	Anticipated Task Loadings During Segments Of A Carrier Launched Escort Mission.
6.5	Dynamic Task Flows For Launch.
6.6	Dynamic Task Flows For Climb

FIGURE NUMBER	TITLE
6.7	Dynamic Task Flows For Rendezvous.
6.8	Dynamic Task Flows For Ingress.
6.9	Dynamic Task Flows For Medium Range Intercept.
6.10	Dynamic Task Flows For SAM Avoidance.
6.11	Dynamic Task Flows For Air Combat Maneuvering.
6.12	Dynamic Task Flows For Air-To-Ground Strike.
6.13	Dynamic Task Flows For Egress.
6.14	Dynamic Task Flows For In-Flight Refueling.
6.15	Dynamic Task Flows For Marshal.
6.16	Dynamic Task Flows For Prelanding.
6.17	Dynamic Task Flows For Landing.
6.18	Biological Signals Considered As Inputs For Biocyber- netic Applications.
6.19	Electroencephalogram Of A Normal Human Adult.
6.20	Vertex Event-Related Potential Elicited By A Matching Letter Presentation During An Item Recognition Task.
6.21	Lateral And Superior Views Of International Electrode Placement System.
6.22	Typical CNV Waveform Recorded From Vertex Electrode Placement.
6.23	(A) The Displayed Scene And (B) The Eye-Scan Pattern And Vertex EEG Of One Subject Before And After Detection Of Vehicle.

FICURE NUMBER	TITLE
6.24	Examples Of Vertex Readiness Potentials Associated With Finger Presses During A Reaction Time Task and During Voluntary Movements.
6.25	Relationship Between Particular Biological Signals And Either The Determination Of Pilot Status Or Control Functions.
6.26	Biocybernetic Applications As A Function Of Pilot Tasks During Launch.
6.27	Biocybernetic Applications As A Function Of Pilot Tasks During Climb.
6.28	Biocybernetic Applications As A Function Of Pilot Tasks During Rendezvous.
6.29	Biocybernetic Applications As A Function Of Pilot Tasks During Ingress.
6.30	Biocybernetic Applications As A Function Of Pilot Tasks During MRI.
6.31	Biocybernetic Applications As A Function Of Pilot Tasks During SAM Avoidance.
6.32	Biocybernetic Applications As A Function Of Pilot Tasks During ACM.
6.33	Biocybernetic Applications As A Function Of Pilot Tasks During A/G Strike.
6.34	Biocybernetic Applications As A Function Of Pilot Tasks

#### **1 OCTOBER 1979**

**MDC E2046** 

FIGURE NUMBER	TITLE
6.35	Biocybernetic Applications As A Function Of Pilot Tasks During In-Flight Refueling.
6.36	Biocybernetic Applications As A Function Of Pilot Tasks During Marshal.
6.37	Biocybernetic Applications As A Function Of Pilot Tasks During Prelanding.
6.38	Biocybernetic Applications As A Function Of Pilot Tasks During Landing.

#### **1 OCTOBER 1979**

#### LIST OF SYMBOLS AND ABBREVIATIONS

**MDC E2046** 

AAA Anti-Aircraft Artillery

A/A Air-to-Air

AAR Air-to-Air Refueling

A/C Aircraft

ACM Air Combat Maneuvering

ADC Air Data Computer

ADF Automatic Direction Finder

ADI Attitude Direction Indicator

AFCS Automatic Flight Control System

A/G Air-to-Ground

AGL Altitude above Ground Level

AGM Air-to-Ground Missile

AGR Air-to-Ground Ranging

AI Air Interdiction

AIM Air Intercept Missile

ALCM Air Launched Cruise Missile

AOA Angle-of-Attack

APC Armored Personnel Carrier

AWACS Airborne Warning And Control System

AZ Azimuth

BCIU Bus Control Interface Unit

BDM Bomber Defense Missile

Bingo Level of fuel necessary to return to airfield

BITE Built In Test Equipment

#### **BIOCYBERNETICS AND PILOT PERFORMANCE**

**MDC E2046** 

LIST OF SYMBOLS AND ABEREVIATIONS (Cont'd)

BLU	Bomb Live Unit
$c^3$	Command,Control and Communication
CA	Counter Air
CAS	Close Air Support
СВИ	Cluster Bomb Unit
C/D	Controls and Displays
CET	Central European Theatre
CNI	Communication, Navigation, Identification
CNV	Contingent Negative Variation
COM	Communication
CRT	Cathode Ray Tube
DAIS	Digital Avionics Information System
DARPA	Defense Advanced Research Projects Agency
DBS	Doppler Beam Sharpened
DME	Distance Measuring Equipment
DP	Detection Potential
ECG	Electrocardiogram
ECM	Electronic Counterneasure
EEG	Electroencephalogram
EL	Elevation
EL	Electroluminescence
EMD	Electromechanical Device
EMG	Electromyogram
EO	Electro-optical

LIST OF SYMBOLS AND ABEREVIATIONS (Cont'd)

EOG Electro-oculogram

EPI Elements Per Inch

ERP Event-Related Potential

F/A Fighter/Attack

FAC Forward Air Controller

FEBA Forward Edge of the Battle Area

FLIR Forward Looking Infrared

FLR Forward Looking Radar

FOV Field-Of-View

ftL Foot Lambert

g Gravity

GBU Guided Bomb Unit

GMTI Ground Moving Target Indicator

GPS Global Positioning Satellite

HACQ Horizontal Acquisition

HMD Helmet-Mounted Display

HOTAS Hands-On-Throttle-And-Stick

HR Heart Rate

HSD Horizontal Situation Display

HUD Head-Up Display

Hz Hertz

IAS Indicated Airspeed

IFA In-Flight Alignment

IFF Identify - Friend or Foe

IFR Instrument Flight Rules

**1 OCTOBER 1979** 

**MDC E2046** 

LIST OF SYMBOLS AND ABBREVIATIONS (Cont'd)

IIR Imaging Infrared

ILS Instrument Landing System

IMFK Integrated Multifunction Keyboard

INS Inertial Navigation System

IR Infrared

JTIDS Joint Tactical Information Distribution System

LC Liquid Crystal

LD Laser Display

LED Light Emitting Diode

LLLTV Low-Light-Level TV

LOAL Lock-On After Launch

LOBL Lock-On Before Launch

LRU Line Replaceable Unit

LSI Large Scale Integration

MFCU Multifunction Control Unit

MFD Multifunction Display

MLS Microwave Landing System

MMD Master Monitor Display

MMD Multimode Display

MOS Metal Oxide Semiconductor

MPD Multipurpose Display

MRI Medium Range Intercept

MSI Medium Scale Integration

NAV Navigation

**1 OCTOBER 1979** 

LIST OF SYMBOLS AND ABBREVIATIONS (Cont'd)

**MDC E2046** 

nm Nautical mile

 $P_{300}$  A late component of an event-related potential

PAL Permissive Action Link

PLD Plasma Device

POL Petroleum, Oil, Lubricant

R Range

RBGM Real Beam Ground Map

RHAWS Radar Homing And Warning System

ROE Rules of Engagement

RP Readiness Potential

RTU Remote Terminal Unit

S/A Surface-to-Air

SAM Surface-to-Air Missile

SAR Synthetic Aperture Radar

SIF Select Identification Feature

SMS Store Management System

TA Terrain Avoidance

TACAIR Tactical Air Power

TACAN Tactical Air Navigation

TACS Tactical Air Control System

TDMA Time Division Multiple Access

TERCOM Terrain Contour Matching

TF Terrain Following

TFR Terrain Following Radar

#### **1 OCTOBER 1979**

**MDC E2046** 

LIST OF SYMBOLS AND ABBREVIATIONS (Cont'd)

TV Television

UFC Up-Front Control

UHF Ultra High Frequency

VACQ Vertical Acquisition

VAS Voice Actuated System

VEL Velocity

Vertex Midline Site For Recording Brain Electrical Activity,

Denoted C<sub>z</sub>

VSD Vertical Situation Display

**1 OCTOBER 1979** 

**MDC E2046** 

#### **PREFACE**

This report presents the results of a study entitled "The Application of Biocybernetic Techniques to Enhance Pilot Performance During Tactical Missions." The research was conducted by the McDonnell Douglas Astronautics Company-St. Louis Division for the Defense Advanced Research Projects Agency (DARPA) under contract MDA-903-78-C-0181. The preparation of a second document - "Proceedings of the DARPA Conference on Biocybernetic Applications for Military Systems," Chicago, April 1978 - also was supported by this contract.

We acknowledge the assistance of Dr. Craig I. Fields, Senior Program Manager, Cybernetics Technology Office of the Defense Advanced Research Projects Agency.

#### 1.0 ABSTRACT

This report describes a rather novel means of enhancing man's performance in highly complex, crew station environments. Specifically, we have related the benefits of on-line evaluation of physiological data to projected mission requirements for a 1990 tactical aircraft.

The salient role that tactical air power must continue to play in the structure of U.S. defense forces has engendered a sophisticated technological approach to weapon system development. Therefore, we begin with an overview of the components of a "high technology" weapon system - real-time command and control, advanced crew station and avionics design, effective defense suppression, sensor aided target acquisition, and precision-guided ordnance. Although a reliance upon advanced technology and a trend toward greater automation of aircraft functions are clearly evident, the importance of the human element should not be underestimated. This is especially true if the system is to retain the capacity to anticipate and respond to unpredictable threats. Herein lies the present dilemma. Man-in-the-loop assures that tactical aircraft will have an inherent flexibility. However, if man is unable to perform increasingly complex tasks both rapidly and accurately under all combat situations, he may severely limit, and perhaps even undermine, the inventive technology of the system he controls.

It may be possible to solve this problem by taking advantage of the same improvements in digital computation and signal processing that currently influence hardware development. That is, we may enhance the pilot's effectiveness if we monitor momentary fluctuations in attentiveness and in his

**1 OCTOBER 1979** 

**MDC E2046** 

ability to process information and make appropriate decisions. The report summarizes research which has demonstrated that these mental activities are manifest in distinct electrophysiological signals, and that such signals, recorded noninvasively and unobtrusively, can be analyzed and interpreted in real-time. On this basis, we suggest that the central computer onboard the aircraft may be able to determine:

- o when the pilot is inattentive,
- o when visual or auditory information has not been processed,
- o when the pilot is task-loaded to the extent that he is unable to accept additional duties,
- o when the pilot lacks confidence in a decision he has made.

For a variety of mission segments we then outline the courses of action which can be taken to unburden or assist the pilot if biological signal processing has forewarned an imminent deterioration in his capacity to perform. The actions include, among others:

- o redistributing task responsibilities,
- o reducing the complexity of or "decluttering" information displays, especially the HUD,
- o cueing the pilot to attend to critical flight, weapons, and target data.
- o displaying adaptive decision aids which present weighted recommendations for mission-related strategies, particularly with respect to fire control functions,
- o furnishing remedial "checklists,"
- o optimizing the physical characteristics (e.g., contrast, focus, etc.) of imagery and symbolic presentations.

#### **1 OCTOBER 1979**

**MDC E2046** 

The recording and analysis of electrophysiological data also may permit a direct coupling of the pilot with aircraft subsystems from a control standpoint. At issue is whether it will be possible to interpret bioelectric patterns related to different thought commands, whereby the pilot can "think" to activate control surfaces.

We are aware that a great deal more must be accomplished (in computer technology, software development, and the design of physiological monitoring equipment) before it is both feasible and practical to apply biocybernetic techniques in dynamic, operational environments. Nonetheless, we have attempted to clarify important basic research issues and to recommend reasonable priorities for future investigations.

#### 2.0 INTRODUCTION

Generally stated, our purpose has been to assess the impact of applying biocybernetic techniques to improve human performance and thus enhance system effectiveness. We have made several assumptions in undertaking this assessement.

- o The human operator is and will remain an integral component of evolving computer-based systems.
- o These advanced systems will be adaptive, that is, the distribution of responsibilities between the operator and the system will be modified as circumstances change.

The computer-based system with which we are concerned is a generic tactical aircraft, and the human operator in this instance is more commonly referred to as a pilot. With respect to the data requirements for implementing adaptive procedures, we find that digital avionics programs within the Air Force and Navy are providing the means for a reliable and efficient flow of information about the current status of aircraft subsystems. Via external sensing, the pilot also will be apprised of the changing posture of the mission.

<sup>&</sup>lt;sup>1</sup> The term denotes a real-time communication link between a human operator and the system he controls, that is based upon the physiological activity which is recorded as the operator performs assigned tasks.

**1 OCTOBER 1979** 

MDC F2046

Rapidly occurring developments in digital computation, data-busing, electronic circuitry, display generation, and input/output technology have created a trend toward greater automation. This will reduce the extent to which the pilot participates in housekeeping functions (i.e., navigation, subsystems monitoring, communications, and aircraft control), thereby allowing him to attend more closely to mission-related activities (i.e., detection, location, identification, decision, execution, and assessment). The pilot will, of course, play a direct role in any housekeeping function if alerted of a subsystem failure.

While the communication link from subsystem to computer is impressive indeed, we may ask whether comparable means are available for the pilot to convey status information to the computer. Status in this context refers to momentary fluctuations in attentiveness or in the ability to process information and make appropriate decisions. An operator usually communicates with a computer via manual responses or perhaps a small vocabulary of verbal responses. It is unlikely that these inputs are sufficiently sensitive to the types of momentary fluctuations mentioned above. And even if overt behavioral measures could provide such information, the process of supplying it on demand might prove extremely disruptive to critical mission-related tasks.

The program of biocybernetics research sponsored by DARPA since 1974 has attempted to alleviate the imbalance in the flow of status information by adding a communication channel from the operator to the computer. In describing the program, Donchin (1979) states:

"A channel carrying psychophysiological data (unobtrusively) acquired from the operator can supply the adaptive controller with at least part of the necessary information. The program thus assumes that mental activities manifest themselves in a variety of physiological signals. It further assumes that it is possible to make strong inferences about mental activity from such signals." (p. 3)

We believe that the virtue of this communication channel is not limited to the conveyance of status information for the purpose of effecting more or less automation. Rather, we also envisage, among other applications, a direct coupling of the pilot to aircraft subsystems from a control standpoint, such as through "thought" commands.

In this report, we examine whether the pilot's ability to satisfy tactical mission requirements can be enhanced if electrophysiological manifestations of brain function are processed by the central computer. We discuss quick-reaction situations, where the crew member is vulnerable and where performance accuracy or the speed of response may be improved. While brain electrical activity is emphasized, we consider other biological signals as well. These include peripheral psychophysiological responses and aspects of ocular behavior.

We have concentrated on pilot tasks associated with (a) the extraction of displayed information and auditory messages, (b) decision-making based upon subsequent processing of these inputs, and (c) control activation. Moreover, the pilot tasks we describe in detail were chosen because they are very difficult and/or are critical to the success of the mission, or because they occur during periods of high workload.

#### **1 OCTOBER 1979**

**MDC E2046** 

The next section of the report summarizes relevant "ground rules" for biocybernetic applications in a 1990 time period. As described previously (Gomer and Youngling, 1978), there are significant technical difficulties to be overcome before closed-loop concepts can be incorporated in operational settings. We feel that the time frame which has been adopted is reasonable, in view of the necessity for long-term planning. Sections 4 and 5 compare general mission requirements for the principal mission types and outline the information needs of the pilot, respectively. Three major subsections comprise Section 6. The first presents time line analyses of pilot tasks for a postulated escort mission. This is followed by a discussion of the particular categories of biological signals which, when interpreted by the computer, offer the most promise for improving task performance. Finally, biocybernetic applications are presented in a matrix format for each of thirteen integrated mission segments. Section 7 then offers conclusions and recommendations.

# 3.0 AN OVERVIEW OF MISSION TYPES AND PROJECTED TECHNOLOGY ADVANCES FOR TACTICAL AIRCRAFT

This summary information is intended to provide a framework for later discussions of pilot tasks and potential biocybernetic applications.

In general, we have assumed a Central European Theatre (CET), and this has influenced mission requirements, projected weapons, and aircrew functions. We also have recognized the trend toward one-man aircraft, due to reduced procurement, operating, and training costs, and the fact that fewer personnel are exposed to combat. We assume that vehicle characteristics will be typical of a two engine high performance aircraft (perhaps with direct force flight control). Ordnance should consist principally of stand-off type weapons, and command, control, and communication (C<sup>3</sup>) should encompass: Airborne Warning and Control Systems (AWACS), Global Positioning Satellite (GPS), and Joint Tactical Information Distribution System (JTIDS).

We believe that advanced crew stations will be developed with a concern for efficient information flow between the pilot and the various aircraft subsystems (cf. Mills et al., 1978). Figures 3.1 and 3.2 illustrate the essential elements of such a crew station. Included will be:

- Vertical Situation Display (VSD) presents attitude and position information in the vertical plane, also predictive or command symbology for all flight phases,
- (2) Horizontal Situation Display (HSD) provides geographic information in the form of terrain images, computer generated navigational maps and supporting symbology, and tactics,

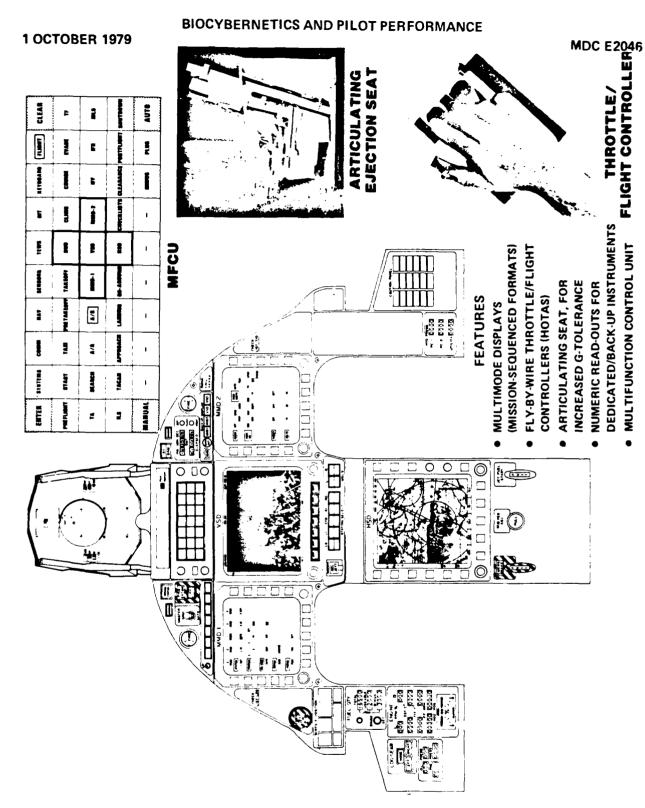
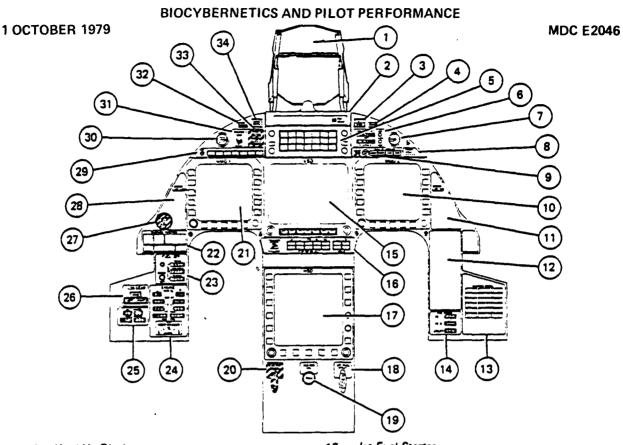


FIGURE 3.1 INTEGRATED CREW STATION CONCEPT FOR A 1990 TACTICAL AIRCRAFT.



- Head Up Display
- 2. Head Up Display Camera
- 3. Threat Alert
- 4. Canopy Alert
- 5. CNI/ECM/HUD Control
- 6. Digital Clock
- 7. R/H Engine Fire Alert and Extinguisher Button
- 8. Radio Call Number
- 9. Back Up Display Select Panel
- 10. Multimode Display No. 2
- 11. Takeoff Check List and Spare
- 12. Spare Panel
- 13. Caution and Advisory Panel
- 14. Hydraulic Pressure
- 15. Vertical Situation Display/Tactical Display (Multimode)
- Armament Station Select/Selective Jettison Panel
- 17. Horizontal Situation Display and Moving Map (Multimode)

- Jet Fuel Starter 18.
- Rudder Pedal Adjustment 19.
- 20. Emergency Brake
- 21. Multimode Display No. 1
- Surface Position Indicators (Ref) 22.
- Fuel Quantity Panel (Ref) 23.
- 24. Critical Engine Status (Ref)
- 25. Oil and Pneumatic Pressure
- 26. Landing Gear Position Indicator
- 27. Clear All Stores Emergency Jettison
- 28. Landing Check List and Spare
- 29. Mode Advisory Panel
- L/H Engine Fire Alert and Extinguisher Button 30.
- 31. Armament Master Arm Switch
- Landing Gear Alert 32
- 33. Fire Extinguisher Arm Switch
- 34. **Master Caution**

FIGURE 3.2 DESCRIPTIONS OF THE INSTRUMENTS DEPICTED IN FIGURE 3.1.

#### **1 OCTOBER 1979**

MDC E2046

- (3) Multipurpose Displays (MPDs) present system malfunctions automatically; operator selects via keyboard the information he desires (e.g., engine parameters, flight checklists, target designations, weapons status, sensor FOV, etc.),
- (4) Head-Up Display (HUD) presents weapons delivery and flight information, also limited video,
- (5) Helmet-Mounted Display (HMD) a virtual image display (as is the HUD) which also presents weapons delivery and flight information; however, it obviates the usual requirement of a fixed display location,
- (6) Multifunction Control Unit (MFCU) allows the pilot to sequence display presentations via keyboard inputs,
- (7) Hands-On-Throttle-And-Stick (HOTAS) Controls fly-by-wire, with critical subsystem (particularly weapons) controls integrated directly into the stick and throttle.

Growing alarm over the task demands imposed on the pilot and severe space limitations within the crew stations of existing tactical aircraft have forced engineers to discard a design philosophy which prescribes the use of dedicated, single-purpose instruments. An accelerated obsolescence of this conventional approach to crew station design and a new emphasis on multipurpose displays and keyboards have been brought about by remarkable developments in two areas: digital avionics and programmable electronic display devices. Large scale integration (LSI) technology has produced the necessary microcircuit/ microprocessor electronics to create an extremely flexible display generation capability. Raster-scan graphics can be used to generate realistic and varied synthetic visual scenes, and they afford an ease of

1 OCTOBER 1979 MDC E2046

mixing display symbology with the outputs of imaging sensors such as radar, low-light-level TV, and forward looking infrared. Display presentations can be limited (in theory) to those sources of information which are most relevant at a specific time period in the mission. Moreover, the sequencing of presentations can be controlled by the particular computer program(s) in operation and by the crew member's keyboard inputs.

While numerous display developments are in progress that offer great promise for the 1990 time frame we have adopted, the cathode ray tube (CRT) in all likelihood will remain preeminent. It has a superb capability for presenting almost unlimited formats (including video).

#### 3.1 MISSION TYPES

Tactical Air Power (TACAIR) recognizes three primary missions to accomplish the objective of successfully waging war. The missions are Close Air Support (CAS), Air Interdiction (AI), and Counter Air (CA). Although there are other TACAIR missions, such as Reconnaissance and Electronic Warfare, they generally are conceived to be supportive of offensive air-to-air and air-to-ground missions. Mission definitions are as follows:

<u>Close Air Support</u> - Air action against hostile ground targets that are in close proximity to friendly ground forces. This requires detailed integration of each air mission with the battle activities and movements of those forces.

**1 OCTOBER 1979** 

**MDC E2046** 

<u>Air Interdiction</u> - Air operations conducted to destroy, neutralize or delay the enemy's military potential before it can be brought to bear against friendly forces. These operations are conducted at such distances from friendly forces that detailed integration of air and ground activities is not required.

<u>Counter Air</u> - Air operations conducted to attain or maintain a desired degree of air superiority by the destruction or neutralization of enemy air forces. Both offensive and defensive air actions are involved. Offensive actions range throughout enemy territory and generally are conducted at the initiative of the friendly forces. Defensive operations are conducted near to or over friendly territory and usually are reactive to the initiative of the enemy air forces.

Figure 3.3 summarizes the scope of TACAIR's three primary missions. Obviously, a large number of mission profiles will emerge for each mission type, given the unpredictable nature of enemy operations and the broad range of weapons, sensors, and  ${\tt C}^3$  systems employed by friendly air forces. Consequently, comprehensive yet <u>representative</u> profiles were developed for the three missions.

We have chosen to concentrate on the basic elements of each mission and to avoid excessive concern for mission details that infrequently impact aircrew tasks. For example, a pilot does not particularly care whether he penetrates at 0.85 or 0.92 Mach, at 25,000 or 29,000 ft, or whether he pulls 6.2 or 7.1 g's. These are operational techniques that take advantage of particular design features and overall aircraft performance. Once a pilot is

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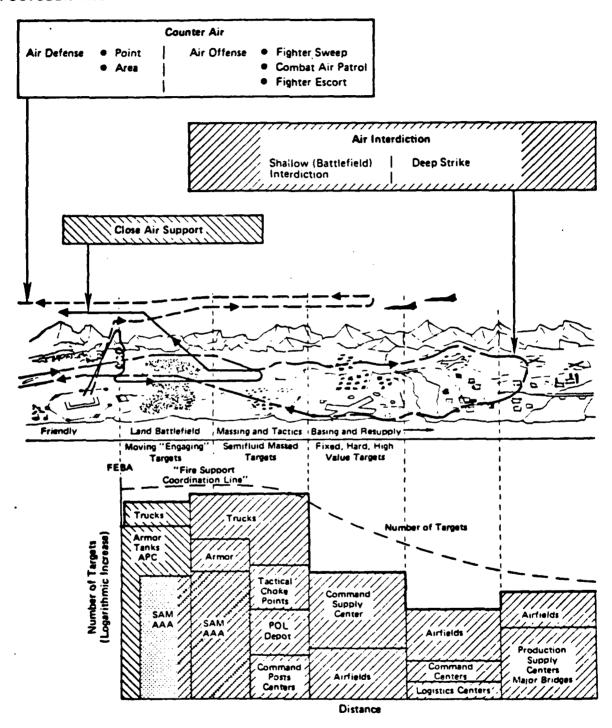


FIGURE 3.3 PICTORIAL REPRESENTATION OF THE THREE TACAIR MISSIONS.

thoroughly familiar with an aircraft he usually develops "rules of thumb" to cover the situations most commonly encountered. A pilot frequently will fly mission segments, such as low level routes, at airspeeds which are even tenths of a Mach (0.7, 0.8, or 0.9) or multiples of 60 knots (420, 480, 540). This is done in lieu of best cruise or other optimum speeds because it facilitates solving in-flight timing problems, an important parameter in all offensive air missions.

The previous example illustrated that mission timing can be more important than optimum penetration speed. Many other examples are possible that would further contrast pilot and aircraft design priorities. The mission profiles that follow emphasize the pilot priorities, especially in terms of decision-making functions and information needs.

General mission requirements are identified in Figure 3.4 for pre-flight, in-flight, and post-flight phases. Representative mission profiles were constructed from these requirements and are illustrated in Figures 3.5 through 3.7 for Close Air Support, Air Interdiction, and Counter Air missions, respectively. A detailed comparison of mission requirements for the three mission types is presented in Section 4.0.

Although we have considered the mission requirements individually, they overlap significantly in operational situations.

- 1. Pre-Flight All aircrew functions leading up to and including takeoff.
  - 1.1 Mission Planning
  - Preflight 1.2
  - Start and System Checks 1.3
  - 1.4 Taxi
  - 1.5 Arming
  - Takeoff
- In-Flight All flight activities beginning with climb and concluding at the termination of the landing roll.
  - Climb to Level-Off
  - 2.2 Cruise
  - Loiter
  - 2.3 2.4 Rendezvous and Air-to-Air Refueling (AAR)
  - 2.5 Coordination
  - Mission Rendezvous 2.6
  - 2.7 Penetration
  - 2.8 Threat Warning
  - 2.9 Detection
  - Location
  - 2.11 Identification
  - Decision
  - 2.12 Execution
  - 2.14 Assessment
  - 2.15 Termination
  - 2.16 Egress
  - 2.17 Cruise
  - 2.18 Rendezvous and Air-to-Air Refueling (AAR)
  - 2.19 Reengage
  - Return to Base 2.20
  - 2.21 Descent
  - 2.22 Approach
  - 2.23 Landing
- 3. Post-Flight All mission-related activities beginning after the completion of the landing roll and ending when the aircrew is free to perform other duties or pursue personal interests.
  - De-arm 3.1
  - 3.2 Taxi

7

- 3.3 System Checks
- Shutdown
- 3.4 Post-Flight
- 3.6 Debrief

FIGURE 3.4 GENERAL MISSION REQUIREMENTS FOR PRE-FLIGHT, IN-FLIGHT, AND POST-FLIGHT PHASES.

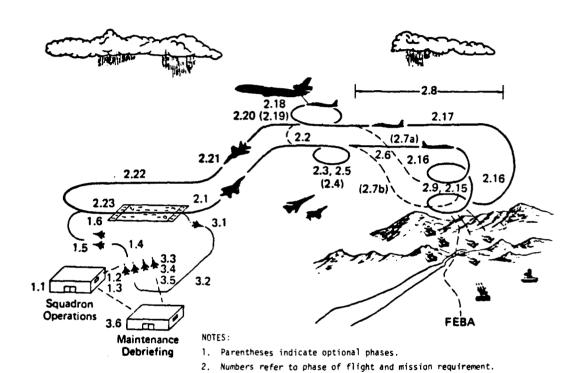


FIGURE 3.5 MISSION PROFILE FOR CLOSE AIR SUPPORT.

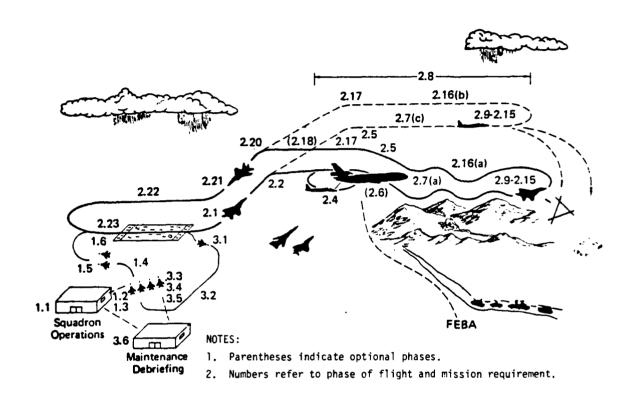
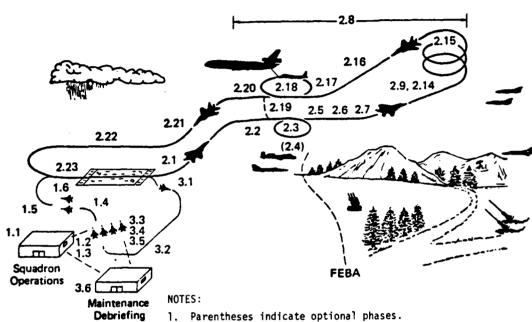


FIGURE 3.6 MISSION PROFILE FOR AIR INTERDICTION.



- 1. Parentheses indicate optional phases.
- 2. Numbers refer to phase of flight and mission requirement.

FIGURE 3.7 MISSION PROFILE FOR COUNTER AIR.

3.2 TECHNOLOGY ADVANCES

**MDC E2046** 

3.2.1 <u>Weapons</u> - The CET's defenses, weather, terrain, and target characteristics (size, mobility, signature) compromise the effectiveness of many weapons which currently are operational (cf. Levine, Beideman, and Youngling, 1978). Therefore, new concepts are needed. Stand-off weapons, both unitary and dispenser types, with midcourse and terminal guidance offer one of the most feasible solutions. Figure 3.8 summarizes the characteristics of various guidance options. These guidance systems also can be used in combination. For example, midcourse guidance can be inertial while terminal guidance can be imaging infrared (IIR) with data link lock-on.

It is important to emphasize that the key features of future weapons are that they be stand-off and that they possess accurate guidance and control. The net results are to increase survivability and effectiveness and to reduce losses and the resources required to destroy a target.

Figure 3.9 defines the various air-to-air missiles that influence pilot tasks. The physical characteristics are listed, as are the system components associated with a successful missile deployment. Important to the crew station designer are the cockpit components that are required to launch each missile. The basic purpose of each missile has been provided in the "Remarks" column.

Figures 3.10 and 3.11 present similar information for air-to-surface ballistic weapons and air-to-surface missiles, respectively.

GP77-0490-26

_			Ch	aracteris	tics	
Generic Guidan <i>c</i> e Type	Adverse Weather	Fixed Targets	Moving Targets	LOBL	LOAL	Launch and Leave
Laser Seeker	_	$\checkmark$	<b>√</b>	<b>√</b>	<b>√</b>	-
EO Tracker (Edge, Centroid, Correlation) Lock On Before Launch Data Link Lock On	- -	<b>∀</b>	<b>∀</b>	<u>√</u>		<b>^</b>
IIR Tracker (Edge, Centroid, Correlation) Lock On Before Launch Data Link Lock On	-	<b>√</b>	<b>*</b>	<u>~</u>	Ţ	<b>&gt;</b> >
IR/Radiometric Seekers (Passive)	<b>√</b>	v'	V		V	<b>✓</b>
Radar Contrast Trackers SAR Line of Sight SAR Azimuth/Range Coincidence Radar Correlation	**************************************	\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\	\\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\	\ \\	>>>>	\ \- \- \-
IR/Radiometric Correlators	<b>✓</b>	<b>√</b>		V	V	✓
Terrain Correlators (TERCOM)	V	V		V	<b>V</b>	V
Antiradiation	<b>√</b>	\_\		<b>V</b>		V
Inertial	V	<b>√</b>		<b>√</b>	V	<b>√</b>

<sup>\*</sup>Lock-on before launch

FIGURE 3.8 GUIDANCE SYSTEMS AND THEIR CHARACTERISTICS.

<sup>\*\*</sup>Lock-on after launch

TYPES OF AIR-TO-AIR MISSILES.

FIGURE 3.9

Integral rocket/ramjet; Unknown homing
Radar homing kulAWS; Stores release
3. FALCON (AIM-4F) 7.1' length; 140 lb;   Semi-active pulse radar   System   Industry; Mach 3; 5 nm   range   range   range
13.0° length; 1000 lb; Radar; Stores release semi-active radar homing system
9.6' length; 800 lb; Stores release system unguided; nuclear war-head
Approximately 10° length; Stores release system 180 lb; infrared homing
12.0' length; 450-500 lb; Radar, mission computer, semi-active (cw) radar Stores release system homing; 12-24 nm range
High velocity projectiles Radar, mission computer, Stores release system
Command inertial mid- course, active radar terminal. Physical characteristics undefined.

## 1 OCTOBER 1979

Designation	Characteristics	Aircraft System Components Required	Cockpit Components Required	Renarks
1. Modular Glide Bomb (CBU-15)	Winged 2000 lb bomb; modules	Stores release system	Slew control, HUD or slght; release switch	Could also use DME or laser guidance
2. MK-82	500 lb; 86 in.	Stores release system	Aiming sight or HUD; release switch	General purpose munition
3. MK-84	2000 lb; 12' 8" length	Stores release system	Aiming sight or HUD; release switch	General purpose munition
4. Guns: 20mm 30mm	High explosive incendiary Cun system	Gun system	HUD or sight; release switch	
5. MK-117	750 lb; 90 fn.	Stores release system	HUD or sight; release switch	General purpose munition
6. KMU-351 (MK-84 laser guided)	2052 1b; 14' length	Stores release system	Laser detector; HUD or sight; release switch	
7. MK-82 Snakeye	560 lb; 84 in.	Stores release system	HUD or sight; release switch	High drag version of MK-82
8. CBU-30/A; 90.4 in.	385 lb	Stores release aystem	HUD or sight; release switch	Cluster munition loaded with 1280 BLU-39/B23 bomblets
9. CBU-24, 49, 52, 820 lb; clam 58, and 71 88 in.	820 lb; clam shell; 88 in.	Stores release system	HUD or sight; release switch	Loaded with 670 bomblets, except CBU-52, which has 217 bomblets
10. Rockeye II	485 lb; 93 in.	Stores release system	HUD or sight; release switch	Dispenser weapon loaded with 247 MK-118 MODO bomblets
11. BLU-IC/B, 27/B Fire Bomb	100 gals; 143 in.	Stores release system	HUD or sight; release switch	Filled with incendigel

FIGURE 3.10 TYPES OF AIR-TO-SURFACE BALLISTIC WEAPONS.

	Characteristics	Aircraft System Components Required	Cockpit Components Required	Renarks
	19' length; 2800 lb; inertial/tercom	Inertial navigation system; Stores release	Control head, SMS; release switch	Strategic nuclear role  With B-1 and B-52
	11.0' length; 571 1b; commund guidance; 250 1b warhead; 6 nm range	System Unknown	Slew control; release switch	Being phased out of Inventory
	13.7' length; 770 lb; radar homing	MIANS; Stores release system	MMD; SMS; release switch	High velocity anti- radiation
	12.5' length; 1170 lb; active radar; 50+ NM range	Stores release system	MD; release switch	Anti-ship missile.
	8.2' length; 475 lb; television guided	Stores release system	MMD; controls, slew control; release switch	Laser and imaging infrared Maverick under development. Useful against tanks
	10.0' length; 400 lb; passive radar homing	RHAWS; Stores release system	Aural tone; release switch	Used on Wild Weasel air- craft (F-105G and F-4G)
	14.0' length; 2230 1b; inertial guidance; 120 nu range	Unknown	Release switch	Short range attack misaile for B-1, FB-111, and late Nodel B-52's
	14.9' length; 1355 1b; radar homing	MHAWS; Stores release system	MMD; lock on indication; release switch	Anti-radiation missile
	11.3-13.25" length; 2400 lb; television data link	Data link; Stores release system	NMD; slew control; communication CNTRL head; release switch	Unpowered TV guided glide weapon
1		A		

FIGURE 3.11 TYPES OF AIR-TO-SURFACE MISSILES.

#### **BIOCYBERNETICS AND PILOT PERFORMANCE**

**1 OCTOBER 1979** 

**MDC E2046** 

3.2.2 <u>Avionics</u> - The reduction in the cost of digital circuitry and the advantages of expandable memory are responsible for an increase in the use of digital computers onboard military aircraft. Most importantly, the avionics are more integrated, with the various subsystems under the control of higher level systems which, in turn, are under the control of the pilot. By effective interconnection of subsystems workload can be reduced, permitting the pilot to concentrate on those tasks which the central computer cannot handle effectively.

The advances in electronics will result in more accurate navigation, improved night and adverse weather target detection, jam resistant communications, and more effective countermeasures. The largest payoff may be in terms of better command, control and communication ( ${\tt C}^3$ ) systems. The programs most likely to affect aircraft systems and crew information requirements are the Joint Tactical Information Distribution System (JTIDS) and the Airborne Warning and Control System (AWACS). Although not directly a  ${\tt C}^3$  program, the Global Positioning System (GPS) may interface with JTIDS for navigation and blind navigation bombing purposes.

A summary of anticipated avionics improvements is shown in Figure 3.12, with reference to the Air Force Digital Avionics Information System (DAIS).

3.2.2.1 <u>Sensors</u> - The information which is displayed during various phases of the mission is derived from two sources: (a) real-time sensors and (b) the <u>a priori</u> data base stored in the aircraft computers before takeoff. In general, those data which are retrieved assist the pilot in performing monitoring, navigation, threat location, and target acquisition functions.

Function	Equipment Type	Anticipated Advances
NAV	Inertial	Strapdown; improved gimballed systems
ł	ADC	Computations performed in processor
1	TACAN, ILS, ADF	Weight, size, power consumption
	GPS, OMEGA	Will become operational
	Hybrids	GPS/inertial available
COM		MSI, LSI; Shared Antenna ; Modulation techniques
	JTIDS	Will become operational
	IFF	Improved performance, cost reductions
ECM		Power management
Air/Ground Attack	Lasers	Laser target seekers will become operational
	FLIR	Reduced costs, weight, size; improved display
	FLR '	Improved performance, reliability
C/D .	Controls & Displays	Current displays and controls will be replaced by computer-driven MPD plus IMFK
Processing	Processors	LSI technology and microprocessors with large central processing capability
		BITE contained within microprocessors
		MOS (metal oxide semiconductors) Bipolar Schottky transistor-transistor logic
Power Supply		Central power supply core element
Interface Equipment	BCIU, RTU	Integrated within sensor or processor

FIGURE 3.12 ANTICIPATED AVIONICS ADVANCES (WITH REFERENCE TO DAIS).

The sensors, on the other hand, provide essential information for all house-keeping and mission-related functions, particularly navigation, communication, threat detection/ location, and the execution of air-to-air or air-to-ground attack.

Representative sensors onboard the aircraft include:

- o multimode radar and forward looking EO or IR sensors for target recognition; laser designator/ranger/tracker for attack.
- o imaging seekers for missile guidance (see Figure 3.8) (implied is the dedicated fire control computer for directing these seekers to the aimpoint coordinates available from the target acquisition sensors),
- o navigational sensors for INS, TACAN, and GPS,
- o radar and laser threat (ground and air) warning sensors, IFF threat identification receivers, and data-link receivers (e.g., JTIDS). We should mention that a more inclusive listing would encompass the feedback devices which are used to monitor the status of the various aircraft subsystems. By restricting our focus to those baseline sensors listed above, however, we can describe their characteristics in greater detail (see Figure 3.13).
- 3.2.2.2 <u>Electronic Displays</u> In the next two sections (4 and 5) of this report, the mission requirements and the associated information needs of the pilot are described for the reader. These sections demonstrate that flexibility in display generation is essential to satisfy future mission objectives. The pilot must be able to select, or be provided automatically, integrated information which is updated or changed accordingly across specific time periods in the mission.

FIGURE 3.13 SENSOR CHARACTERISTICS.

SENSOR	TYPE OF INFORMATION PROVIDED	FORMAT IN WHICH INFORMATION IS DISPLAYED	SENSOR	TYPE OF INFORMATION PROVIDED	FORMAT IN WHICH INFORMATION IS DISPLAYED
AIR-TO-GROUND RADAR			AIR-TO-AIR RADAR		
o Real Beam Ground Map (RBGM)	Video Imagery; Cursors; AZ and R Markers	Sector with R vs. AZ	o Pulse Search o Velocity	Target Symbols: Acquisition Symbol;	R vs. AZ VEL vs. AZ
o Doppler Beam Sharpened (DBS)	Video Imagery; Cursors; Sector - AZ and R Markers Patch - AZ and R Numerics	Sector or Patch	Search O Range While Searching O Track While Scanning	AZ and tL Carets; Flight Symbology	
o Synthetic Aper- ture Radar (SAR)	Video Imagery; Cursors	Passing Scene Map or Track; EL vs. AZ	HIGH RESOLUTION TV LOW-LIGHT-LEVEL TV	TV Imagery TV Imagery	EL vs. AZ EL vs. AZ
o Ground Moving Target Indication (GMTI)	Target Symbol on Ground Map Video; Designating Cursor	RBGM Sector, DBS Sector/ Patch or SAR Map	(LLLTV) FORWARD LOOKING INFRARED (FLIR)	8 to 11 pm Imagery	EL vs. AZ
o Air-to-Ground Ranging (AGR)	Target Symbol at Messured Slant Range;	Sector with R vs. AZ	MISSILE THERMAL IMAGING	8 to 11, M Imagery	EL vs. AZ
	Numerical Readout of Range		MISSILE EO	TV Imagery	EL vs. AZ
o Terrain Following	Terrain Profile;	EL vs. Log R	EO TRACKER	TV Imagery	EL vs. AZ
	Altitude		IR TRACKER	3-5 M Imagery; Hit/Miss Warning	EL vs. AZ
o Terrain Avoidance (TA)	Video of Return Above Clearance Plane; Horizon Line Symbol	Sector with R vs. A2	IR WARNING	Hit/Miss Video; Bearing of Missiles and Inter-ceptor	Symbol on EL vs. AZ Display
			LASER TRACKER	Direction of Line-of-Sight	Symbol on EL vs. AZ Display or on HUD
			LASER OR RADAR WARNING	Hit/Hiss Video; Bearing and Identification of Illuminator or Emitter	AZ vs. Lethal Range Rings: Threat Symbols

### **BIOCYBERNETICS AND PILOT PERFORMANCE**

#### **1 OCTOBER 1979**

**MDC E2046** 

We stated earlier that flexibility is afforded, in part, by the incorporation of multifunction display techniques. Our purpose in this subsection is to present a brief comparison of those electronic display media which will be competing for inclusion in tactical crew stations during the 1990s. Recall that Figures 3.1 and 3.2 illustrated the display units which will present visual information to the pilot. They consisted of:

- o Multipurpose Displays (MPDs),
- o Horizontal Situation Display (HSD),
- o Vertical Situation Display (VSD),
- o Head-Up Display (HUD),
- o Helmet-Mounted Display (HMD).

Presentation media for alpha numerics, symbolic characters, graphics, and full video should include:

- o Cathode Ray Tubes (CRTs),
- o Liquid Crystals (LCs),
- o Plasma Devices (PLDs),
- o Light Emitting Diodes (LEDs),
- o Electroluminescence Devices (ELs),
- o Laser Displays (LDs),
- o Electromechanical Devices (EMDs).

To allow a meaningful comparison of these display devices, consider the information formats and coding schemes (Figure 3.14) with respect to which the competing technologies will be assessed (see Figure 3.15). Other factors certainly must be taken into account when establishing evaluation criteria, and these are illustrated selectively in Figure 3.16.

- A On-off: legend
- B Alpha numerics: fixed alphabet
- C1 Characters (symbols and alpha numerics), raster or matrix generated: variable alphabet
- C2 Characters, stroke generated: variable alphabet
- D Graphics and characters: fixed format and alphabet
- E1 Graphics and characters, raster or matrix generated: variable format and alphabet
- E2 Graphics and characters, stroke generated: variable format and alphabet
- F Full video (or fixed image)
- G Color coding
- H Size coding
- I Depth coding
- J Time coding (typically a software problem)

FIGURE 3.14 INFORMATION FORMATS AND DISPLAY CODING SCHEMES.

DISPLAY TECHNOLOGIES

		LD	EMD	EL	LED	PLD	LC	CRT
	Α	Х	X		X	X	X	x
	В	х	X	X	X	X	X	х
	cl	х			X	X	X	х
	$c_2$	х						į
DISPLAY	D	х	X		X	X	X	х
FORMAT TYPES (SEE FIGURE	E <sub>l</sub> .			X	X	X	X	x
(SEE FIGURE 3.14)	E <sub>2</sub>	х						х
	F	Х			?	?	X	x
	G	Х	X	X	?	?		?
	Н	х	?	χ	X	X	X	х
	I					•		1
	J	Х	?	Х	Х	Х	Х	χ

FIGURE 3.15 COMPARISON OF DISPLAY TECHNOLOGIES, WITH RESPECT TO INFORMATION FORMATS AND CODING SCHEMES.

PARAMETER	PLASMA PANEL	ELECTROLUMINESCENCE (THIN FILM)	LIGHT EMITTING DIODES	LIQUID CRYSTAL	CRT
Resolution	30-60 epi	20-50 epi, large panels 500 epi, small panels	30-50 epi	100 epi, large panels 588 epi, small panels	120-1600 epi
Brightness	>30-50 ft L	>30 ft L	>100 ft L .	Ambient Dependent	>30-1000 ft L
Contrast	>20:1 Binary	≥10:1 Full Video	100:1	20:1 Full Video	20:1 Full Viaea
Display Size	<1024 elements T7 inch sq (ac driven)	240 elements, 6 inch 500 elements, 1 inch	250 elements, 6 in. sq.	350 elements, 3.5 in. 600 elements, 1 in.	480-830 elements 1-25 inch sq.
Color	Primarily Neon Orange Full Color Possible	Phosphor Dependent Full Color Possible	Red; green and yellow also available	Primary Black/White but Contrasting Colors are Available	Phosphor Dependent, Discrete Colors Also
Power Requirements	200-300W	10W	1.5-2.0W/cm <sup>2</sup> 01.5-2.0 volts	5_w/cm <sup>2</sup> to 1.0 mW/cm <sup>2</sup> @3-15 volts	100 Watts
Thickness	<1 inch	<1 inch	-1 inch	<1 inch	=12-18 inch
Weight	50 1bs	-	-	•	Up to 50 Lbs.
Environment	Rugged	Rugged	Rugged	Rugged but Tempera- ture Limits	Ruggedized
Aspect Viewing	Wide Aspect	Uniform, Wide Aspect	Slight Gain But Basically Uniform	Restricted	Uniform Wide
Time Constants	Std. Video (dc driven)	Compatible With Std. Video	Compatible With Std. Video	10-500 ms (with scan converter LC can display std. video)	3-10MHz Std. Video
Storage/ Refresh	yes (ac driven) no (dc driven)	Yes, but is config- uration dependent	No	Limited Storage	None (storage CRTs are available)
Reliability/ Maintainability	100,000 hrs LRU	1,000 to 10,000 hrs, LRU	10,000 to 100,000 hrs, LRU	20,000 hrs, LRU	15 to 15,000 hrs. LRU
Status	Operational/ Commercially Available	Laboratory Demonstra- tion Models	Commercially Available/ Operational	Operational/Labor- atory Demonstration Models	Operational

FIGURE 3.16 EVALUATION PARAMETERS AND CORRESPONDING PERFORMANCES OF SELECTED DISPLAY DEVICES.

The CRT will probably remain the fundamental display medium in the 1990s, because of its versatility in generating data for all display units listed above (i.e., MPD, HSD, VSD, HUD, HMD). However, as new display media are developed (e.g., electrophoretic, magnetic particle, and electrochromatic devices) the pilot's information needs, which translate to display formats and codes, must be considered as the most significant evaluation criterion.

3.2.2.3 <u>Multifunction Control Unit</u> - Multifunction switches, a relatively new concept in control technology (see Figure 3.1), are a counterpart of computer generated multifunction displays. They may be thought of as versatile sets of switch contacts which perform different switching operations.

Each switch within a multifunction control unit addresses computer logic, which determines the specific function of that switch and initiates the desired action when the control surface is activitated. Since the function of a particular switch changes, <u>current</u> status of the switch must be displayed. There are several ways in which this can be accomplished:

- o rear projecting legends onto pushbutton switches,
- o generating legends remotely and transmitting to switch face with fiber optics,
- o generating legends on an electronic display with switches located in the periphery,
- o generating legends on an electronic display and then activating the area which has been designated (touch, light pen, photo-sensitive detectors, etc.),
- o generating legends directly on the switch face.

3.2.2.4 <u>Voice Actuated Controls</u> - As modern aircraft become increasingly more sophisticated, additional responsibilities are created which add to an already burdensome pilot workload. The concept of voice actuated systems (VAS) is one potential solution to this problem. In fact, the applications can be extended in principle to incorporate "thought" commands, as discussed in Subsection 6.3.

The purpose of VAS is to permit more direct communication with the computer, thereby alleviating conventional workload and providing additional time to perform higher level tasks.

The particularly desirable features of VAS are: a reduction in manual control procedures (especially during critical flight phases), a reduction in eye-hand coordination problems, and a reduction in visual demands inside the cockpit. A representative application may be found in weapons delivery. During this mission phase, the pilot is concerned with the sequence of events necessary to properly arm and deliver weapons on a target, to maintain altitude and airspeed within acceptable limits, to perform required communications, and to remain alert for enemy threats. With VAS, it may be possible for the pilot to change radio channels and to arm and deliver weapons by simple voice commands. Hence, more time and attention can be given to target acquisition, lock-on, flight control and threat avoidance.

Voice data entry systems are already in use for a number of interactive command and control functions. However, these systems have a limited vocabulary and are speaker specific. Vocabularies typically consist of digits and a small set of control words and phrases.

3.2.2.5 <u>Joint Tactical Information Distribution System (JTIDS)</u> - JTIDS is a digital communication system for secure, real-time command and control of combat operations. JTIDS will interconnect the tactical defense elements of all services with surveillance/intellegence centers and with command and control centers in the theatre of operations. Precise time-of-arrival measurements, coupled with the transmission of emitter location, are used to generate a common grid coordinate system containing the location of all active net participants. The system uses Time Division Multiple Access (TDMA) to interconnect all users via one common channel for the distribution of information. Each authorized element is allocated a number of transmission time slots. When not transmitting, each element monitors the transmissions of all other elements and extracts the information as needed.

Although the specific benefits of JTIDS have not been documented, several generic benefits affect tactical missions. These are:

- o jam resistant communications,
- o intercept enhancement,
- o "Beyond Visual Range" threat identification,
- o supplementary threat warning,
- o relative nagivation,
- o blind NAV bombing capability.

JTIDS is a multiphased program which is scheduled to achieve full operational status by 1984.

## **BIOCYBERNETICS AND PILOT PERFORMANCE**

## **1 OCTOBER 1979**

**MDC E2046** 

- 3.2.2.6 <u>Airborne Warning and Control System (AWACS)</u> AWACS will provide air surveillance for command, control and communication functions throughout the U.S. and overseas. Its radar will detect and track aircraft at any altitude over land and water.
- 3.2.2.7 <u>Global Positioning Satellite (GPS)</u> GPS will provide precise, three-dimensional position and velocity information to aircraft, ships, and ground forces. The GPS development is currently in the validation phase, with early results indicating that the system can meet anticipated accuracy requirements. Full scale development should begin in 1982. Eventually, 24 satellites will be orbited for full global coverage by 1984.

**1 OCTOBER 1979** 

# 4.0 MISSION REQUIREMENTS

General mission requirements were listed earlier (see Figure 3.4) for pre-flight, in-flight, and post-flight phases. From these requirements, representative mission profiles were illustrated for Close Air Support, Air Interdiction, and Counter Air missions (see Figures 3.5, 3.6, and 3.7, respectively).

Our intent in this section of the report is to describe the mission requirements in more detail. This is followed (in Section 5) by an examination of pilot information needs. After establishing tactical objectives and defining the corresponding information requirements, we present (in Section 6) time line analyses of the specific tasks which must be performed.

We have chosen a presentation format (adapted from Mills et al., 1978) which will facilitate comparison of the requirements for the principal TACAIR missions. The numbering scheme used to identify individual mission requirements is the same as that used in Figure 3.4. Notes are included when further clarification or differentiation is necessary. We should emphasize, again, that the mission requirements overlap in combat situations.

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
<ol> <li>Pre-Flight</li> <li>Mission</li> <li>Planning</li> </ol>	tht 1.1 All tasks related to safe and effective conduct of the mission. These include gat a minimum:	1.1 Same as Close Air Support. (See Note 2)	<pre>1.1 Same as Close Air Support. (See Note 3)</pre>
	(a) Briefings: weather, intelligence, rules of engagement, targets, and weapon deliveries.		
	(b) Computations: time, distance, headings, fuel, and ballistics.		
	(c) Maps.		
	(d) Enroute Procedures: for- mation, navigation, communica-		

Close Air Support missions emphasize quick response and receive the highest priority when friendly forces are engaged with the enemy and in desperate need of all available resources. Therefore, if a mission is scrambled from an alert status or if aircraft are diverted from another mission, there may be little or no time for mission planning. In these cases, the aircrew must rely on experience, on-board systems, and in-flight communications.

Since Air Interdiction missions are generally conducted against large, less mobile targets, there is frequently more time for detailed flight planning. With the exception of mobile targets, target location and description are usually known very accurately. Although Counter Air missions demand considerable pilot skills, they generally require less detailed mission planning than the air-to-ground missions. However, more in-flight decisions must be made. (2)  $\widehat{\mathbb{C}}$ 

FIGURE 4.1 PRE-FLIGHT MISSION REQUIREMENTS FOR CLOSE AIR SUPPORT, AIR INTERDICTION, AND COUNTER AIR.

(e) Air refueling procedures and call signs.

(f) Alternate missions.

(See Note 1)

COUNTER AIR	Same as CAS. (See Note 4)		
AIR INTERDICTION	Same as CAS.		
CLOSE AIR SUPPORT	(a) Personal Equipment: all tasks associated with the inspection and donning of flight gear (e.g., g-suits, parachutes, helmets, etc.).	(b) Aircraft: assure aircraft is in physical condition for flight and that it is equipped properly for the proposed mission objectives. Check fuel or hydraulic leaks, weapon configurations, reservoirs, tires, panels, etc	(c) The aircrew is responsible for checking the AFTO 781 to assure that the aircraft has been properly serviced and released for flight by maintenance. Open discrepancies (items not repaired) and their effect on the proposed mission are evaluated at this time.
PHASE	.2 Pre-Flight		

NOTE: (4) Most pre-flight tasks are performed in advance of an alert. Otherwise, these tasks are performed just prior to the mission. Alert status requires specified aircraft and aircrews and, therefore, reduces the resources available for other missions or duties.

FIGURE 4.2 PRE-FLIGHT MISSION REQUIREMENTS (CONTINUED).

	PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR	
<b>ن</b>	.3 Start and System Checks	1.3 Check normal system operations. Verify equipment applicable to mission. Make go/no go decision depending on equipment status.	1.3 Same as CAS.	1.3 Same as CAS.	
₹.	.4 Taxi	1.4 (a) Receive clearance.	1.4 Same as CAS. (See Note 5)	1.4 Same as CAS. (See Note 5)	
		(b) Texi aircraft from parking area to arming area, exercising caution to avoid other aircraft and maintenance vehicles and equipment.			
		(c) Aircraft must be taxied in appropriate formation, regardless of location on ramp.			
		(See Note 5)			

NOTE: (5) Ramp congestion or damage, night operations, a new eirfield with unfamiliar takeoff procecures, and other unusual circumstances can significantly confuse taxi cperations. The resuits can vary from a delayed mission to damaged or even destroyed afroraft.

FIGURE 4.3 PRE-FLIGHT MISSION REQUIREMENTS (CONTINUED).

Nuclear weapons will

		cept: (d) does	igh much less	e a lesser errect performance.		
COUNTER AIR	1.5 Same as CAS.	1.6 Same as CAS ex	tions generally weigh much less	and therefore have a lesser er on the aircraft's performance.		
AIR INTERDICTION	1.5 Same as CAS.	1.6 Same as CAS.	(See Note 7)			
CLOSE AIR SUPPORT	1.5 Aircraft weapons and munitions are armed and mechanically unsafed by ground crews. The aircrew is required to maintain hands off all switches and controls and, if possible, in such a position that they are visible to the ground crew.  (See Note 6)	1.6	(a) Receive clearance.	<ul><li>(b) Lineup checks (systems and configuration checks).</li></ul>	(c) Takeoff Roll - One of most important phases of flight because of the criticality of major subsystem malfunctions.	(d) With live munitions on-board, this phase of flight is further complicated by different handling characteristics, including rotation and lift off speeds and slower acceleration.
PHASE	1.5 Arming	1.6 Takeoff				
۵.	٠.	9.				

Arming is generally conducted in a clear area as close to the end of the runway as possible. No additional maintenance can be performed on the aircraft unless it is once again safed. Although arming is a relatively simple procedure, it ranks very high in importance. If done improperly, it can result in an ineffective mission or damage to and loss of the aircraft. In some aircraft, it is impossible for the aircrew to confirm that the ground crews have properly performed their tasks. There are no cockpit indications that "pins" have been pulled or that a weapon, such as a gun, is armed.

MOTES: (6) A If nuclear weipons are carried, then takeoff, abort and jettison procedures may change. not be jettisoned from the aircraft, even during an emergency.

3

FIGURE 4.4 PRE-FLIGHT MISSION REQUIREMENTS (CONCLUDED).

NOTE: (8) Formation procedures for a specific aircraft are very clearly defined and allow for little deviation.

AIR INTERDICTION COUNTER AIR		2.1 Same as CAS except: 2.1 Same as CAS.	ion considerations t always be a factor. ir interdictions are ted by a single air-	craft.							2.2 Same as CAS. In 2.2 Same as CAS.		may be much greater than for CAS or Counter Air.	(b) Time of arrival con-
CLOSE AIR SUPPORT		2.1	(a) Monitor systems and perform additional system .	(b) Monitor formation. Formation considerations are involved throughout the re-	mainder of the flight and vary depending on phase, weather, and enemy action (The	requirements related to leading a formation are so	complex that it is impossible to briefly address them.	They are, however, a primary consideration in multi-aircraft missions.)	(c) Follow clearance instructions.	(See Note 8)	2.2	(a) Monitor systems.	(b) Review mission.	(c) Follow clearance in-
PHASE	2. In-Flight	Climb to	- -								2.2 Crufse			

FIGURE 4.5 IN-FLIGHT MISSION REQUIREMENTS FOR CLOSE AIR SUPPORT, AIR INTERDICTION, AND COUNTER AIR.

	FHASE	CLUSE AIR SUPPURI	AIR INIERDICITON	COUNIER AIR
2.3	2.3 Loiter	2.3 Remain in a specified location awaiting further enemy activity or until additional instructions are received.	2.3 Not normally required.	2.3 Same as CAS.
2.	2.4 Rendezvous and Air- to-Air Refueling (AAR)	(2.4 Optional. Otherwise same as Air Interdiction. AAR would only be required if there were extensive delays in getting to a target or if the airbase is at a greater distance from the target area than generally desired.)	(a) Rendezvous consists of Joining mission aircraft with tanker aircraft. This phase relies heavily on standard procedures and precise inflight timing and navigation.  (b) This phase must be conducted in a "safe" area or the tankers must be provided cover against enemy aircraft.	(2.4 Optional. Otherwise same as Air Interdiction. Would be used to extend loiter time awaiting engagements.)
			(See Note 10)	

Both loiter and AAR extend mission time. This has several advantages, such as for decreasing the response time to get friendly air power over enemy targets. As long as adequate weapons are on-board an aircraft, it is advantageous to keep it airborne awaiting conflict. Limits to be considered are fatigue, exposure to risk, and the availability of support forces (such as tankers). AAR can also be directed from ground radar control facilities. This is a more efficient method which reduces aircrew task loading. However, it relies heavily on radar and communications and, therefore, is subject to jamming or deception. (9) B (0)

FIGURE 4.6 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

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Since many of the aircraft dedicated to this mission have a nuclear capability, the aircrews must be knowledge-able of all applicable directives and safety requirements.

(15)

The Tactical Air Control System (TACS) coordinates both preplanned and immediate CAS operations. This system provides hardware, command and control, and an approved interface with Army units for CAS and some shallow interdiction missions.

(See Note 11)

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.5 Coordination	2.5	2.5	2.5
	(2) Aircrew receives general instructions and sufficient details to contact a specific ground or airborne forward	(a) Receive final clearance for primary mission or a change to a secondary mission. Recent information will be	(a) Fighters will be directed to areas of suspected enemy activity.
	<pre>air controller (FAC). (b) FAC provides details of the targets, such as type, number, and location. Enemy</pre>	added, such as a weather up- date or the position of enemy fighters. (b) Authenticate message if a	(b) Warnings will be transmitted when enemy aircraft are observed to be maneuvering for an attack on friendly fighters.
	defenses and other restric- tions are communicated. Friendly troop positions are designated.	change in mission is conveyed. This may also be required to confirm go-ahead on primary mission.	(c) SAM warnings will be issued.
	(c) The FAC, in many cases, will act as the controller and give positive clearance to release ordnance. This is particularly true in cases where enemy and friendly forces are in very close contact.	(c) Additional information may be relayed in the "blind", such as fighter escort call signs, tanker location, and SAM warnings.	
	(d) The mission aircraft will normally hold clear of the target area until this information is completed. An exception is a permissive air environment.		

FIGURE 4.7 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

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Elements joining together in a common strike force may be from different units or bases. They may not have been briefed together and, therefore, will rely on instructions provided through the chain of command. Rendezvous relies heavily on radar, UMF communications, and pre-briefed signals or codes.

NOTE: (13)

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.6 Mission Rendezvous	2.6 Join-up at a specified position and time with other elements of the strike force. These may include fighter cover and screening forces, such as ECM or Chaff dispensing aircraft. Accurate detection and identification are important during this phase to avoid errors. Common radio frequencies facilitate this type of join-up.  (See Note 13)	(2.6 Optional: Not normally required on deep interdiction missions. May be required on shallow interdiction missions, in which case the same as close air support.)	(2.6 Optional:  (a) On many missions a counter air strike force operates independently. In this case, positive target identification is required prior to attack.  (b) Join-up with other elements in the strike force may be required. Usually, the counter air forces are tasked to provide attack cover. This Join-up could be with elements of either CAS or Air Interdiction missions and may require different tactics to maintain formation integrity.)

FIGURE 4.8 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

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2.7 Penetration

(a) Low: For deep interdicmasking for concealment untion this is a preferred method. It uses terrain til the aircraft reach be used primarily when enemy forces have no ground de-(a) Medium: This tactic will fire. Also used when enemy fighter cover is provided. CAS aircraft will fly into fenses or only light arms air is not a threat or if

to patrol for enemy

fighters.)

that counter air scrties would, by themselves, penetrate heavily defended areas to patrol for enem [2.7 Optional: It is unlikely

> including terrain following radar (TFR) and very accurequires sophisticated airadverse weather conditions, craft systems for night or rate navigation equipment. reduces exposure to enemy defenses. This mission advantage of surprise and target area. This takes

> > to combat altitudes. Fighter

the target area and descend

cover would remain at higher

altitudes.

ployed when the CAS aircraft defended area. The formation

(b) Low: This tactic is

are operating in a heavily

would be split or staggered and would penetrate at very

ow altitudes using the

self-defense mechanisms could every enemy defense system an apportunity to detect and fire at the aircraft. dered an unattractive opti<mark>on</mark> (b) Medium: This is consibecause it provides almost Advances in ECM and other make it viable option. terrain to mask its position. In the vicinity of the target

area the aircraft would "pop

to deliver the weapons

up.

(c) High: This option allows the aircraft to fly above most duce exposure to the few enemy aircraft capable of sustained high altitude operation. relatively high speeds to reenemy threats. It requires

target area must be somewhat familiar or easily distinand when the ordnance is expended, egress at low altitudes. This profile is considerably more demanding on the crew, since the low ittle time for search and level route must be flown with great precision. The juished, since there is dentification

IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

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CLOSE AIR SUPPORT

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2.8 Basically, the same requirements as for CAS aircraft. In general, air interdiction aircraft are sometimes defensive systems. However,

ground-based alerts or cautions. since they operate at a greater distance from friendly forces, they cannot always make use of more capable of evading enemy (a) Aircrow must be prepared well in since CAS aircraft are usually slower and less maneuverable, especially advance of actual enemy air attacks,

when heavily loaded with ordnance. Threat information may come from on-board systems or be relayed from

ground radar.

are the most significant threats to counter air aircraft. Since new count-er air aircraft are extre-mely capable, they can

avoid or defeat SAMs and enemy fighters if detected

are necessary to provide additional protection and to relieve self-defense task loading.

(b) Some countermeasures well enough in advance.

(a) SAMs and enemy fighters

(b) The location of ground-based enemy systems must be known, pre- . ferably in advance of mission. CAS aircraft are very vulnerable to such systems, particularly SAMs and heavy

(c) The prescribed action to defeat the detected enemy threat should be inherent in a detection system. This can vary from maneuvering to the use of chaff; flares, or ECM

2.8

Threat Warning

PHASE

FIGURE 4.10 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

COUNTER AIR	2.9 Since enemy aircraft are the only targets considered, information related to their positions must be near realtime. Therefore, there is a much greater emphasis on aircraft detection systems.		
AIR INTERDICTION	2.9 This task, because of the general target types, is usually completed well in advance of the mission. The target type and characteristics are briefed in the mission planning.  (a) Air interdiction is concerned only with enemy ground targets.		
CLOSE AIR SUPPORT	(a) First phase of actual combat. Necessary information may have been provided prior to the mission. (b) This task can be accomplished by a ground-based or airborne FAC. The FAC will then advise the mission aircraft of the target positions.	(c) In designated free fire zones, the flight leader may detect targets of opportunity without "detailed coordination" with ground forces.	(d) CAS is concerned only with detecting enemy ground targets.
PHASE	2.9 Detection (Determination that a potential target may exist. Sources may vary.)		

FIGURE 4.11 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.10 Location (Potential target positioned within a designated ref-	2.10 (a) Location must be known very accurately, particularly	2.10 iocation must be known precisely. The tasking of this mission is so demanding that on-board systems must	2.10 Location for counter air targets is in terms of rela- tive position. This information must be known for air-to-air
erence system. The inpurpose is to locate T the target with sufficient accuracy by to support the refunctions.	in reference to friendly forces. This task can be initiated before flight, but usually must be completed in-flight since targets may have moved or may be camouflaged.	direct weapons as close to the target as possible.	tactics and for selection of weapons. Pre-mission planning is usually of limited value. In-flight updates are necessary to give counter air forces an overview of the air battle.
identification, de- cision, execution, and assessment.)	(b) The aircrew must make final determination of the target location. (See Note 14)		(See Note 15)

The value of detailed mission planning is reduced because air targets can vary their locations over relatively broad spectrum of altitudes. NOTES: (14) Future weapon concepts present options that may eliminate aircrew involvement in target location.

(15)

FIGURE 4.12 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

COUNTER ATR	2.11 Identification in an air battle is extremely important, particularly when different formations and aircraft types are involved. Therefore, "beyond visual range" targets classification as friend or foe. These include IFF and radar signature. Close-in combat is ideally suited for visual identification.
AIR INTERDICTION	2.11 Identification is important to mission success, but to a lesser extent than for CAS. Since air interdiction attacks are conducted well behind enemy lines, errors are less likely to affect friendly forces.
CLOSE AIR SUPPORT	2.11 The target must be clearly identified as enemy. If this cannot be done, the attack cannot continue. Visual confirmation is usually required.
PHASE	2.11 Identification (Potential targets are classified to support decision making. This may require information from several sources or simple visual identification.)

In an all-weather CAS scenario, not only is it important to identify a target as a tank, but also as a friendly or enemy tank. This difference, of course, is substantia! and the requirement further complicates an already NOTES: (16) I

This is not always true, since the "rules of engagement" (ROE) may require positive target identification. The goal in this regard is either to limit nonmilitary, collateral damage or to avoid acts that might precipitate alignment of other countries with the enemy.

(1)

FIGURE 4.13 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.12 Decision (Potential targets and available strike resources are assessed and a course of	2.12 The final decision regarding actual weapons release is made by the pilot. He makes this decision based on the information received up to and including the time of release. (In combat there is very real	2.12 Same as CAS. Additionally, the attack may be conducted against targets not, visually identified.	2.12 (a) Same as CAS except: In close combat, the decision is only "whe to fire" since there is usually little doubt as to target identification.
mined. This function is in- tegrative in nature.)	pressure to release the ord- nance and egress from the target area.)		(b) Beyond visual range targets must be classified by other metho and, therefore, the confidence level of any decision is reduced.
			(See Note 18)
2.13 Execution (This is the function in which the chosen course of action is implemented. This action may be either defen- sive or offen- sive.)	2.13 Execution implies that the command be given to the release mechanism to free the weapon from the aircraft. (In advanced weapon systems, the weapon may require guidance until final impact.) Frequently, the actual time of release is computed automatically and, provided there is consent, the weapon is delivered. Execution presumes all other arming functions are completed.	2.13 Same as CAS.	2.13 Same as CAS.

The decision-making process is not easy to explain. Training, ROEs, previous experience, and many other factors affect split-second decisions to release weapons. NOTE: (18)

FIGURE 4.14 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

NOTE: (19) Each combat mission has the potential of acquiring additional intelligence information.

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.14 Assessment finis is the function in which the impact of aerospace operations upon enemy forces and capabilities is appraised. The appraised of this function is to provide an objective evaluation of the degree to which current operations achieve assigned objectives.)	2.14 This task requires objective evaluation of resultant damage. The aircrew can contribute to the success of this task, however, conditions may reduce the aircrew's ability to observe or assess damage in the target area. Included are such limiting factors as smoke and dust, enemy activities, weather, etc. Therefore, this task is frequently performed by the FAC or during a reconnaissance flight. The importance of assessment is directly proportional to the priority of the target. Frequently, the pilot has no way to record his assessment and must rely on memory during debriefing.	2.14 Same as CAS, except the FAC may not be in a position to evaluate the mission success.	2.14 In close combat, assessment may be immediately obvious or on the other hand, impossible. The task is further complicated since later reconnaissance flights are of little value.

FIGURE 4.15 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

(See Note 19)

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
.15 Termination	2.15 At some point, the air- crew must terminate the engage.	2.15 Same as CAS except:	2.15 Same as CAS. An addi- tional consideration is
	ment. This decision is based on	(a) Deep strike air inter-	that the aircrew must be
	the following variables.	diction missions terminate	able to disengage from the
	The state of the s	at weapons delivery. Addi-	enemy fighters. Termination
	(a) larget status	usually planned, however, more	is not diways as easily achieved as in air-to-
	(b) Fuel status	than one target may be fired	ground missions.
		nbon.	
	(c) Orgnance remaining	(b) Shallow intendiction	
	(d) Other circumstances, such	missions may allow for mul-	
	as mission recall, heavy enemy	tiple attacks on the same	
	defenses, damaged aircraft,	targets, in which case the	
	deteriorating weather, etc.	same criteria as for CAS will	
		apply.	
2.16 Egress	2.16 Depart from target area.	2.16 Low: Depart from target	2.16 Normally not required.
	This task includes all evasive	area in the same manner as	unless escorting other air-
	tactics required to avoid	penetration (see 2.7a), usu-	craft or engaging in other
	enemy defenses and to rejoin	ally along a different route.	operations over enemy terri-
	the other aircraft in forma-		tory. In these cases,
	tion. Weather and other visi- bility restrictions complicate	High: Same as penetration (see 2.7c), except that route and	altitudes are usually medium to high.
	the task. If returning from	altitudes may vary.	
	enemy territory, "safe-passage" procedures may be required.		

FIGURE 4.16 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

Air 2.18 Same as 2.4 except: This Refueling AAR is much more critical. If there are any errors in planning or in estimating fuel consumption, distances, or times, then all aircraft on the mission ould be lost. If the mission is terminated too	must be flown. Alternate or emergency plans are formulated depending on existing conditions. (See Note 20)  2.18 Same as 2.4 except: This (2.18 Optional. Same AAR is much more critical. If as CAS.) there are any errors in planning or in estimating fuel consumption, distances, or times, then all aircraft on the mission could be lost. If the mission is terminated too	2.17 Same as CAS. May demand multiple AAR to meet station time and fuel requirements.
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The aircrew must have a variety of inputs on the status of friendly airfields. These include weather, navigation aids, approach procedures, etc. Much of this information decays with time and, therefore, is not useful if stored prior to flight. NOTE: (20)

FIGURE 4.17 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.19 Re-engage	(2.19 Optional: Additional attacks after AAR may be possible depending on fuel, ordnance, position, and aircraft systems.)	2.19 Not required.	2.19 Frequently required, but dependent on fuel, ordnance, crew fatigue, and aircraft systems.
	(See Note 21)		(See Note 21)
2.20 Return to Base	2.20 Knowledge of field conditions and alternatives is needed. This task is also dependent on aircraft systems, fuel, distances to the airfield, and other operations which might affect landings.	2.20 Same as CAS. If additional missions are flown from an alternate base, then consideration must be given to specialized flight planning aids.	2.20 Same as CAS. (See Note 22)

MOTES: (21) The ability to re-engage in combat after refueling is very desirable. While this will depend on remaining ordnance, it does eliminate the need for ground facilities and maintenance support.

Since an aircraft may have to divert at the last minute, information updates are required through the final landing.

(22)

FIGURE 4.18 IN-FLIGHT MISSION REQUIREMENTS (CONTINUED).

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
2.21 Descent	2.21 Requires planning for a specified approach or in conjunction with enroute radar descent. The key element is to save as much fuel as possible by trading altitude for airspeed.	2.21 Same as CAS.	2.21 Same as CAS.
2.22 Approach (All elements of aircraft maneuvering required to align the aircraft with the runway.)	2.22 This task requires maneuvering 2.22 Same as CAS, via instruments or radar vectors to align the aircraft with the landing runway. It is also necessary to maintain separation from other aircraft operations.	2.22 Same as CAS.	2.22 Same as CAS.
2.23 Landing (Actions re- quired to position air- craft safely on the	2.23 The final phase of flight requires precise manipulation of flight trajectory. It necessitates close adherence to commanded airspeeds, angles of attack, rates of descent, and flight path along the ground. Smooth	2.23 Same as CAS.	2.23 Same as CAS.

FIGURE 4.19 IN-FLIGHT MISSION REQUIREMENTS (CONCLUDED).

PHASE	CLOSE AIR SUPPORT	AIR INTERDICTION	COUNTER AIR
Post-Flight			
De-Arm	3.1 Maintenance crews safe remaining weapons and/or ejector cartridges, and advise aircrew of safety status. If an aircraft cannot be "safed," it normally will be shut down in a safe area and will not be taxied with other aircraft. Other systems requiring de-arming are flare and chaff ejectors. Procedures and requirements differ for guns, rockets, missiles,	3.1 Same as CAS.	3.1 Same as CAS.
Taxi	3.2 Same as 1.4.	3.2 Same as 1.4.	3.2 Same as 1.4.
System	3.3	3.3 Same as CAS.	3.3 Same as CAS.
ב הלאל אי	(a) Perform all system checks or self-tests required to evaluate the aircraft's condition for the next mission. These checks emphasize mission essential equipment.		
	(b) For damaged equipment, diagnosis is required so that maintenance crews can identify the problem.		

FIGURE 4.20 POST-FLIGHT MISSION REQUIREMENTS FOR CLOSE AIR SUPPORT, AIR INTERDICTION, AND COUNTER AIR.

PHASE	CLOSE ATR SUPPORT	AIR INTERDICTION	COUNTER AIR
Shutdown	3.4 Aircraft systems turned off and pilot exits aircraft. All personal equipment and debriefing aids are taken from the aircraft. Expended supplies are also removed.	3.4 Same as CAS.	3.4 Same as CAS.
Post-Flight	3.5 Aircrew examines aircraft for battle damage or other unusual conditions, such as lost panels, bird strikes, etc.	3.5 Same as CAS.	3.5 Same as CAS.
Debrief	3.6 Completion of all forms and documentation tasks. The key element is the ability to relay all useful information to maintenance, operations, intelligence and higher command units. Mission-related information includes damage assessment, location of enemy SAMs, troop concentrations, enemy tactics, etc.	3.6 Same as CAS.	3.6 Same as CAS.

FIGURE 4.21 POST-FLIGHT MISSION REQUIREMENTS (CONCLUDED).

**1 OCTOBER 1979** 

**MDC E2046** 

# . 5.0 INFORMATION NEEDS

It should be obvious, from even a cursory reading of the mission requirements, that a combat pilot must spend considerable time processing displayed information. In Section 3, we reviewed the various display units which will present visual information during the 1990s (see Figures 3.1 and 3.2). In this section, we are concerned with the specific types of information which must be displayed at different stages of each TACAIR mission. Therefore, we have provided comprehensive listings of information requirements. We believe that the mission and information requirements, coupled with the aircraft characteristics, determine the assignment and sequencing of pilot tasks.

The tabulations which appear in Figures 5.1 through 5.12 (adapted from Mills et al., 1978) indicate whether the displayed data are: required (1), frequently required (2), or merely of additional value (3) at a certain stage of the mission. Thus, an entry of "2" or "3" signifies that this particular information is less essential than information assigned a weight of "1."

			Γ		Prei	Flig	ght		7		-		_					_	_	ln·F	ligh	ıt	_	_				_	_		_	7	_	Post	·Flig	<u></u>
		1.2 Parison Plan	T. S. S. L. S.	Ten and Spain	1.5 Arm	1.6 7.4	2.1 C.		23 [2	(2.4 Paris)	2.5 C. markyous pract	2.6 . Condination AAR)	2.7 c Maission Remai	2.8 - Penetration	2.0 Incat Warm	2.10 Jetection	2.11 Courson	2.13 Gentification	2.13 Detion	2.14 Execution	2.15 Tulminan	2.16 commetion	2.17 Cores	2.38 5.38	12.19 Runnelsman	2.20 Barbary MARR	2.21 Selum to Ban	2.22 Smoons	223 198000	3 - Landing	3. D. L.m		3.4 Shiem Change	3.5 Parisonn	2.6 Call Page 1	
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Time Briefing Time Station Time Statis Engine Time Flight Check In Time Taxi Time	11111	1 1	!!!	1 1				<del></del>	<del>-</del>										·=··	-			- <del>-</del> -					<del></del>	_							
Takeoff Time Air Refueling Contact Time Time on Target Time to Climb Time to Go Control Times	1 1 2	2 2	2 2	2 2 2			1	2 1		1 2 1	1	1 1 1	- 1 1	2	1	1 1 1	1	1	۔ ۔ ۱ ا	 . ز	 נ.	2	2	1	<u>.</u>	1		- 1	i		_		-			
Landing Time Rendezvous Time Finance Times Time of Day  Caution and Warning Systems	1 1 1	2	2	2	2	2	1 2	1 2	1 2	1 1 2	1 1 2	1 2	1 1 2	1 2	1 1 2	1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 1 2	1 2	1	1 2	2	2	2	2	2	2	
Master Caution Configuration Warnings Operating Limitation Warning Primary Systems Fail Fuel Low	,	1	1 1 1 2	1 1 1 1 1 1	1 1 1 1 1	1	1 2 1 1	1 2 1 1	1 2 1 1	1 2 1 1	1 2 1 1 1 1	1 2 1 1 1	1 2 1 1 1 1	1 2 1 1 1	1 2 1 1 1 1	1 1 1	1 2 1 1 1	1 2 1 1	1 2 1 1	1 2 1 1	1 2 1 1	1 2 1 1 1 -	1 2 1 1	1 2 1 1	1 2 1 1 1	1 2 1 1	1 1 1	1 1 1 1	1111	1 1 1 2	1 1 2	1 1 2	1 1 2	1	1	
Oxygen Low Environmental Control System Warning Autopilot Disengaged Autonics Malfunctions Altitude Low Airspeed Low	,	1	1	1	1		2	2	1 2 2 2 2 2	2	1 1 2 2 2 2 2	1 1 2 2 2 2 2 2	1 1 1 1 1 1 1	1 1 1 1 1	1 1 1 1 1	1 1 1 1 1	1 1 1 1 1	1 1 1 1 1	1 1 1 1 1 1	1 1 1 1 1 1	1 1 1 1 1 1		1 1 2 2 2 2	1 1 2 3 2	1 1 1 1 1	1 1 2 2 2 2	1 1 2 1 1 1	2 1 2 1 1 1	2 1 2 1 1 1	3	3	1	1 2	3	1	

FIGURE 5.1 INFORMATION REQUIREMENTS FOR CLOSE AIR SUPPORT.

				_	Pref	ligh	nt.	7		_		_				_		i	Fhe	hi		_		_	_	_	_	_	_	$\overline{T}$	_	·ost·	Flight
	/	1.1 Maulon Pr.	1.3 C.	1.4 Finand Syle	1.5 Armer	16 Takeoff	2.1 Climb is	2.3 Comma Com Of 1.5	- Lane	2 E Mendezague	26 Condination and AAR!	2.3 Manion Res	2.8 - Martistion Marketinge	2.9 Threat Warming	2.10 . Merinan	211 Contract	2.12 De Delication	2.13 Ere	2.14 Assess	2 16 Terminales	2.17 Comme		(2.19 Run	2.20 Australia AAA	2.21 C. 10 Bee	2.22 Ac.	2.23 James	2) Calledon	32 % ALM		3.4 St. Chart.	3.5 Parie	3.6 Dabnet
Miscellaneous Fuel Off Loading Gross Weight Autopilot Submodes Aucraft Lighting 9			1 1	1	1 1	1/2	2	2	1	2				1					1 1			t			2	; 1			1	,		1	2
Communications Controlling Agency Auctail Call Sign Authentication FFF, SIF Clearances		)   	2 1	1 1 2 1	1 1 2 2	2 2	2 2	2 2	1 1 1	1 1 1 1 1	1 1 1 1	2 2 2	2 1	1 2	1 2			1	1 1 2 2	1 2	2 2	1 1 1 1 1 1	1 1 1 1	1211	2	2	1111	2	1	2	2	•	
Secure Communication Frequencies Interpretation (If Required) Navigation Aids Identifiers Mission Reports		_	1	1	1 1	,	1 2 2	1 2	1 2	1	1	T	1 2	1	1			1	i i 1 1	,	1 2 2	-i- i	1		1		,		_	2 2	2 2	<u>-</u>	1
Flight Aids Normal Checklists Emergency Checklists Procedures Auproach Aids Emergency Airfields Personal Techniques		1 1	1 2 1	1 2 1	1 1 2 2 1 1	2 2 2 1 2 3	. 1	1	1 2	3 2 1	3 2 1	3 2 1	3 2 1	3 2 1	3 2 1	1	1	1	3 3 2 2 1 1	[	3 2 1	1 2 1	2	2	2		1 2 1	1 2 1	2 2 1	1 2 1	1 2 1	1	1
Nevigation Course Heading Flight Path Pirch Steering Bank Steering			2 2 2	2	1			1 1	1 1 1 1	1 1 1		1 1 1 1 1	1 1 1 1 1	1 1 1 1 1		1 1 1 1	1 1 1	)       	1 1 1 1 1 1 1 1 1 1	1 2 1 2	1 1 1 1 1	1 t 1	1 1 1 1	111111	11111	1 1 1 1 1			2	1 2 2 2 2	-	•	1 2
Glide Stope Localizer Distance to Destination Bearing to Destination Destination Selected Marker Beacon			3 3	2 2 2 2	 3 2 1		1 1	1	1 1	1 1	1 1 1	1111	1111	1	1 1		1 1	1 1	1 1	1	1	1 1	~ - !	1	2 2 1 1 1	1		1	_	1222222	_		
Present Position Destination Offsets Magnetic Variation Coordinates Navigation Point Planned Route of Flight			-,- -,-	2 2 2 2 2 2	1			1	12 11	1 2		1 1 1	1	1	1 1		1	1		1		1 2		1		i 1				122211	_		
Holding Patterns Minimum Descent Allitude Missed Approach Point								1	Ť		1						_	_						3		1	211		_	_			

FIGURE 5.2 INFORMATION REQUIREMENTS FOR CLOSE AIR SUPPORT (CONTINUED).

			_		_		_		_															_	_			_		_	_				_
			L		Pre	Flo	ht		4			_	_						, "	·Flig	m		_					,		_	4		out	Fligh	<u>.                                    </u>
	12	1.2 o Flance	1.3 C. Heliphi	Pod Syne	1.5 Armin	1.6 Tate	2.1 0.1.5	22 C. To Lowing	2.3 Laure 104	(2.4 p.	2.5 C mder Pour	2.6 - Condinentian AAR!	2.7 P. Raman	2.8 T. STANDON	2.9 Careat Werning	2.10 Latinon	2.11 ld.	2.12 De De Callon	2.13 Ex.	2.14 Abanta	2.15 Terming	2.35 fg/mg	/ Ser. 2	ander sou	2.20 B PARES	2.21 C. millen 10 B.	2.22 A MIGAN!	2.23 Taploan	3.1 De A	32 74	3.5	3.4 Shine Charles	3.5 Post E.	3.6 Dabrios	/
Altitude Altitude Atlove Ground Level (AGL) Altitude Atlove Mean Sea Level Altitude Setting Command Altitude Target Elevation Minimum Enrouse Sale Altitude Terrain Altitude Terrain Clearance	3 2		1 1 1			2	2 1 1 1 2	3 1 1 1 1 1 1 1	1 1 1 1 1	1 1 1 2	2 1 1 1 1	1 1 1 1	1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1	1 1 2 1 1	1 2 1 1 1	1 1 2 1	1 2 2 1 1 1	1 2 1 1 1 1	1 1 1 1 2 2 1 2 1 1 1 1		1 1 1 2	3 1 1 1 2	1 1 2 1 1 1 1	3 1 1 1 1 2	1 1 1 1 1 1 1	1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			1 -	_	_		
Velocity Vertical Velocity Rotation Speed Takeoft Speed Check Speeds Climb Speert (Normal) Climb Speert (Maximum Performance) Maximum Runge Crusse Maximum Endurance Maximum Runge Descent Approach Speed	_					1 1 1 2 3	] 	1	1	1	1	-	1	-	1	1	1		-	1 1		1	1	1	2 2 2 2	2 2 2 2 2	1	1						1	
Landing Speed Maximum Safe Speed Minimum Controllable Speed Limitations Corner Speed True Anspeed	1 3	1	1	1	1	3	 2 1	3 · 1	21	1	1	- !	1	1	1.	1 3	1 3	- 1 3	2 1	1 1	1 2 3	-3 1	 1	 1 3	-	2 1 3			,	1	1	1	1	1	
Angle of Attack (AOA) Ground Speed Wind Velocity and Direction Turn Rate Best Rate of Climb Best Angle of Climb Mach Number	3_				 : :	3 1 2 2	3 -	2 - 2 - 2 - 2 - 2	<u>3</u>	3 2	3	3 3 2	1 2	$\frac{3}{3}$	3 3 2	3	3	3	2 2	3 3	L.	3	۔ ۔ ّ۔		2 2 2 2 2	2 2	3.22	3		2		<u> </u>	_		
Selected Speed Emergency Airspeeds Weapons Release Speeds IAS	3 2					2 2	2	2	2	1	2	7	2	2	2	2	2	2	2	1 2	2	2	1	2	7	2	5	2							
Systems Fuel Flow Fuel Remaining Fuel Required to Destination Fuel Management Hydraulic Electrical	  -  -	1 2 2	2 1 1 1 1 1 1	3 1 3 2	2 :	+	3 3	3 3	21133	211333	2 1 2 3 3	1 3 3 3	2113333	2 1 2 3 3 3 3	2 1 2 3 3 3 3 3	2 1 2 3 3 3 3	2 1 2 3 3 3 3 3	3 3 3	3 3 3	2 2 1 1 2 1 3 3 3 3	72	_3 _3	2 1 1 3 1 3 3	2 1 3 3 3 3	211333	1 3 3 3	211333	2	3 -	3 2 3 3 3 3 3	3 .	1 3 3 2 2 2 2		2 2 2 2 2 2	
Oxygen Engine		2	1			,			3	3	3	3	3	3	3	3	3			3 3 3 3			3	3	3	3	3	3		3		2	2	2 2	

FIGURE 5.3 INFORMATION REQUIREMENTS FOR CLOSE AIR SUPPORT (CONTINUED).

	Prof	light /		In-Flight		Post-Flight
	1.1 Naturan Pannang 1.2 Prefilipit 1.3 Suprand System Checks 1.5 Armana	2.1 Climb to Lowel Ory 2.2 Climb to Lowel Ory 2.3 Lotter 2.5 Consecutions	2.5 Coordinates 2.5 Massion Breakstone 2.8 Threat Name 2.9 Lives Name 2.10 Lives Name 2.10 Lives Name 3.10 Liv	2.11 16m11/16m10m 2.12 Denson 2.13 Ensuron 2.13 Australion 2.14 Australion 2.15 Gens 2.19 Chung (2.19 Renderron and AAR) 2.20 Australion	221 Disami 222 Apropos 223 Landing 31 Disami	J.J. System Checks 3.4 Shutdown 3.5 Past Figure 3.6 Cabrel
Penetration Aids Al Warning SAM Warning Threat Avoidance Disposables Status ECM Status ECM Tactics Mutual Support Special Threat	1 1 2 1 3	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1 2 1 1 1 1 1 1 2 1 1 1 1 1 1 1 2 1 1 1 1 1 1 1 2 1 1 1 1 1 1 1 1 2 1	2 3 3 2 3 3 3 3 3 3 3 3 2 3 3 2 2 2 2 2	2 1 - 2 - 1 1
Weapons Information Ballistics Weapons Envelope Weapons Ready Weapons Release Weapons Remaining Weapon Impact Weapon Selected Bumb Fall Timp/Impact Point Weapons Options Selected Weapons Detivery Salected		1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 1 1 2 2 2 2 3 1 3 1 1 1 1 3 3 3 2 2 2 2	1111	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
Air to Air Target Range Bearing Overtake Altitude Differential Target Turn Rate Target Attitude identification Target Altitude in Range Aim Point Breakaway	3	1 1 1 2 2 1 1 2 3	1 1 1 1 1 1 1 2 2 2 2 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		
Air to Ground Target Acquisition Target Range Target Bearing Positive Target ID Aim Point Break Away (Pullup)			1 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 1 2 1 1	1

FIGURE 5.4 INFORMATION REQUIREMENTS FOR CLOSE AIR SUPPORT (CONCLUDED).

			_	_	Pref				7-					_	_				Flig					_		_				7	_		1
			⊬			114	n (		∤		_		7			_		<del>,"</del>		PNI	_	7				_	_	_		╀	_	-	hght
		12 Pullan Planne	1.3 Similar	1.4 This and System	1.5 Armen	Token	21.0	2.2 Church Coming		William Server	NA PROPERTY OF	2.7 P. Handen Rena	2.8 "enetretion	2.9 Threat Warm	2.10 Jetherlon	211 1600 mg	2.12 Contification	2.13 Exe.	2.14 Asimilar	2 Termines	215 Epina	(2.18.0. C. Mark	WANTED TO SOUTH OF A SAN	2.20 Re.	2.21 Day 10 Bas	2.22 Apr.	2.23 / 7000	3.1 De interna	3.2 To	33.5	3.4 Share Chart	3.5 Post Flant	S Charlet
Advanistrative Furmation Call Signs Flight Position Aucralt Assignment Parking Spot Spare Procedures Aucralt Configuration Frequences			2 1 2 -	2 2 1	2 1 1 1 3 2 1 1 1 2 1			2 1 - 2 - 2	1 1 2		1 1 2 1	2 1 2 2 2		1	1		,	·ı-	2 1 1	21	2 1	1		2	2 :	ī ī -	2	1 1 1	1	2 1 1 2 2 2	1 1	1 1 1 1 2 1	
Frequencies IFF/SIF Procedures Weather Regulations Airheld Status	i	2	2 1	2	2 1		1 3	2 2 1	,		121	1	2	2	2		2	2	2 2	1 2	2	1		1	1	1	2	-		_	1	1 1	
Airfield Description Landring Runway Runway Length Barriers Approachs Assed Approach Instructions Airfield Elevation	12111	·	-		1 1 1 2 1 2 1 2 2 2 2 2						-					i i -			_					2	2 2 1 2	1 1 1 1	1 1 1 1 1 1	1		1		1 	
Decision Height Parking Area Taxi Routes Arming Area Dearming Area Atternate Airfield	1	1	•	1			2 :	2	2		2	2	2	2	2	2	2	2	2 2	,	2	2			2	i	2		1 1 2 2	1 ;	1	<u> </u>	
Time Briefing Time Station Time Statt Engine Time Flight Check In Time Taxi Time	1	1	! !	1												~			_									_	_				
Takeoff Time Air Refueling Contact Time Lime on Target Time to Climb Lime to Go Control Times		2 2			1 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	?		2 2 1 1	; 2 1		1	1 1 L	2 1 1	2 1	2	2 1	1 1 1		2 1 1 1 1	1 2		1		1	1	L	1					1	
Lamling Time Rendezvous Time Enroute Times Time of Day	1	1			1 1	1	1	2 1 1	1		1111	1	1	1	1	1	;		1 1	ļ١	1	1		2		1	1	1		1	,	1 1	
Caution and Warning Systems Master Caution Configuration Warnings Operating Eimitetion Warning Primary Systems Fail Fuel Low Oxygen Low Environmental Control System Warning Autophot Disengaged Aviour's Malfunctions Altitude Low Auspeed Low	-	! 1			1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	2	1	1 1 1	1 1 1 1 1 1 1 2	_	1 - 1 - 1 - 1 - 1	1 1 1 1 1 1 1 1 1 1	1111111111	1 1 1 1 1 1 1 1 1 1		1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1	ì	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	i	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		1 1 1 1 1 1 1 1 1	1 1 1 1	F - 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		1 1 2 3 1 3	1 1 2 2 1 1 1	1 1 2 1	1 1 1 1	

FIGURE 5.5 INFORMATION REQUIREMENTS FOR AIR INTERDICTION.

			$\Gamma$		Prof	- luci	ht	7			-		_		_			10	ı-Fh	-ht	-				_				_	7			Flight	
	<u>/-</u>	1.2 Parison Plans	7.3 Series	7.4 To The System	3			2.2 Cruse Com Of	Z. R. R. L.	WK H WELDON	12.6 M	2.7 P. Rende	2.8 T.	2.9 Carat Warning	2.10 1 metion	2.11 16 Callon	2.12 Dentification	Τ	L'estion.	£ ;	2.16 Epress	(2 % Change	William Sendersons and As	2.20 Ret	2.27 Out to Bas	2.22 A.	2.23 The Day	3.1 Canding	32 T. A.m.	<i>†</i>	34 Shirt Call			
Miscellaneous Fuel Off Loading Gross Weight Autopilot Submodes Aucraft Lighting 9	\ 	•	1 2 1	1	i	,	1 1	1	1		1			1					1 1		1	1	<u>- 13</u>		1		,			,	-	2 2 2		
Communications Controlling Agency Aircraft Call Sign Authentication IFF-SIF Clearances	1 1 1 1 1 1		1 1 2 2 1	1 1 2 3	2 1 2 1 3 2 3 2 1	2	1 1 2 2 2 2 2 2 2 2 1 2	2	1 1 2 1		1 1 2 2 1	2 2 2 2 2 2	1 1 2 2 2 2	2 2 2 2 2 2	2 2 2 2 2 2	2 2 2 2 2	2	2			2 2	2			1 2 2 2 2 2	1 2 2 2 2 2	1 2 2 2 1	1 2 3	1 2 3	2 2 3 2	2	1		
Secure Communication Frequencies Intercom (If Required) Navigation Aids Identifiers Mission Reports	1		2	2 2 2 2 2	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2		2 2 2 2 2 2 2 2 2 2 2	? ?	2 2 2 2 2			2	2 2 2 2	2	2 2 2 2	2	2 2 2 2	2	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	2 2	? 2	2 2 2		2	5 2 2 2	22222	2222	2	2 2 2		2			
Flight Aids Normal Checklists Emergency Checklists Procedures Approach Aids Emergency Air helds Personal Techniques		1 2 1	-	3	1 1 2 2 1 1 3 2 1 1		1 1 2 2 1 1 3 2 2	<u>!</u>	1 2 1		1 2 1 2 1	1 2 1 2 1	1 2 1 2 1		1 2 1 2 1	1 2 1 2 1	1 2	1 2	1 1 2 2	2   2	2 2 2 1 1	1 2		3 2	1 2 1 2 2 1		1 2 1 2 1	1 2 1	1 2 1	1 2 1	1	1 2 1		
Navigation Course Heading Flight Path Pitch Steering Bank Steering	,		2 1 1 1 1 1	2	             		1 1		1 1 1		1 1	1 1 1	1 1 1 1 1 1	1 1 1 1 1 1	1 1 1 1 1 1 1	1 1 1 1 1 1	1 1 1 1	1 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1					- 1	1 1 1 1	1	11111		2	1 2 2		1		
Glue Stope Localizer Distance to Destination Bearing to Destination Destination Selected Marker Beacon			1 1 1		3 2 1		1		1		1 1 1	2 1 1	1 1 1	1 1 1	1111		2 2 2	!	1 1		2 1	1 1			1 1 1 1 2	1			_	722222				
Present Position Destination Offsets Magnetic Variation Condinates Navigation Point Plannel Route of Flight Holding Patterns Minimum Descent Attitude Missert Approach Point	3 3		1 1 1 1 1 1 1		1 _ <u>1</u>	֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓	2 2 1 1 1 1 1 1		1 1 1		1 1 1 1 1 1 1 1 1	1 1 1 1 1	1 1 1 1	-	1 1 1	1 2 1	i t	1	1 1	,	1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	<u> </u>	1 1 1 2 3	1 1 2 2 2 2 2 2 2	1 -	1 1 2 1 1	_	_	1 2 2 1 1 1 1	-	1		

FIGURE 5.6 INFORMATION REQUIREMENTS FOR AIR INTERDICTION (CONTINUED).

			1	_	Pro	Fh	gh I	_	7		_		_	-	_			•	n Fi	i gh l					_			_		7	_	Past	·Flegh
	/=	1.3 Mandon Plans	1.3 C. T. Shipper	1.4 Start and Suc.	1.5 Act	1.6 T.	2.1 C. C.	23 Climb to Le	William St.	William Bare and Bare	(2.6 M	2.7 Pasion Rende	2.8 T. metration	2.9 C.	2.10 1 10chon	2.11 County	2.12 Contification	2.13 E Children	2.14 Assembles	2.15 Torn	2.16 Form			2.35 MANNERS OF BARRY	4	23. Dalant	22. Apr 86.5	3 - Landing	3.5 De Arm		3. Spitem Change	3.5 Puridonm	3.6 Debrief
Altitude Altitude Above Ground Level (AGL) Altitude Above Mean Sea Level Altimeter Setting Command Altitude Targel Elevation Minimum Erroute Sale Altitude	3 2	<del></del> -	1 1 1			2 2	1	3 1 1 1	3 1 1 - <del>1</del>		2 1 1	1	1 1	i L			1 1 1 1	1 2	1	;	1 :	2 3	3    -	3 1 1 1	1111	1 1 1 1	1 1 1			1			
Terrain Altitude Terrain Clearance	١,		1					1 2	1 2		i	1	1	1	;	1	1	1	1		1			2	i	i	1			1			_[
Velocity Vertical Velocity Rotation Speed Takeoff Speed Check Speeds Climb Speed (Normal) Climb Speed (Maximum Performance)					!	1 1 1 2 3	1	1	1		)	,	,	1	,	1	1	1	1		?	1		-	1	-	1		_	_	_		
Climb Speed (Maximum Performance) Maximum Range Cruise Maximum Endurance Maximum Range Descent Approach Speed Landing Speed	-				;	3	-	2 3													•			2 2 2	2 2 2	1 2	1						
Maximum Sale Speed Minimum Controllable Speed Limitations Corner Speed True Airspeed	3	,	,	1	1	,	2	2 1 3	1 3	1			1	1	١	1	3	1	3 :	3 3	}   1 	1 1		3	2	1	1		,	,	1	,	7
Angle of Attack (AOA) Ground Speed Wind Velocity and Direction Turn Rate Best Rate of Climb Best Angle of Climb	2			2	<del>3</del> -	1 2	- 2	2 2	3	-d.	-	2	<u>-</u>	2	2	2	2	_		2	_			2 2	2	312		2	_	_	-	· <u> </u>	1
Mach Number Selected Speed Emergency Airspeeds Weapons Release Speeds 1AS	3 2					2	3.	3 1 2	<u>. 3</u> .	- ;	<b>L</b> :	2	<u>3</u> _	3; 2	2	3 2	3_ 1 2	3 1	3; 1	2	3_3	1		3 2	2 2	1 2	1 2	ŀ	_	_	_	. <del></del>	1
Systems Fuel Flow Fuel Remaining Fuel Required to Destination Fuel Management Hydraulic	1 1 3	2	1 1 1	2 2 2 2 2 2	2 2	2 2 2 1 2	2	3 2 1 3 3	3 1 1 1			3 1 1 3 3	3 1 1 3 3	3 1 1 3 3	3 1 1 3 3 3	3 1 1 3 3 3	3 1 1 3 3	3 1 1 3 3			2 3 1 1 1 1 2 3 2 3		l 	3 1 1 3 3	3 1 1 3 3	3 1 1 2 3	3 1 1 2 3	3 2 3 3	3 2 3 3	3 2 3 3	1 1 3 2	2	1 2 2 2
Electrical Oxygen Engine		2	1	3	2	2 2 1	3 - 3	3 3	<del>3</del> - 3 3		3 3	3 3		3	3 3 3	3	3 3 3	3	3	3	2 3	1 3		3 3	333	3 3	3	333	3 3	3 3	2 2	3 5	2 2 2 2

FIGURE 5.7 INFORMATION REQUIREMENTS FOR AIR INTERDICTION (CONTINUED).

				<del></del>	· - · · · · · · · · · · · · · · · · · ·
	<u> </u>	Flight	····	in-Flight	Past-Flight
	1.2 Manus Panning 1.2 Profit pr 1.3 Sura and Stream Cauge 1.5 Tau	2.2.2.0 0 m to 1	(15 km) (15 km	2.2.2 Parell Par	
Penetranen Aids At Warning SAM Warning Threat Avoidance Disposables Status ECM Status ECM Tactics Mutual Support Special Threat	2,	2 2		1 1 1 1 1 1 2 2 3 1 1 1 1 1 2 2 3 1 1 1 1	2 2 2 1
Weapons Information Ballistics Weapons Envelope Weapons Release Weapons Remaining Weapons Impact Weapon Selected Burnt Fall Yime/Impact Point Weapons Dotions Selected Weapons Dotions Deliceted Weapons Delivery Selected		2 2 3 2 2 2 1 2 2 2 2 2 2 2 2		1 1 1 1 1 2 3 2 1 1 1 1 1 1 2 2 2 1 1 1 1	
Aur to Air Target Rainge Beering Overtake Affitude Differential Target Turn Raire Target Affitude Identification Target Affitude In Range Aim Point Breakaway		1 1 1 2 2 1 1 2 3 1	2 -1	2	
Air to Ground Target Adquisition Target Range Target Bearing Positive Target ID Aim Puint Bresk Away (Pultup)			1 1 1 1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 1 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1

FIGURE 5.8 INFORMATION REQUIREMENTS FOR AIR INTERDICTION (CONCLUDED).

	PreFlight In-Flight	Post-Flight
	1.1 Massion Pannang 1.2 Pathippi 1.3 Start and System Chacks 1.5 Anning 2.5 Anning 2.5 Ciuse 2.5 Ciuse 2.6 Massion Rendestrous 2.8 Threst Warning 2.9 Description 2.10 Location 2.12 Location 2.13 Easterdon 2.13 Easterdon 2.14 Easterdon 2.15 Ciuse 2.16 Easterdon 2.17 Ciuse 2.18 Easterdon 2.18 Easterdon 2.18 Easterdon 2.19 Easterdon 2.10 Easterdon 2.10 Easterdon 2.11 Ciuse 2.12 Anningon 2.13 Ciuse 2.15 Ciuse 2.15 Ciuse 2.16 Easterdon 2.17 Ciuse 2.18 Easterdon 2.18 Easterdon 2.18 Easterdon 2.19 Easterdon 2.10 Easte	2.23 Aspenses 3.13 Landing 3.2 Tou. Arm 3.4 Shutton 3.5 Post Figure 3.6 Debrief
Administrative Formation Call Signs Flight Position Aircraft Assignment Parking Spot Spare Procedures		2 2 2 2 2 1 1 1 2 2 1 1 1 1 1 1 1 1 1
Aircraft Configuration Frequencies IFF/SIF Procedures Weather Regulations	$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
Airfield Status Airfield Description Landing Runway Runway Length Barries Approaches Missed Approach Instructions	1 2 2 1 2 1 2 1 2 1 2 1	1 1 1 2 1
Arrheld Elevation Decision Height Parking Area Taxi Routes Arrining Area Dearning Area Alternate Airfield	1 1 2 1 2 1 2 1 1 1 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1
Time Briefing Time Station Time Static Engine Time Flight Check In Time Taxi Time	1 1 1 1 1 1 1 1 1 1 1 1 1 1	
Takeoff Time Air Refueling Contact Time Time on Target Time to Climb Time to Go Control Times	1 2 2 2 2 2 2 1 1 1 1 1 1 1 1 1 1 1 1 1	1
Landing Time Rendezvous Time Enroute Times Time of Day	1 2 2 2 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 1 1 1 1 1
Caution and Warning Systems Master Caution Configuration Warnings Operating Limitation Warning Prinary Systems Fail Fuel Low		1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
Oxygen Low Environmental Control System Warning Autopilot Disengaged Avanus Malfunctions Altitude Low Anspeed Low		2 3 3 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1

FIGURE 5.9 INFORMATION REQUIREMENTS FOR COUNTER AIR.

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			F		Pro	rk	i de		+		_			_			_		7	a-F	n dy	•	_	,	_				,			4	'	****	Fligh	<u>-</u>
	/	1.1 Mission Pin	13 C Preflight Many	Te Stort and S.	1.5 Aug.	7.6 T.	2.1 C. C.	2.3 Climb to Louis	23 Cruite 196	(2.4 p	2.5 Candervon	2.6 Condination	Z Million Ran	Z.B Pertellion	2 o Threat Warn		211 Common	2.13 Samufication	2.73 Canson in	2.74 Execution	2.15 T.	2.16 commence	2.17 C.	2.18 P. P.	2.19 Real Book and A.	2.20 Paris	2.21 C. Illin to But	2.22 Mamr	2.23 Page 8.55	3.1 Challing	32 S. A.M.	,	34 Shirth Chapter	3.5 Part F	3.6 Debrief	
Miscellaneous Fuel Off Loading Gross Weight Autopilot Submodes Aircraft Lighting g		l I	1 1 1	2		1 2 2	1 2	1 2		1 1 2	1 2	1 2	1 2	1 2	1 2	1 2	1 2	1 2	1 2	1 2	1 2		1	1	1	,	2	1		2	2	1 2	2		2	
Communitations Controlling Agency Aurorat Call Sign Authentication IF F/SIF Clearances			1 2 2 1 2	1 2 2 2	2	1 2 2 2 1	2 2 2 2 2	2 2 2 2 2 2	2 2 2 2 2 2	1 1 1 2 1	11121	1 1 2 1	2 2 1 2 2	1 1 1 2 2	2 2 1 2 2	2 2 1 2 2 2	2 2 1 2 2	2 2 1 2 2	2 2 1 2 2 2	2 2 1 2 2	2 2 1 2 2	2 2 1 2 2	2 2 2 2 2 2	1 1 1 2 1	1	2 2 2 2 2 2 2	2 2 2 2 2 2 2	2 2 2 2 2 2 2	2 2 2 1	2 2	2 2					
Secure Communication Frequencies Intercom (It Required) Navigation Aids Identifiers Mission Reports	,		2 1 2 2	2 1 2 2	1 2	2 1 2 2	1	2 1 2 2	2 1 2 2	1 1 2 2	21122	1 1 2 2	2 1 2 2	2 1 2 2	2 1 2 2	2 1 2 2	22122	2 1 2 2	2 1 2 2	2 1 2 2	2 1 2 2	2 1 2 2	2 1 2 2	2 1 1 2 2	2	2 1 2 2	2 1 2 2	2 1 2 2	2 1 2 2		1 2		_			
Flight Aids Normal Checklists Emergency Checklists Procedures Approach Aids Emergency Autholds Personal Techniques		1 1 2 1 1	1 2 1	1 2 1	1	2		1 2 1 2 1	1 2 1 2 1	1 2 1 2 1	1 2 1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1 2 1	1 2 1 2 1	1		1 2		1 2 1 2 2	1 2 1 1 2	1 2 1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	1 2 1	•	
Navigation Course Heading Flight Path Pitch Steering Bank Steering	,	_	1 1 1 1 1 1 1 1 1	2	1	,	1	1 1 1	1 1 1 1 1	1 1 1 1 1	11111	1 1 1 1	<u>-</u> 11111	1 1 1 1 1	1 1 1 1	1 1 1	1111	1 1 1	1 1 1 1 1	1 1 1 1 1	1 1 1 1 1	1 1 1	1 1 1	1 1 1 1 1	1	;	1 1 1 1 1	1 1 1 1 1	1 1 1 1 1	2	2	1 2 2 2 2	<u> </u>		1 2 2	
Glide Stope Localizer Distance to Destination Bearing to Destination Destination Selected Marker Beacon		· ·	1 1 1 1 1 1 1	_		2	3	3	3 1	3	3	3	3 1 1	3 1 1	3 1	3 1	3	3 1 1	3 !	3	3 1 1	3	3	3	1	i]	1 3 1 1 1	1 1 1 1	3 1 1 1	3 2	3 2	2 2 3 1 2 2				
Present Position Destination Offsets Magnetic Variation Crondinates Navigation Point Planned House of Flight			1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		;	2	3	1	1	T 2 3 1 1	2 3 1 1	3	3 1 1	1 1 1	3 1 1	3	3 1 1	3	1 1 1 -1 -	3	3			3	3	1	3	3	3 1 1	3	3	T22322	_	_	,	
Holding Patterns Minimum Descent Attitude Missed Approach Point	2 2 2 2	?				2	2 2	_		_								_				_	_			3 [	2 2 2	1	1 1			_	_			

FIGURE 5.10 INFORMATION REQUIREMENTS FOR COUNTER AIR (CONTINUED).

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	1.1 Million E.	1.3 Sterlight	1.4 Tau System Checks	Takanga	2.1 Climb to !	160 mg 2mg 27	C.4 Pers	2.5 Coordinate and Add.	2.6 Million P	2.8 - Penetration	2.9 Charat Warmen	2.10 Louisethon	2.11 1000	2.12 Decuination	2.13 Execution	2.15 T. Albertoment	2.16 - inmation	217 Com	2.18 Render	2 3 Renger and AAR	2 3. Return to B.	ق	7 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 -	3.7 60411	3.2 Cm. Alm		3.4 Sh. Check	3.5 Patrician	Debrief
Altitude Altitude Above Ground Level (AGL) Altitude Above Mean Sea Level Altitude Setting Command Altitude Target Elevation Minimum Entoute Safe Altitude Terrain Altitude Terrain Clearance	3 1 1 3 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1	2 1 2 1 1	2	1 2 1 2	2 1 2 1 2 1	3 1 2 1 2 1 2 1 2 1	2 2 1 1 2 2 1 1 2 2 1 1 1 2 2 1 1 1	2 1 2 1 2	1 2 1 2 1	2 1 2 1 1 1	2 1 2 1 1 1 1	2 2 1 1 1 2 2 1 1 1 1 1 1 1 1 1 1 1 1 1	2 1 1 2 2 1 1 1 1 1 1 1	2 1 2 1 2 1	2 1 2 1 1 2 1	1 1 2 1 2 1 2 1	2 1 1 2 1 1 2 2 2 2	3 1 1 1 2 2 1 1 1 1	3 1 1 1 2 1	1111	1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			1		1	
Valocity Vertical Velocity Botation Speed Takentt Speed Unick Speed Unick Speed Climb Speed (Normal) Climb Speed (Maximum Performance) Maximum Range Grouse ritaximum Endurance Maximum Range Descent Approach Speed Uniting Speed			<del>-  </del>	1	1 2 2 2 2 2 2	1 3 2 2 2 2	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	1 1 2 2 2 2 2 2 2 2 2 2 2	2 2	2_	2	2 2 2 2 2	2 2 2 2 2 2 2 2 2	2 2 2 2 2 2	2 2 2 2 2 2	1 2222	2 2 2 2 2 2	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	2 2 2	2 2 2 2	1 33 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	1 3322221	1 3322211	-	_	_	_		
Coming speed Maximum Sale Speed Min mum Controllable Speed Corner Speed Truc Airspeed Ar ple of Attack (AOA) Ground Speed Wind Velocity and Direction Tarn Bate Best Bate of Climb			$-\frac{3}{2}$ 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 2 3 1 1 2 1	2 2 1 2 3 2 2 2 2 2	Jan-23 Janes	2 1 2 3 3 2 2 2 2 2 2 2	]	2 1 2 3 2 2 2	2 1 2 3 2 2 2 2 2	3 2 2	2	2 2 2 1 1 1 2 2 2 3 3 3 2 2 2 2 2 2 2 2	2 2 2 2 2 2	2 1 2 3 2 2 2 2	2 2 1 2 3 2 2 2 2 2 2	2 1 2 3 .2 2 2 2	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	1 2 2 3 2 2 2 2	3 2 2 2	3 2223	1 1 3 3 2 1 2 3	1 3 3 2 1 2 3		<del>-</del>	_	_	1	
Best Angle of Climb Mein Number Sciented Speed Emergency Airspeeds Urspeeds Urspeeds Urspeeds Systems			i <del>i</del> i	3 1 2 2 2	3 1 2 2 2	331222	3 1 1 2 2	3 3 3 3 1 1 2 2 2 2	3 3 1 2	3 3 2 2 2 2	2	ī	2 2 3 3 1 1 2 2 1 1 2 2	2 1 3 1 2 1 2	3 3 1 2 1 2	3 1 2 1 2	3 3 1 2	3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	3	3 1 2 2	3 3 1 2 2	3 1 2	3 1 2		_	_	_		
Foot Flow Forth Remaining Fuel Required to Destination Fuel Management Hydraulin Electrical Oxygen Engine	2 2 2 2 2		2 2 2 2 2 1 1 1 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	311333333	3113333	31133333	3 3	3 3 1 1 1 1 3 3 3 3 3 3 3 3 3 3 3 3	3	3 1 3 3 3 3	3 3	3	3 3 1 1 1 1 3 3 3 3 3 3	3	3 1 3 3 3 3			3 3 1 1 1 3 3 3 3 3 3 3 3 3 3	3		1103333	3113333	3 3 3 3 3	30000000	30000	2222222	2 2 1 1 2 1	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	

FIGURE 5.11 INFORMATION REQUIREMENTS FOR COUNTER AIR (CONTINUED).

	Preflight /																																	
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	1.1 Man.	12 Profitation	Start and Sur.	1.5 Armine	Taleor		Come Come of	Louise (2.6.)	2 E Renderen	25 Coordination AAR	(2) Mission Br	2 e Penetration	28 Threat Way	2.10 Caternon	2.11 ( Gallon	2.12 Cantification	2.13 Challen	2.14 Execution	2.15 T. Manument	2.16 E manabon	2.17 Comme	2.18 Par.	2.19 Reservoir and a.s.	2.20 Relieur	221 OF 12.	222 Amr	2.23 E. See	31 Children	3.2 Te.	\	3.4 Spilling Change	J.S. Par. C.	3.6 Charles	
Penatration Aids All Warning SAM Warning Threat Avoidance Disposables Status ECM Status ECM Tactics Mutual Support Special Timeat		1 1 1 2 2 2 2	2	2 2 2 2 2 2 2 2 2 2	3 3 3 3 3	2 2 3 2 2 2 2	2223222	2221221	22227212	2222212	1 1 2 1 1 1	111111	1111111	1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1	11111111	1 1		:	1 1 2 1 1 1	2 2 2 2 2 2 2 2 2 2	2 2 2 1 2 2 1 2 1	1	2 2 2 3 2 3 2	3 :	3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	33313223	)	2 2 2	1 1 1 1 1	2 2 2	Ļ	1	
Weapons Information Ballistics Weapons Envelope Weapons Reday Weapons Release Weapons Remaining Weapons Impact Weapon Selected Bomb Fall Time/Impact Point Weapons Options Selected Weapons Options Selected Weapons Optiors Selected	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	2		1 2 2 2 1 2 1 2 2 2 1	332	322333	2222222	2 1 2 2 2 2 1 1	222222222	22222222	1 1 2 1 1 1 1	1 2 1 1 1 1 1 1	1 1 1 1 1	1 2 1 1 1 1 1 1	1 2 1 1 1 1 1	1 2 1 1 1 1 1	1 1 1 1 1 1 1 1	1 1 1 1 1 1	11211111	2 1	2 2	2 1 2 2 2 2 1 1	2	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2		3 -	3 3 3 2	2 2 2 2 2 2 2	2	2 2 3	2		1	
Aurito Air Torget Range Blearing Overtake Altitude Differential Target Turn Rate Target Attitude Identification Target Attitude In Range Aim Point Breakaway	2						_	1111111111	1		222222222222	1 1 1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1		1 1 1 1 1 1 1 1 1 1	1111111111	1 1 1 1 1 1 1 1 1 1 1 1	1 1 7 1 1 1 1 1 1 1		_	; ; ; ; ;	: : : : : : :								_		J	
Air-to-Ground Target Acquisition Target Range Target Bearing Positive Target ID Aim Point Break Away (Pullup)																																		

FIGURE 5.12 INFORMATION REQUIREMENTS FOR COUNTER AIR (CONCLUDED).

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# 6.0 PILOT PERFORMANCE AND BIOCYBERNETIC APPLICATIONS

The preceding sections have provided a framework for discussing pilot tasks. As we noted earlier, the assignment and sequencing of tasks are dependent upon the aircraft characteristics, the tactical objectives, and the information requirements.

Subsection 6.1 attempts to convey the dynamic, time-varying nature of pilot tasks. Subsection 6.2 then describes the biological signals which reflect the pilot's status or which may permit a direct coupling of the pilot with aircraft subsystems. Finally, Subsection 6.3 examines the manner in which biocybernetic techniques can be expected to improve pilot performance and thus enhance weapons system effectiveness.

# 6.1 TIME LINE ANALYSES OF PILOT TASKS

To avoid the redundancy of presenting detailed task listings for each mission type, we have generated time lines for those tasks commonly associated with segments of a carrier launched escort mission. Figure 6.1 shows that most of the mission segments integrate several of the general mission requirements illustrated previously in Figure 3.4. Further, we have included both air-to-air and air-to-ground scenarios, as evident from the representative mission profile depicted in Figure 6.2.

MISSION SEGMENT	MISSION REQUIREMENT	MISSION SEGMENT	MISSION REQUIREMENT
Launch	1.6 Takeoff	Air-to-Ground (A/G) Strike	2.5 Coordination 2.7 Penetration
Climb	2.1 Climb to Level-Off	Strike	2.8 Threat Warning 2.9 Detection
Rendezvous	2.2 Cruise 2.5 Coordination 2.6 Mission Rendezvous		2.10 Location 2.11 Identification 2.12 Decision 2.13 Execution
Ingress	2.7 Penetration 2.8 Threat Warning		2.14 Assessment 2.15 Termination
Medium Range Intercept (MRI)	2.5 Coordination 2.8 Threat Warning 2.9 Detection	Egress	2.16 Egress 2.5 Coordination 2.8 Threat Warning
	2.10 Location 2.11 Identification 2.12 Decision 2.13 Execution	In-Flight Refuel	2.17 Cruise 2.18 Rendezvous and AAR
	2.14 Assessment 2.15 Termination	Marshal	2.20 Return to Base 2.21 Descent
Surface-to-Air-Missile (SAM) Avoidance	2.5 Coordination 2.8 Threat Warning	Prelanding	2.22 Approach
	2.10 Location 2.12 Decision 2.13 Execution	Landing	2.23 Landing
Air Combat Manuevering (ACM)	2.5 Coordination 2.8 Threat Warning 2.9 Detection 2.10 Location 2.11 Identification 2.12 Decision 2.13 Execution 2.14 Assessment 2.15 Termination		

FIGURE 6.1 SEGMENTS OF A CARRIER LAUNCHED ESCORT MISSION. (THE FIGURE SHOWS THE RELATIONSHIPS BETWEEN SEGMENTS AND MISSION REQUIREMENTS PRESENTED PREVIOUSLY IN FIGURE 3.4. SOME OF THE SEGMENTS (E.G., MRI) ARE COMPRISED OF SEVERAL MISSION REQUIREMENTS.)

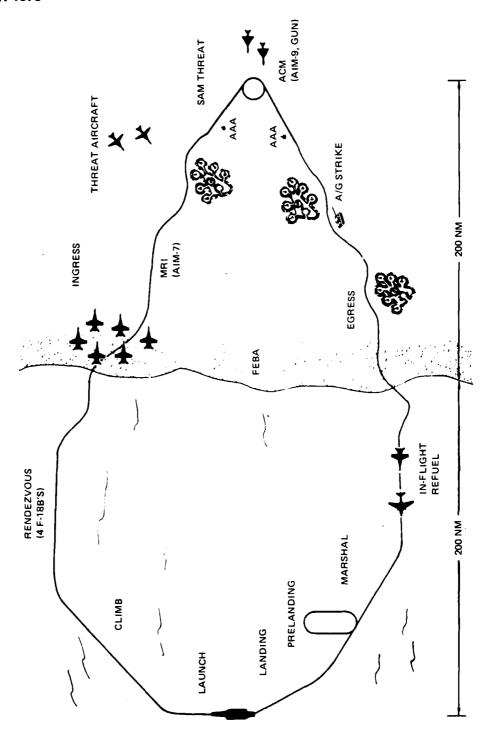


FIGURE 6.2 REPRESENTATIVE PROFILE FOR A CARRIER LAUNCHED ESCORT MISSION.

In the construction of dynamic task flows (Figures 6.5 through 6.17), we have incorporated existing documentation on the multimission F/A - 18 aircraft (Wise and Asiala, 1977). The crew station of this aircraft (see Figure 6.3) has been designed for both air-to-air and air-to-ground modes, and it features many of the avionics advances highlighted in Section 3.

We stated in the Introduction that we would restrict the application of biocybernetic techniques to those pilot tasks which are very difficult, are critical to the success of the mission, or occur during periods of heavy workload. In order to eliminate less essential tasks from later consideration (more inclusive listings of tasks appear in the time line analyses), military pilots familiar with the F/A - 18 crew station and with the projected mission requirements rated each task on the basis of difficulty and criticality. The following factors were taken into account when judging task difficulty:

- o amount of information that must be processed,
- o degree to which the relevant cues are discernible,
- o number of control actions and precision with which they must be performed.
- o time available to perform the task (or a cluster of related tasks),
- o dependence upon an integrative process in reaching a decision,
- o other variables (such as stress, fear, or fatigue).

While the relation of the task to the success of the mission was emphasized in estimating criticality, the pilots also indicated whether performance of the task affected flight safety.

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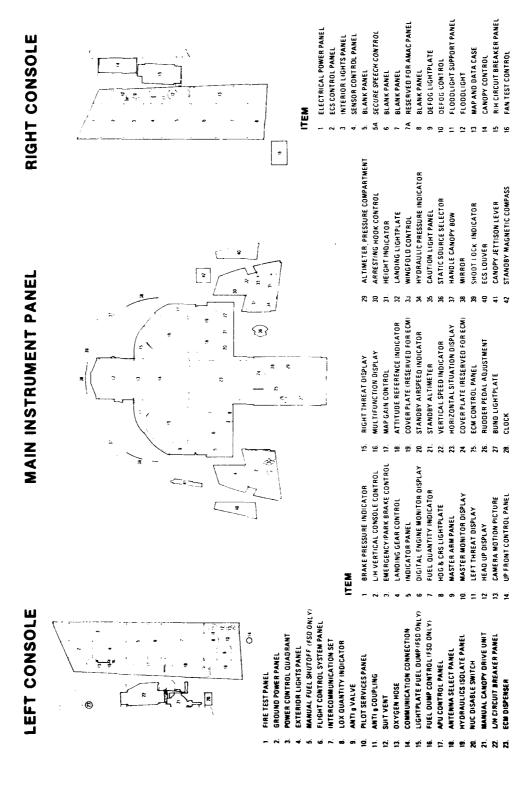


FIGURE 6.3 SCHEMATIC CONFIGURATION OF F/A 18 CREW STATION.

**1 OCTOBER 1979** 

**MDC E2046** 

We have defined workload in terms of the number of tasks a pilot must perform in an arbitrary period of time. For each mission segment, we counted the number of individual tasks, including repetitions of the same task, which are presumed to occur in a 5 second interval (see Figure 6.4). Tasks which continue in successive intervals were counted for both time periods. Descriptive statistics were derived from these tabulations. Across the entire mission, mean "task load" per 5 second interval was equal to 3.93 tasks, with a standard deviation of 1.77 tasks. High task load intervals were assumed to have task frequencies greater than 7.47 (two standard deviations above the mean). The tasks which appear in these intervals and the tasks assigned high ratings with regard to difficulty or criticality comprise the row headings of the biocybernetic matrices in Subsection 6.3.

The time lines which follow are meant to be illustrative, since the duration of any mission segment will change dramatically as conditions change. To demonstrate the process of creating dynamic task flows, we have presented complete time lines for launch, surface-to-air missile (SAM) avoidance, air combat maneuvering (ACM), and landing. For the remainder of the mission segments, time lines are provided for only the high task load intervals. In general, the task listings should be read from bottom to top. Further, the reader may find it helpful to refer again to the relevant mission and information requirements described in Sections 4 and 5, respectively. A brief narrative introduces each mission segment.

6.88 IS THE AVERAGE TASK LOADING PER FIVE SECOND INTERVAL THROUGH 6.17), THIS TABLE SUMMARIZES THE NUMBER OF TASKS WHICH MUST BE PERFORMED ACROSS ALL FIVE SECOND INTERVALS. THUS, FOR AIR COMBAT MANEUVERING, THE ENTRY OF "19" IN THE TASKS COLUMN LABELED "6" INDICATES THAT 6 TASKS MUST BE PERFORMED DURING 19 OF THE 60 (TOTAL) FIVE SECOND INTERVALS. FOR THIS SEGMENT, THE ARE ASSIGNED TO THE PILOT DURING EACH SEGMENT (SEE FIGURES 6.5 (FROM THE TIME LINE ANALYSES OF TASKS WHICH ESCORT MISSION.

FIGURE 6.4

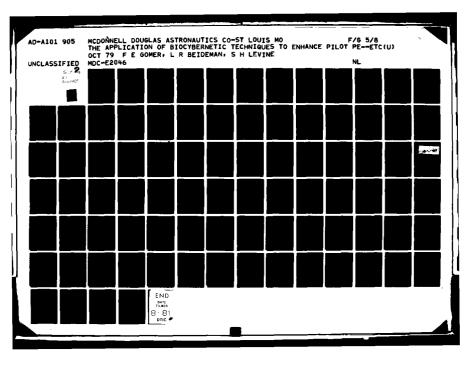
**MDC E2046** 

	I×	7.83	3.04	2.90	4.80	2.94	5.00	6.88	4.22	3.45	3.55	3.56	5.58	2.65	3.93
9	12+	0	0	0	0	-	0	0	0	0	0	0	0	0	1
PERI	12	_	0	0	0	0	0	က	0	0	0	0	0	0	4
ONC	11	0	0	0	0	0	-	7	0	0	0	0	0	0	3
SECC	10	-	0	_	0	0	0	-	0	0	0	0	_	0	4
A FIVE SECOND PERIOD	6	2	0	-	0	0	-	4	0	0	_	0	7	0	11
	<b>∞</b>	3	7	0	-	-	0	9	0	0	_	_	ო	2	20
	7	2	0	_	_	7	0	14	_	0	ည	4	7	1	38
JRM	9	2	_	_	9	_	_	19	4	_	9	∞	∞	1	29
TASKS PERFORMED IN	D.	0	9	7	4	7	4	9	7	4	78	6	4	2	28
KS P	4	1	ω.	6	ო	7	0	4	9	4	59	14	10	7	125
TAS	က	0	16	28	ო	က	7	-	9	ω	39	11	4	13	134
A OF	2	0	11	14	7	7	-	0	ო	വ	30	12	<u>-</u>	15	96
NUMBER	1	0	6	12	0	17	-	0	0	0	13	1	,0	15	78
Ž	0	0	0	_	0	0	0	0	0	0	7	0	0	-	4
	MISSION SEGMENT	LAUNCH	CLIMB	RENDEZVOUS	INGRESS	MEDIUM RANGE INTERCEPT	SAM AVOIDANCE	AIR COMBAT MANEUVERING	A/G STRIKE	EGRESS	IN-FLIGHT REFUEL	MARSHAL	PRELANDING	LANDING	TOTAL

**1 OCTOBER 1979** 

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6.1.1 <u>Launch</u> - Within this 60 second segment, the pilot catapults from the carrier, establishes and maintains proper attitude for climb, and retracts the landing gear and flaps for normal flight. Throughout the segment, critical flight data are processed and attitude adjustments are performed. Visual search of surrounding airspace is required to monitor the position of the other escort aircraft (assuming a dual launch).



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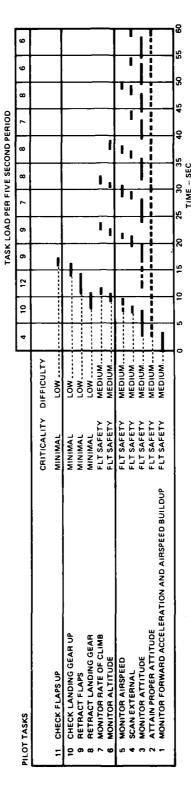


FIGURE 6.5 DYNAMIC TASK FLOWS FOR LAUNCH.

**1 OCTOBER 1979** 

**MDC E2046** 

6.1.2 <u>Climb</u> - The lead pilot joins with his wingman, and they climb to their assigned altitude. Both pilots are in constant communication with the air traffic controller and with each other. Climb and level-off checks are frequent, as is the monitoring of other flight parameters. Once clear of the carrier, the pilots establish proper spacing between their aircraft and then engage the Automatic Flight Control System (AFCS). The flight monitoring tasks continue throughout the mission segment. Additionally, the Identification-Friend or Foe (IFF) system is activated, and preliminary subsystems checks are performed.

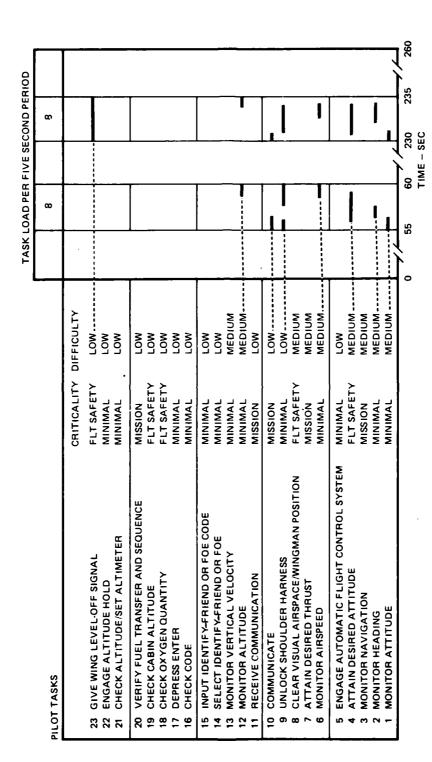


FIGURE 6.6 DYNAMIC TASK FLOWS FOR CLIMB.

**1 OCTOBER 1979** 

**MDC E2046** 

6.1.3 <u>Rendezvous</u> - The escort pilots are principally concerned with joining the strike force. Radar parameters are selected (to assist in the location process) and then a system check is performed. The pilot also must achieve UHF communication with the strike force. Since most of the flying is conducted under AFCS, task load is relatively low. The pilot continuously monitors displayed flight data and the position of the other escort aircraft. Once the strike force is joined, the pilot disengages the AFCS to begin formation flying.

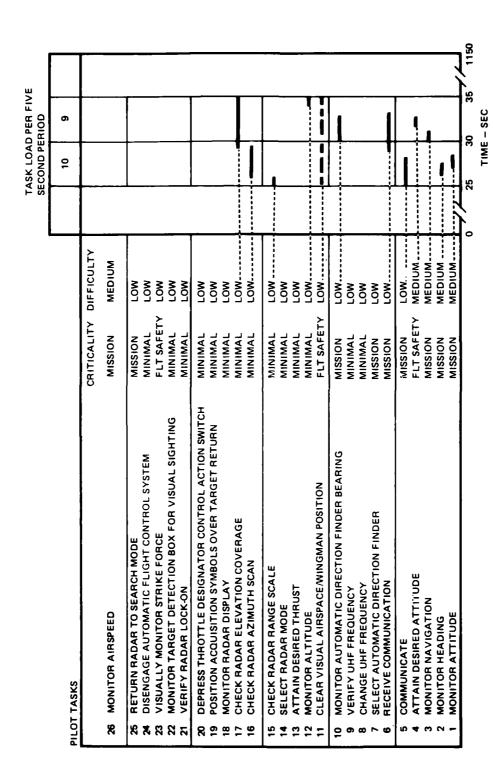


FIGURE 6.7 DYNAMIC TASK FLOWS FOR RENDEZVOUS.

**1 OCTOBER 1979** 

**MDC E2046** 

6.1.4 <u>Ingress</u> - This segment commences as the aircraft cross the forward edge of the battle area (FEBA). Counter-threat and armament subsystems are set and checked, and the pilot flies a level weave maneuver until the strike force reaches the target or until ingress is interrupted. To reduce the possibility of detection by enemy ground forces, the radar altimeter is deactivated.

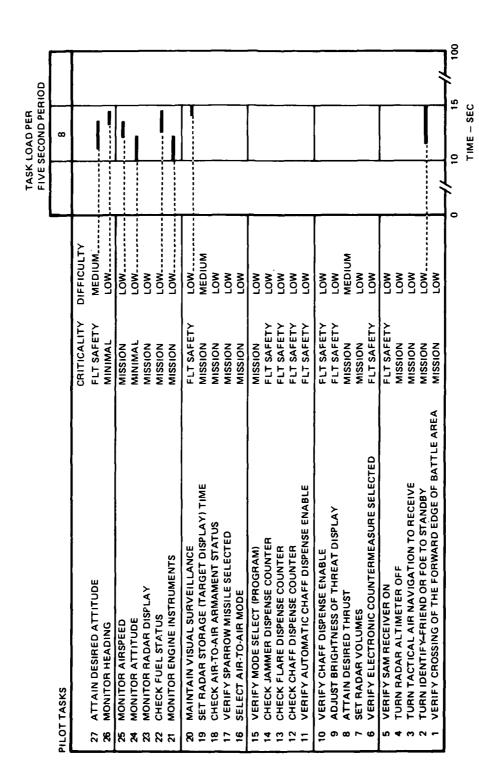


FIGURE 6.8 DYNAMIC TASK FLOWS FOR INGRESS.

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- 6.1.5 <u>Medium Range Intercept (MRI)</u> Should enemy fighters be encountered, the escort aircraft must defend the strike force. Therefore, fire control functions have the highest priority within this segment. Pilot responsibilities include:
  - o configuring the aircraft subsystems for the appropriate attack mode,
  - o selecting azimuth and elevation coordinates for radar antenna,
  - o monitoring radar display and IFF evaluator,
  - o arming and assessing status of selected missile,
  - o achieving target lock-on,
  - o maneuvering so that the enemy aircraft is positioned within the missile launch envelope,
  - o firing the missile,
  - o observing flight path of the missile.

The outcome of the engagement is communicated, the master arm is set to safe, and navigation data are processed. The pilot then will proceed to rejoin the strike force.

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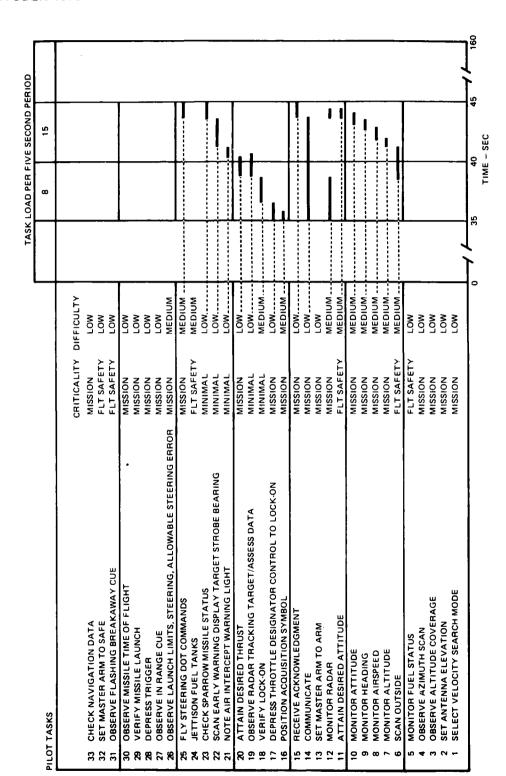


FIGURE 6.9 DYNAMIC TASK FLOWS FOR MEDIUM RANGE INTERCEPT (MRI).

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**MDC E2046** 

6.1.6 <u>Surface-to-Air-Missile (SAM) Avoidance</u> - Should a warning of possible SAM launch be displayed to the pilot, he must determine the azimuth of the SAM site and inform the other pilots of the direction of the threat. If a SAM launch is confirmed, the pilot must begin to search visually for the missile. He also will dispense chaff and flares to decoy the missile or will initiate jamming to confuse the missile's radar. If the missile cannot be detonated or avoided in this manner, the pilot must perform evasive flight maneuvers. He will track the smoke trail of the missile so that he may time the necessary abrupt changes in altitude, velocity, and direction.

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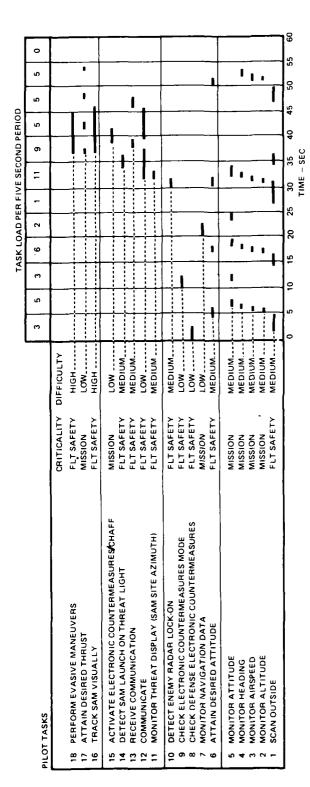


FIGURE 6.10 DYNAMIC TASK FLOWS FOR SAM AVOIDANCE.

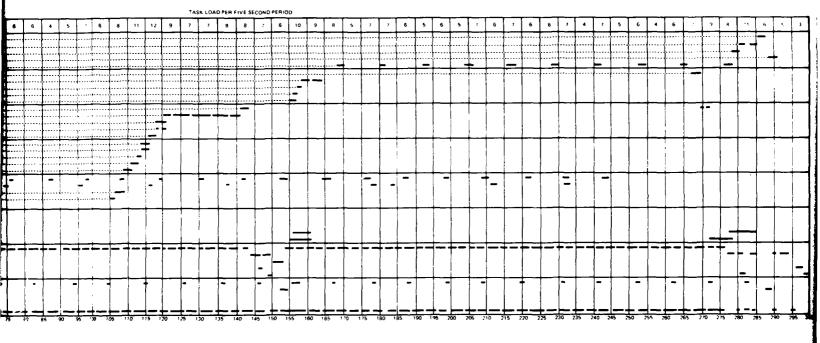
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**MDC E2046** 

6.1.7 <u>Air Combat Maneuvering (ACM)</u> - When "beyond visual range" threats are first encountered, long or medium range missiles (Subsection 6.1.5) are deployed. Should these weapons prove ineffective (due to enemy maneuvering or an inability to achieve lock-on) or should enemy aircraft successfully avoid radar defenses, then close-in combat will result. Consequently, within this segment we are concerned with those tasks related to either AIM-9 (Sidewinder) missile or gun attack. It should be noted that crucial flight and weapons data are displayed on the HUD in order to maximize the time available to the pilot for visually tracking the enemy aircraft. A complicating factor is the high g environment in which the tasks must be performed, since violent manuevering is required to properly align the weapons with the target and to position the target within the missile launch or gun envelope.

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40 DETECT STRIKE FORCE RADAR	MISSION	ι <b>0₩</b> -		····				·····	<b>∤</b>	· · · · · ·			·	·	·	•	4		<b>{······</b>			<b>4</b>	
39 SELECT NAVIGATION MODE	MISSION	.OW		·				<b></b>				····	<b></b>				. •		ļ		• • • • • • •	ļ	
38 DETERMINE NEW HEADING TO STRIKE FORCE	MISSION	MEDIUM .		}					•	{		ł·			*****	4		·	<u> </u>	ļ	4	÷ · · · ·	,
37 SET WASTER ARM TO SAFE	FLTSAFETY	:OW				••••	}		ļ	<b>(</b>		• • • • • • •	<b>}</b> -		•				<b> </b>		· • • • • • • • • • • • • • • • • • • •		
36 CHECK ANGLE OF ATTACK	FLTSAFETY	LOW							<u> </u>							<u> </u>	<u></u>		1		4		1
35 OBSERVE RETICLE OVER TARGET	MISSION	MEDIUM		.}	ļ		· · · · · ·	ļ		<b>{</b>	<b></b> .	ļ	<b>}</b>	ļ	<b></b>	ļ	<b></b>	·	-}	ļ	ļ		
J4 OPTIMIZE RETICLE DYNAMICS	MISS:ON	MEDIUM								····	<b> </b>	i	<b>{</b>	·	d		· • · · · · ·	÷		<b></b>		· •	
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32 OBSERVE GUN SYMBOLOGY	MISSION	. OW	ļ	ļ	ļ								····	ļ	ļ	·}	• • • • • • •		ļ	·	į		
31 SELECT GUN MODE	MISSION	LOW					*****		<u> </u>					4			4		4				
30 VISUALLY ASSESS MISSILE SUCCESS	MISSION	HIGH		ļ	· · · · · ·		ļ		ļ	ļ			ļ	<b>∤</b> -	}	d	<b></b>	<b></b>	ļ			·, · · · ·	
29 VISUALLY MONITOR MISSILE FLIGHT	MISSION	MEDIUM		<b></b>		<i>.</i>	h	<b>∤</b> .	h	ł		·	ļ	<b>∤</b> -	<b>}</b>	4		4	· · · · · · ·	<b> </b>	4	•	
28 VISUALLY VERIFY MISSILE LAUNCH	MISSION	LOW		<b></b>	•		} · · · · ·			· · · · · ·	} · · · · ·		· · · · · ·	į	<b></b> .	·	·	4		ļ	4		
27 DEPRESS TRIGGER	MISSION	LOW		ł	}				ł			<b>}</b>	·	·		. <del>.</del>	· <b></b> .	·} · · · · ·	<b>∤</b>		•		
26 OBSERVE IN RANGE CUE:SHOOT LIGHT	MISSION	MEDIUM							<u> </u>				<u> </u>						d		4		
75 HEAR MISSILE TONE CHIRP	MISSION	MEDIUM		Į	Į				<b>∤</b> .	<b>}</b>		ļ	ļ	ļi		4		Į	ļ	. <b>.</b>		Ţ.,	
24 MAINTAIN STEER DOT WITHIN ALLOWABLE STEERING ERROR	MISSION	HIGH			<b></b>				ļ	<b>}</b>		<b>∤</b> -	ļ	<b></b>	ļ	ļ	4	<b></b>	<u> </u>	<b>{</b> -		4	• • • • • • •
23 DEPRESS UNCAGE SWITCH	MISSION	WEDIUM .		<b></b>						Į			{		·	· · · · · ·			·		*****		
22 HEAR MISSILE TONE ABOVE THRESHOLD	MISSION	MEDIUM .			{·					<b></b> .	·			•	<b></b>	ļ	4		ł		• · · · ·		•
21 OBSERVE TARGET DETECTION BOX-SIDEWINDER CIRCLE ALIGNMENT	MISSION	MEDIUM .															4			••••	1	· · · · ·	·
20 PUSH RUDDER PEDALS	MISSION	LOW			ļ					1	-			ė .	,	i <del>-</del>	1	-		-	ï	· -	
19 EXTEND RETRACT SPEED BRAKE .	MISSION	10W		Į	} · · · · ·					Į.		Į.	} -	4	1	ì	ì	+	]	1		-	
18 OBSERVE TARGET DETECTION BOX OVER TARGET	MISSION	WEDIUM		à	<b>}</b>				}				ł	ļ	·		· <del> </del>		d	· · · · ·	4		,
17 OBSERVE RADAR LOCK-ON	MISSION	MEDIUM.			ļ	¦				d	·			ļ		4		4	· • • • • • •	<b></b>		4	• • • • • •
16 SELECT VISUAL ACQUISITION MODE, HACQ, VACQ, BORESIGHT	MISSION	10₩,			ļ				<u> </u>	_				<u>i                                    </u>	<u> </u>		1		1				
15 OBSERVE SIDEWINDER SYMBOLOGY STATUS	MISSION	LO <b>W</b>							} · •	1		1	I	1		1		I		I	Ĭ.		
14 SET MASTER ARM TO ARM	MISSION	(OW)			ļ				+	ł			i	1	l	1	i	1		1			
13 SELECT SIDEWINDER MODE	MISSION	: OW			ļ				Į.	i				1		ı		ţ		1			
12 RECEIVE COMMUNICATION	MISSION	. DW		ļ					<b>}</b>			┢		<del>-</del>	-	1		1	1	1	1		
11 COMMUNICATE	MISSION	; OW		<b></b>							L		i	<u> </u>		1	1	<u></u>	i .		1		
10 TRACK TARGET VISUALLY	FLTSAFETY	ысн		ļ						<b>⊢</b> –		_		-	-	- +	<del>-</del> -		-	+-		<del>-</del> -	<del></del>
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5 ATTAIN DESIRED THRUST	WISSION	MEDIUM		Ţ	-				-		•		-		-		T -					1	
4 CHECK FUEL STATUS BINGO SETTING	FLTSAFETY	LOW		l —	į .				l			1	l	1	ł		1	1	i	Ì	i	1	
3 CHECK ENGINE PARAMETERS	MINIMAL	, ow	l	⊢	; 1				J	[ ]		l	ĺ		1	Į	1	1		1	i		
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FIGURE 6.11 DYNAMIC TASK FLOWS FOR AIR COMBAT MANEUVERING (ACM).



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6.1.8 <u>Air-to-Ground (A/G) Strike</u> - Escort aircraft may participate in attack of ground targets, especially during Air Interdiction. We have assumed low altitude penetration and that the location of the targets is stored within the TACAN computer prior to the mission. Moreover, we have assumed delivery of ballistic weapons and, given consent, that the actual time of release is computed automatically. The pilot must select the appropriate armament program and determine that he is in the vicinity of the target area before he initiates "pop-up." When he reaches target-unmask altitude and acquires the target visually, the pilot will maneuver the aircraft until the target is observed to be within the HUD field-of-view. He then will depress the "pickle" (weapons release) button and continue to track the target until the flashing symbology indicates release of the bomb. This cue also signals pull-up, and the pilot subsequently will establish heading for egress from the target area.

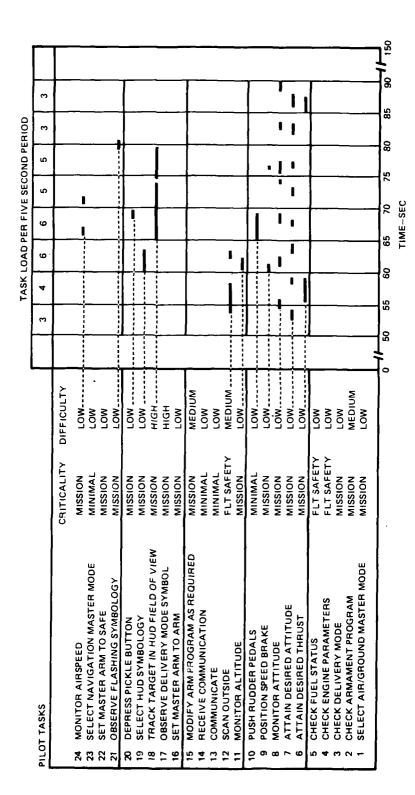


FIGURE 6.12 DYNAMIC TASK FLOWS FOR AIR-TO-GROUND STRIKE.

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6.1.9 Egress - This segment commences as the aircraft leave the target area. The pilot frequently processes flight data, adjusts flight controls, and monitors surrounding airspace. Further, counter-threat and amament subsystems are reset and rechecked.

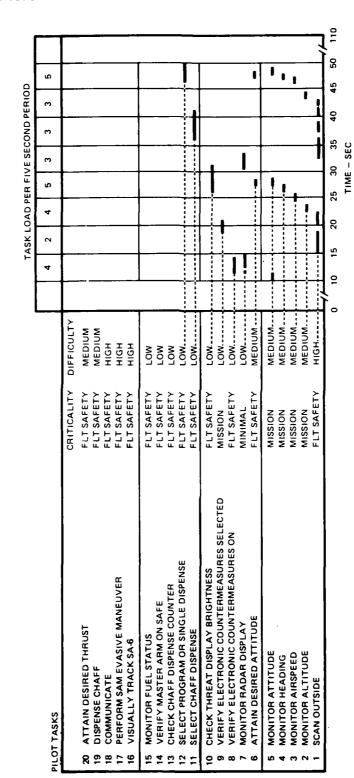


FIGURE 6.13 DYNAMIC TASK FLOWS FOR EGRESS.

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6.1.10 <u>In-Flight Refuel</u> - While in cruise, the pilot will select altitude, heading and attitude coordinates for the Automatic Flight Control System (AFCS). He also will designate the approximate location of the tanker aircraft and will adjust radar parameters for a more precise determination of the refueling site. Once lock-on is achieved, the pilot will deactivate the AFCS to maintain close formation during approach to the tanker fleet. Rendezvous typically will occur at 20,000 ft., at least 50 nm from the FEBA. The remainder of the tasks within this segment are associated with the actual refueling operation (e.g., extension of the probe, activation of the refueling switches, and retraction of the probe).

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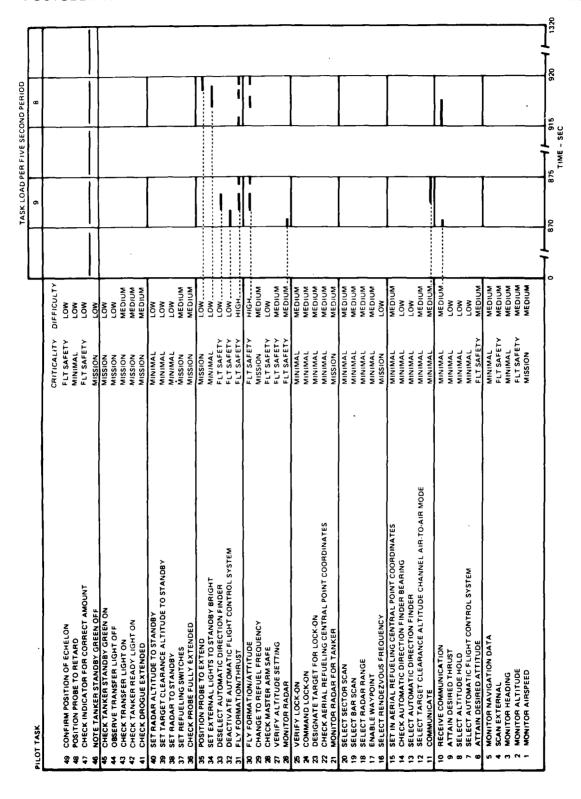


FIGURE 6.14 DYNAMIC TASK FLOWS FOR IN-FLIGHT REFUELING.

1 OCTOBER 1979

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6.1.11 <u>Marshal</u> - Prior to approach for carrier landing, the escort and strike aircraft will break formation and will descend to establish a holding pattern until clearance is received. While in the pattern, the pilot will engage the AFCS in order to perform various housekeeping checks and to configure the subsystems for approach and landing. He also will monitor his position in the pattern to assure proper in-flight alignment (IFA) during approach. In addition, IFF codes are again verified to maximize classification of "beyond visual range" threats. Finally, the Instrument Landing System (ILS) is set to standby and the proper heading is selected.

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ILOT TASKS				8	
	CRITICALITY	DIFFICULTY			<u> </u>
5 CHECK MULTIMODE DISPLAY FOR AUTOMATIC CARRIER LANDING STATUS	FLT SAFETY	row			
34 VERIFY DATALINK FREQUENCY	MISSION	row			_
33 SELECT HEADING	MISSION	row			
	MISSION	LOW			_
31 SELECT INSTRUMENT LANDING SYSTEM ON STANDBY ATTITUDE DIRECTOR INDICATOR	MISSION	LOW			
D VERIFY INSTRUMENT LANDING SYSTEM CHANNEL	MISSION	row			
	MISSION	LOW			
	MISSION	LOW			
	MISSION	LOW			_
3 VERIFY IDENTIFY-FRIEND OR FOE CODE	MISSION	LOW			
S ENTER IDENTIFY-FRIEND OR FOE DIGITS	MISSION	LOW			_
24 SELECT IDENTIFY-FRIEND OR FOE	MISSION	LOW			
	MINIMAL	LOW			_
22 CHECK FUEL QUANTITY	MISSION	NO.			_
CHECK ENGINE INSTRUMENTS	MISSION	LOW	_		
20 POSITION SPEEDBRAKES	MISSION	MOT			
19 ADJUST CABIN AIR AS REQUIRED	MISSION	row			
	MISSION	LOW			_
17 SELECT WAYPOINT	MISSION	LOW			
16 ACTIVATE RADAR ALTITUDE	MISSION	row			,
15 DEPRESS ENTER	MISSION	LOW			
14 SELECT TRANSMIT/RECEIVE AND TACTICAL AIR NAVIGATION	MINIMAL	row			_
	MISSION	MEDIUM			
12 MONITOR ALTITUDE	FLT SAFETY	MEDIUM.		-	
1 RECEIVE COMMUNICATION	MISSION	LOW			_
10 COMMUNICATE	MISSION	row			
9 SELECT TACTICAL AIR NAVIGATION	MINIMAL	LOW			_
B SCAN OUTSIDE	FLT SAFETY	MEDIUM	-		_
7 ATTAIN DESIRED THRUST	MISSION	MEDIUM		•	
6 MONITOR AIRSPEED	MISSION	MEDIUM			_
5 ENGAGE AUTOMATIC FLIGHT CONTROL SYSTEM	MINIMAL	row	_		
	FLT SAFETY	MEDIUM		{ !	
	MISSION	MEDIUM	-	1	
2 MONITOR HEADING	MINIMAL	MEDIUM		1.	_
MONITOR ATTITUDE	FLT SAFETY	MEDIOM			_

FIGURE 6.15 DYNAMIC TASK FLOWS FOR MARSHAL.

- 6.1.12 <u>Prelanding</u> During approach, the pilot's principal objectives are to arrive at the outer marker:
  - o in the assigned time slot,
  - o at the appropriate altitude, attitude, heading and airspeed.

Therefore, housekeeping functions receive the highest priority within this segment.

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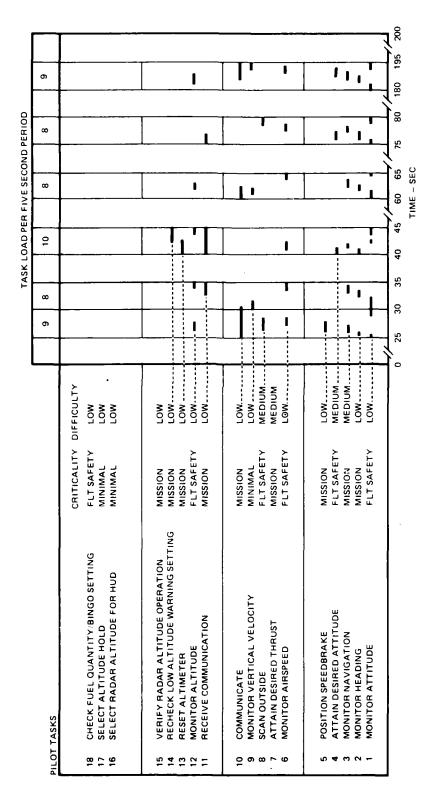


FIGURE 6.16 DYNAMIC TASK FLOWS FOR PRELANDING.

**1 OCTOBER 1979** 

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6.1.13 Landing - The tasks in this segment are performed during a fully coupled approach. A carrier-based system munitors relevant flight parameters (attitude, altitude, heading, glide slope, vertical velocity, and airspeed) and, via data link, automatically adjusts the aircraft's flight controls to compensate either for deviations from commanded levels or for changes in the orientation of the flight deck. Essentially, the pilot must process displayed flight data and must remain prepared to assume manual control if a system discrepancy is detected. He also must extend the gear, flaps and arresting hook. Just before touchdown occurs, the pilot must visually track the glide path and observe the altitude indicator (meatball) at the end of the runway. At touchdown, the pilot must apply thrust until the arresting hook catches.

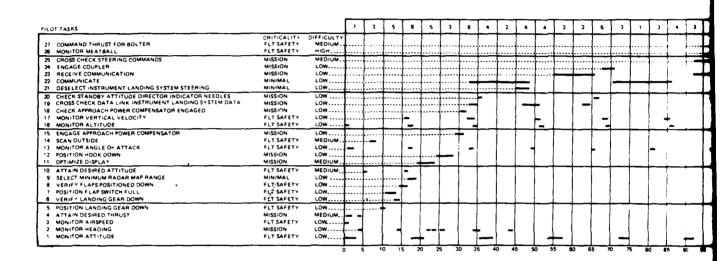
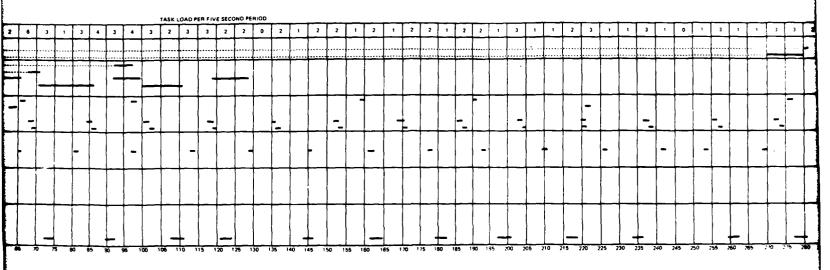


FIGURE 6.17 DYNAMIC TASK FLOWS FOR LANDING.

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## 6.2 REAL-TIME ANALYSIS AND INTERPRETATION OF BIOLOGICAL SIGNALS

Obviously, the pilot is extensively involved in monitoring displays and operating controls (flight controls and various subsystem controls). Overall mission success will be compromised when individual objectives cannot be achieved because the inherent requirements for processing information and initiating control actions exceed the capabilities of the pilot. We stated earlier that programmable, electronic displays and multipurpose keyboards offer one approach to limiting and sequencing display presentations according to the pilot's information needs at a given stage of the mission. By restricting presentations to essential flight, subsystem, and target parameters, the possibility that extraneous visual events will vie for the pilot's attention is minimized. Similarly, by integrating critical subsystem (e.g., weapons) controls directly into the stick and throttle, the speed of response is increased and the associated expenditure of physical effort is reduced.

These and other engineering advances are certainly important. However, we believe that the pilot's effectiveness could be further enhanced if (a) his current status as a processor of information and as a decision-maker were monitored, and (b) he were coupled more directly with the aircraft subsystems from a control standpoint.

We noted in the Introduction of this report that the program of biocybernetics research sponsored by DARPA has attempted to develop a communication channel for biological signals elicited during different mental activities. It is assumed that through real-time analysis and interpretation of these signals the computer will be able to determine (within certain limits):

- o when visual or auditory information has not been processed,
- o when the pilot is inattentive,
- o when the pilot is task-loaded to the extent that he is unable to perform additional duties,
- o when the pilot lacks confidence in a decision he has made.

Subsection 6.3 outlines the courses of action which may be taken should it be necessary to unburden or assist the pilot. They include, among others:

- o redistributing task responsibilities by effecting greater automation of certain housekeeping functions,
- o reducing the complexity of or "decluttering" information displays, especially the HUD,
- o cueing the pilot to attend to critical flight, weapons, and target data,
- o displaying adaptive decision aids which present weighted recommendations for mission-related decision strategies, particularly with respect to fire control functions,
- o furnishing remedial "checklists,"
- o optimizing the physical characteristics (e.g., contrast, focus, etc.) of imagery and symbolic presentations.

We also will discuss "thought" commands and eye position sensing as two means of supplementing manually operated or voice actuated control systems.

In order to justify the recording (noninvasive and unobtrusive) of biological data, there are certain <u>a priori</u> conditions which must be satisfied. According to Donchin (in press), it first must be shown that clearly

### **1 OCTOBER 1979**

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delineated and unique patterns of biological activity are, in fact, associated with specific mental (cognitive) functions; the signals cannot be ambiguous. Moreover, the signals must be interpretable in real-time as the pilot performs his tasks. Progress has been made in defining patterns of biological activity that are related to the efficacy of various cognitive functions (cf. Thatcher and John, 1977; John, 1977). Further, very significant improvements have been reported regarding the procedures used to extract (from noise) and classify these "messages" in real-time (John et al., 1978). These advances notwithstanding, a great deal more must be accomplished (in computer technology, software development, and the design of physiological monitoring equipment) before it is both <u>feasible</u> and <u>practical</u> to implement biocybernetic techniques in dynamic, operational environments. Nonetheless, we assume that the necessary breakthroughs will continue to occur.

Our intent in this subsection is to briefly review the biological signals which are most relevant for our purposes. Figure 6.18 distinguishes the two categories of signals we have considered, that is, brain electrical activity and peripheral activity. Since our principal concern is with monitoring brain function, the term "peripheral" denotes biological activity in response systems other than the central nervous system. The figure also lists two types of signals within each category, and, at a more molecular level, presents distinct informational features or components of two of the four types.

Of necessity, we have omitted many important details of the basic research findings in electrophysiology and biological signal processing which provide a foundation for the applications we have suggested. Therefore, we

	PERIPHERAL ACTIVITY	OCULAR ACTIVITY	EYE EYE MOVEMENTS POSITION
		PSYCHO- PHYSIOLOGICAL RESPONSES	
NALS	İ		CONTINGENT NEGATIVE VARIATION
CATEGORIES OF BIOLOGICAL SIGNALS		ITIALS	ENDOGENOUS COMPONENTS TON READINESS TAL POTENTAL
		EVENT-RELATED POTENTIALS	ENDOGE COMPON DETECTION POTENTIAL
		EVEN	P300
	W ELECTRICAL ACTIVITY		EXOGENOUS
	BRAIN ELECTR	APHIC	
		ELECTROENCEPHALOGRAPHIC ACTIVITY	
		ELECTA	

FIGURE 6.18 BIOLOGICAL SIGNALS CONSIDERED AS INPUTS FOR BIOCYBERNETIC APPLICATIONS.

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encourage the reader to consult the <u>Proceedings of the DARPA Conference on Biocybernetic Applications for Military Systems</u> (Gomer, in press) and the many excellent textbooks and chapters (cf. Chapman, 1973; Desmedt, 1979; John and Schwartz, 1978; McCallum and Knott, 1973, 1976; Otto, 1979; Thompson and Patterson, 1974) for more complete descriptions of the progress that has been made in these areas.

6.2.1 <u>Brain Electrical Activity</u> - In order to evaluate brain function while a crew member performs a demanding task, we must rely upon electrophysiological techniques to observe the underlying interactions within and between populations of cortical cells. Donchin (in press), John (1977), and Thatcher and John (1977) have suggested that cognitive operations can be conceptualized in terms of coherent patterns of activity within distributed cell groups. And importantly, orderly neural behavior gives rise to rhythmic voltage fluctuations in the two types of scalp recorded electrical activity identified in Figure 6.18.

Electroencephalographic (EEG) activity consists of spontaneous or on-going voltage fluctuations (see Figure 6.19). Event-related potentials (ERPs), on the other hand, are transient voltage fluctuations which are associated with a critical inducing event (i.e., a sensory stimulus or a cognitive operation) and which are imbedded in the EEG activity. Since the EEG is generally more pronounced in amplitude, it usually obscures the waveform of the ERP. However, there are several strategies for extracting and measuring ERPs (or selected "components") in real-time (John et al., 1978).

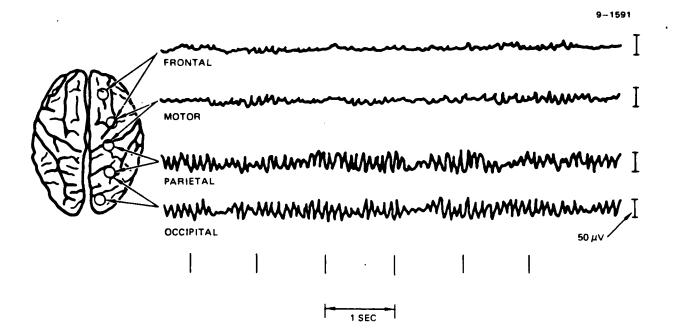


FIGURE 6.19 ELECTROENCEPHALOGRAM OF A NORMAL HUMAN ADULT. (ALPHA WAVES AT ABOUT 10 PER SECOND PREDOMINATE IN ALL REGIONS BUT ARE LARGEST POSTERIORLY. ONLY A FEW SMALLER AND FASTER BETA WAVES ARE VISIBLE IN THE ANTERIOR REGIONS. (FROM LINDSLEY, 1948))

If a discrete visual stimulus is presented to an observer who must classify it and then signify the decision he has reached with a behavioral response, the resultant ERP is composed of successive positive and negative deflections continuing for up to 750 msec post stimulus. Figure 6.20 depicts such a waveform in which the major deflections or components have been labeled by a character-number designation. The character refers to the polarity of the component (P = positive, N = negative), while the number indicates the temporal delay or latency between the eliciting event and the peak voltage of the component.

The morphology of the components occurring within 200 msec after stimulus onset is influenced markedly by the physical attributes (e.g., wavelength, intensity or contrast) of the evoking stimulus (cf. Regan, 1972). Consequently, these so-called early components are referred to as exogenous. If the input must be processed and a decision reached, the prominence of late or endogenous component activity, particularly  $P_{300}$ , is affected. The term "endogenous" signifies that these components are not affected by the sensory qualities of external events.

The location of recording electrodes must be given careful consideration when denoting the amplitude and latency of ERP components. This follows from the previously stated position that neural representations of information processing and decision-making involve the coordinated behavior of disparate cell populations. Thus, the spatiotemporal distribution of late component activity, when referenced to the International Electrode Placement System (Figure 6.21), is a most important indicator of cognitive function (cf. Adam and Collins, 1978; Courchesne, et al., 1975; Goff, et al., 1978; Thatcher, 1976).

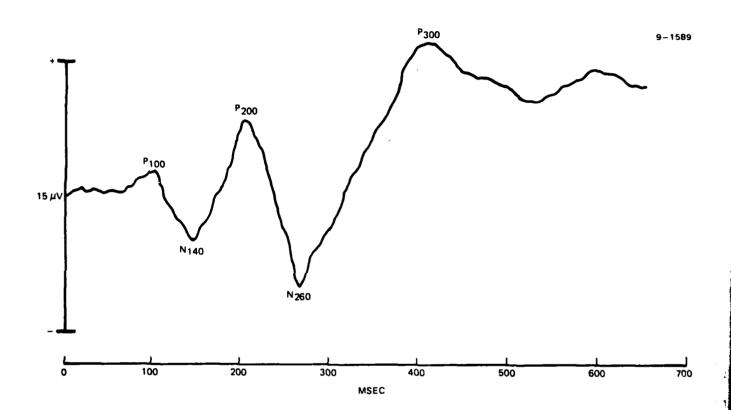


FIGURE 6.20 VERTEX EVENT-RELATED POTENTIAL ELICITED BY A MATCHING LETTER PRESENTATION DURING AN ITEM RECOGNITION TASK. (FROM GOMER ET AL., 1976)

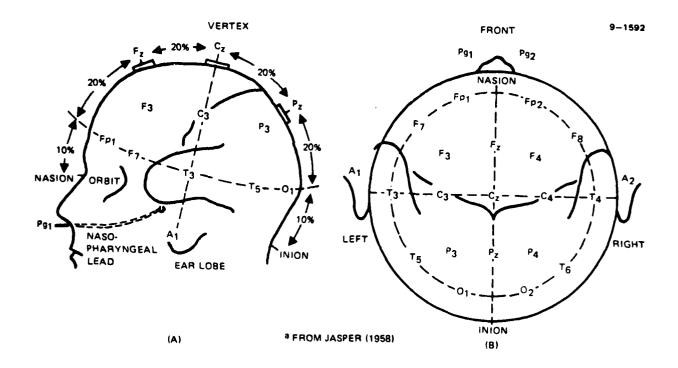


FIGURE 6.21 LATERAL AND SUPERIOR VIEWS OF INTERNATIONAL ELECTRODE PLACEMENT SYSTEM.

One may question whether it will be possible to evaluate some of the ERP components during tactical missions, since <u>discrete</u> visual events are rarely presented to the pilot (other than in the form of check list items). Rather, alpha numerics, symbolic characters, and sensor imagery are displayed <u>continuously</u>, and they change in content or value <u>dynamically</u>. Donchin (in press) has shown that discrete probe stimuli (in this case auditory) can be introduced artifically into those situations in which dynamic visual events predominate. By analyzing "background" responses (principally  $P_{300}$ ) to these probes we can infer how well the operator is performing "foreground" tasks.

"Foreground" activities can be assessed directly however. That is, we can evaluate the EEG changes which accompany information processing and decision-making operations that are integral to the housekeeping and mission-related tasks performed by the pilot. The EEG is categorized with respect to two basic dimensions, frequency and amplitude. Usual frequency bands are:

- o delta (.5 4Hz),
- o theta (5 7Hz),
- o alpha (8 12Hz),
- o beta (18 30Hz).

Energy distributions can be measured within these and more restricted frequency bands at each recording site. In fact, energy asymmetries in homologous leads (left vs. right hemisphere) may be quite sensitive to subtle differences in cognitive function.

**1 OCTOBER 1979** 

MDC E2046

Figure 6.18 lists other endogenous components, two of which are of additional value for monitoring pilot status. The contingent negative variation (CNV) is a slow, negative shift in EEG baseline which develops in the interval between successive presentations of discrete stimulus events (see Figure 6.22). The time course, amplitude, and scalp distribution of this waveform provide a general index of attentiveness (Donchin, in press). Thus, the probe technique which has been used to elicit  $P_{300}$  can also be used to generate CNV responses. The detection potential (DP), in contrast, is a transient change in EEG activity which is associated with the detection of dynamic target events (McCallum, in press; Cooper et al., 1977). McCallum and his colleagues discovered this event-related slow potential during preliminary studies of extended vigilance performance. Operators viewed a static landscape which was displayed on a television monitor. At random time periods, one of several vehicular targets would appear in the scene and traverse the terrain along a prescribed route. These scientists found that a well defined, positive-going potential reliably preceded behavioral indications of target detection (see Figure 6.23). Moreover, control experiments have established that this component is not related to the initiation of motor responding per se and that it has a predominant centroparietal scalp distribution. If the scope of these investigations can be expanded to incorporate more demanding target acquisition requirements, then the pilot's participation in a very important fire control function can be monitored directly.

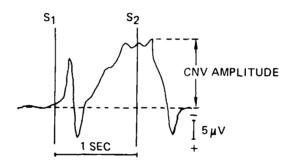
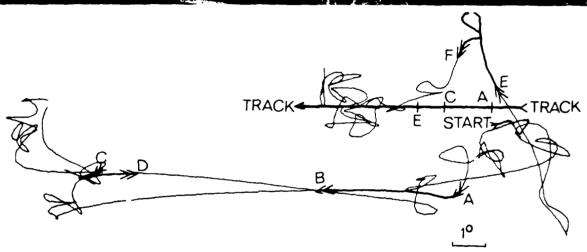


FIGURE 6.22 TYPICAL CNV WAVEFORM RECORDED FROM VERTEX ELECTRODE PLACEMENT.

(S1 AND S2 DENOTE SUCCESSIVE STIMULUS EVENTS.)





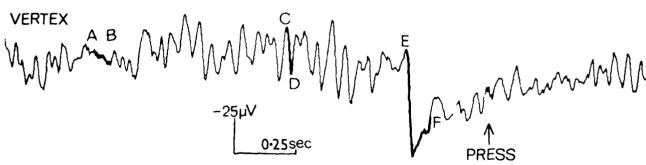
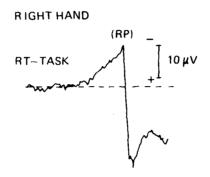


FIGURE 6.23 (TOP) THE DISPLAYED SCENE AND (BOTTOM) THE EYE-SCAN PATTERN AND VERTEX EEG OF ONE SUBJECT BEFORE AND AFTER DETECTION OF VEHICLE. (THE VEHICLE TRAVELED FROM RIGHT TO LEFT ACROSS THE MIDDLE OF THE DISPLAY. AT THE START OF THIS SECTION OF THE RECORD THE EYES WERE LOOKING AT THE CENTER OF THE DISPLAY. AT A THE EYES MOVED TO THE LEFT SIDE AND SCANNED THERE UNTIL B WHEN SACCADES TO THE UPPER CENTER C AND RIGHT DE OCCURRED, FOLLOWED BY DETAILED MOVEMENTS LEADING TO THE DETECTION POTENTIAL AT F AND TRACKING THEREAFTER. ALL SIX VEHICLES ARE SHOWN. (FROM COOPER ET AL., 1977))

We stated earlier that the recording and analysis of brain electrical activity also may permit a direct coupling of the pilot with aircraft subsystems from a control standpoint. At issue is whether it will be possible to interpret bioelectric manifestations of different thoughts as they occur. whereby the pilot can "think" to activate switches or guide control actions. Pinneo and his colleagues at Stanford Research Institute were funded by DAKPA to develop this particular communication link (Pinneo et al., 1975). Their primary objective was to isolate features in the EEG data that are associated with specific thoughts. They also attempted to devise computer pattern recognition programs that would identify these features on-line. For a small vocabulary of individual commands, they concluded that consistent patterns are present in that EEG activity which is coincident with the thinking of a particular word. Further, these patterns can be recognized and classifed by a computer a statistically significant percentage of the time. It must be noted, however, that this process is not yet sufficiently reliable to be used in a practical system.

There often is a need to increase the speed with which a control action is initiated, such as in weapons release during air-to-air gun attack. Researchers have discovered a negative slow potential shift, termed the "readiness potential" (RP) (see Figure 6.24), which precedes the actual execution of a voluntary manual response by as much as several hundred milliseconds (cf., Donchin, 1979; Gaillard, 1978). It is an electrophysiological indication of the intent to commence responding. This negative wave should not be confused with the CNV that was described previously. Whereas the RP is movement-related, the CNV is influenced by variables which modulate



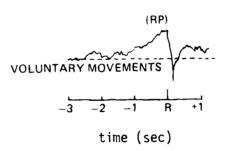


FIGURE 6.24 EXAMPLES OF VERTEX READINESS POTENTIALS ASSOCIATED WITH FINGER PRESSES DURING A REACTION TIME TASK AND DURING VOLUNTARY MOVEMENTS. (FROM GAILLARD, 1978)

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attentiveness to task demands. Moreover, Gaillard (1978) notes that the RP is distributed more posteriorly than the CNV and that the RP is bilaterally asymmetrical while the CNV is bilaterally symmetrical.

To reiterate, there are certain conditions which must be satisfied before electrophysiological signals will be valuable in the context of military applications. With respect to monitoring pilot status, clearly delineated and unambiguous patterns of brain electrical activity must be associated with information processing, attentiveness, and decision-making. Further, optimal patterns of activity (profiles) must be defined for these cognitive functions as a pilot successfully performs housekeeping and mission-related tasks. The profiles must be continuously updated and stored in computer memory onboard the aircraft. Current electrophysiological data which are recorded during the various stages of the mission must be evaluated in comparison with the profiles for deviations from acceptable levels. In the case of control system applications, we again must identify specific features of brain electrical activity which, in this instance, correspond with distinct thoughts and with the intent to initiate a manual response. And as we stated with regard to pilot status, current electrophysiological data must be matched, in real-time, with these unique profiles.

electrical activity and cognitive function, we recognize the importance of monitoring the pilot's reaction to chronic and acute stress. We are referring not only to environmental stressors, such as g forces or hypoxia, but also to psychological stressors, such as workload. If considerable bodily resources must be mobilized for extended time periods to restore the "status quo," then

1 OCTOBER 1979 MDC E2046

the pilot's proficiency in performing critical tasks will deteriorate. Therefore, we assume that <u>sustained</u> increases in certain types of biological activity (or perhaps in the variability of a particular measure) will reflect difficulty in coping with stress. We very briefly describe psychophysiological activity in three response systems that have been studied to assess levels of anxiety, tension, or physical effort.

Within this subsection we also review ocular activity from two perspectives. We begin by examining the dynamics of eye movements (i.e., timing, velocity, and pattern) as they relate to information extraction and processing. Then we discuss the concept of eye control.

6.2.2.1 <u>Psychophysiological Responses</u> - Surface recordings of electromyographic (EMG) activity are widely used to evaluate changes in tension and physical effort (cf. Goldstein, 1972; Johnson, in press). Electrodes are placed on the skin over various muscle groups, although forehead and neck leads are usually included when the effects of psychological stressors are examined. The electrical activity generated in the muscles is frequently rectified and integrated over an interval of time, the duration of which depends upon the purpose for the recording.

Shifts in cardiovascular activity also occur in response to stress and changes in physical effort. The parameters of interest to most investigators have been (a) heart rate and rhythm and (b) blood flow and pressure (Gunn et al., 1972). Heart rate (HR) has been the preferred datum when evaluating anxiety or tension, due largely to practical difficulties in acquiring the other data during simulation and operational testing. Direct measurement

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techniques consist of (a) recording the electrical activity generated by contraction and relaxation of cardiac muscle - the electrocardiogram (ECG), and (b) detecting the corresponding heart sounds - the phonocardiogram. Indirect techniques usually involve some form of sensing the changes in peripheral blood flow. There is a question, however, whether fluctuations in mean HR or in HR variability provide the most sensitive indication of physical status.

Similarly, both the rate and depth of respiration increase significantly when subjects are exposed to periods of stress. The most common methods for measuring these parameters incorporate impedance, strain gauge, or thermistor techniques. It must be noted that respiratory patterns are disrupted by speech, thereby limiting the value of rate and amplitude recordings during certain mission segments.

Our goal, as far as monitoring the pilot is concerned, is to identify consistent patterns of activity - not only in individual psychophysiological measures but in a group or groups of measures - that are related to distinct changes in anxiety, tension, and physical effort.

6.2.2.2 Ocular Activity - It is generally accepted that eye movements reflect the observer's distribution of attention within visual space (cf. Krebs et al., 1977). However, the timing and velocity of these movements may also reveal the effectiveness with which a pilot processes information. We suggest that ocular measures may supplement electrophysiological measures of cognitive function during display monitoring tasks. A more detailed discussion of the types of eye movements and their relation to perception and cognition is presented by Cumming (1978).

Large movements orient the eye so that the high resolution area of the retina (the fovea) is directed toward the point of interest within the visual scene. We can define three classes of large eye movements. Saccades are voluntary, abrupt changes in fixation between points located at the same viewing distance. They are characterized by accelerations and decelerations of up to 40,000 deg/sec<sup>2</sup> and by peak velocities of 480 to 600 deg/sec. Saccadic eye movements during visual search typically subtend 1 to 40 deg. Smooth movements, on the other hand, occur when the eyes track an object which moves vertically, obliquely, or laterally in the range of 1 to 30 deg/sec. These movements are also produced to compensate for head or body motion as the observer fixates on a stationary object. Finally, vergence movements allow the eyes to adjust to changes in viewing distance or depth.

Even when an observer carefully attempts to maintain precise fixation, small movements inevitably persist. These movements are classified as tremor, microsaccades, and drifts. According to Cumming (1978):

"Tremor is a small, irregular lateral oscillation with frequency components up to 100 Hz and an amplitude equivalent to a few foveal cone diameters. Microsaccades are small, fast, conjugate flicks, taking some 20 msec and moving the eyes a few minutes of arc. Between microsaccades the eyes drift haphazardly at about 5 min of arc/sec, with tremor superimposed on the drifting motion. The two eyes drift independently and probably also undergo tremor independently, and so these movements have usually been attributed to unavoidable residual instability in the oculomotor system." (p. 224)

We are most concerned with the dynamics of large eye movements that occur as the pilot scans displays and the visual scene outside the crew station. Two-stage models of visual perception hold that events of interest are located initially via peripheral vision; this leads to fixation and,

therefore, more detailed analysis. Many investigators believe that fixation implies attention. However, the relationships between the velocity and the amplitude (angular distance) of saccadic eye movements may provide additional insights concerning fluctuations in attentiveness or in decision strategy (Stern, 1978). Equally as important in this regard are temporal parameters, such as fixation duration (dwell time), since unusually long or short fixation pauses may reflect periods in which visual information is not being

processed.

Young and Sheena (1975) have surveyed most of the procedures for measuring eye movements and eye position within the laboratory. Unfortunately, a major shortcoming of many measurement techniques is that they require varying degrees of head stabilization to achieve accurate determinations of eye position. This obviously limits their usefulness in operational crew stations. Thus, it is noteworthy that the military has developed procedures for transmitting head position (and, in an indirect manner, line-of-sight) coordinates to create a close coupling of the pilot with aircraft subsystems from a control standpoint.

Furness (in press) has reviewed several methods for deriving line-of-sight data in the cockpit to facilitate the aiming of weapons, the designation of ground targets, or the activation of control surfaces. One system which has been developed incorporates a tightly fitted helmet and a parabolic visor. A gunsight reticle is projected via an optical assembly onto the center of the visor. The reticle appears as a 10 mil ring within a 50 mil ring and is collimated. Two lead sulfide photodiodes are located on each side of the helmet, and, of course, their positions are constant with respect

1 OCTOBER 1979 MDC E2046

to the position of the reticle. In principle, as the crew member moves his head and superimposes the reticle over a target of interest, the relative positions of these two photodiodes are measured and translated into line-of-sight information. To accomplish this, an infrared scanning device is located behind the crew member on the canopy rail. It generates two parallel planes of infrared light that rotate throughout the cockpit and illuminate the two photodiodes. The signals from the photodiodes and the timing signals from the scanner are transmitted to a special-purpose digital computer. The computer determines the positions of the photodiodes in three-dimensional space and resolves the resultant vector into azimuth and elevation angles. Once relative azimuth and elevation coordinates are known, they are combined with information about aircraft boresight to specify, in an absolute sense, precisely where the crew member is aiming his head (and thus directing his gaze).

Electro-oculographic (EOG) and, to a lesser extent, oculometer techniques permit the recording of eye movements without placing restraints on allowable head movements. Merchant (in press) has recently described the development of a helmet-mounted oculometer, the design of which is based on the visor-projected reticle system used to infer line-of-sight in the cockpit. In this adaptation, the reticle generator on the side of the helmet is replaced with miniaturized versions of the oculometer sensor and the infrared illumination source. The latter projects light rays off the parabolic visor, while the former views a reflected image of the eye. Not only can the dynamics of large eye movements be studied in this manner, but eye position can be determined directly by measurement of the distance between the corneal reflection (of the light rays from the illumination source) and the pupil center.

#### **1 OCTOBER 1979**

### 6.3 BIOCYBERNETIC APPLICATIONS

Our premise has been that system effectiveness can be improved dramatically if the central computer is made aware of momentary shifts in operator status. In addition to monitoring transient aspects of cognitive function, we also believe it worthwhile to assess more long term or general status, as influenced by such factors as fatigue, anxiety, and physical well-being.

The previous subsection (6.2) documented that mental activities are indeed manifest in a variety of biological signals. Subsequent to computer analysis of the informational content (both phasic and tonic) of these signals, we suggested that it would be possible to unburden or assist the pilot through:

- o redistribution of task responsibilities by effecting greater automation of certain housekeeping functions,
- o reduction in the complexity of information displays, especially the HUD.
- o cues for the pilot to attend to critical flight, weapons, and target data,
- o presentation of adaptive decision aids which recommend mission-related strategies for fire control functions,
- o recall of checklist items,
- o more optimal adjustments of the physical characteristics (e.g., contrast, focus, etc.) of imagery and symbolic displays.

**1 OCTOBER 1979** 

**MDC E2046** 

Moreover, we noted that (a) pattern analysis of brain electrical activity associated with either distinct "thought" commands or the intent to initiate movement and (b) eye position sensing, provided separate means of augmenting (especially in terms of speed) manually operated and voice actuated controls.

Figure 6.25 presents the biological signals listed earlier in Figure 6.18, but now links them to specific applications. With respect to electroencephalographic activity, we presume that distinctive features are sensitive to continual as well as sudden cognitive demands. Remember that "features" refer to energy distributions within restricted frequency bands and to possible asymmetries in these distributions across standard recording sites. Other features of EEG activity may emerge as trained pilots "think" particular commands, thereby creating the necessary inputs to initiate and guide control actions. As evident from the figure and the text of Subsection 6.2, exogenous and endogenous components of event-related potentials are also valuable sources of pilot information. For example, by referencing the time course, amplitude, and locus of maximal response for exogenous components of visual ERPs (since each of these attributes change as image quality and contrast are manipulated (cf. Gomer and Bish, 1978)), display settings can be adjusted automatically to achieve criterion levels of display performance. Regan (in press) recently has proposed such a "feedback loop" for a form of exogenous activity labelled steady-state.

Returning to Figure 6.25, the suggested applications for endogenous components of ERPs require no further explanation. However, we should review the information-bearing properties of the different types of peripheral activity, as well as some of the headings we have employed to depict their

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RELATIONSHIP BETWEEN PARTICULAR BIOLOGICAL SIGNALS AND EITHER THE DETERMINATION OF PILOT STATUS OR CONTROL FUNCTIONS. FIGURE 6.25

**1 OCTOBER 1979** 

**MDC E2046** 

usage. Previously, we described psychophysiological activity in three response systems that have been monitored extensively to assess tension and physical effort, particularly following the onset of stress. Therefore, we have grouped these response systems together and assumed that the principal virtue of recording from them will be to denote changes in "physical status." Again, sustained increases in the behavior of these response systems, if caused by prolonged exposure to stress, should forewarn an imminent deterioration in pilot proficiency. Finally, the dynamics of eye movements (i.e., timing, velocity, and pattern) should serve as an additional indication of the effectiveness with which a pilot extracts and processes displayed information.

In taking the process to its logical conclusion, Figures 6.26 through 6.38 present biocybernetic applications as a function of individual pilot tasks. These are tasks which occur within the various segments of the escort mission we described before in Subsection 6.1. For each of the tasks which must be performed, we have indicated whether we are seeking status information of some sort, or whether we intend to supplement conventional methods of control system activation.

We recommended earlier that biocybernetic applications be restricted to those pilot tasks which are very difficult, are critical to the success of the mission, or occur during periods of heavy workload. Whereas the original task listings in Figures 6.6 through 6.17 are quite complete, we now have eliminated less essential tasks (with respect to difficulty, criticality, or

Workload considerations) in constructing the biocybernetic matrices which follow. Moreover, to avoid unnecessary redundancy in the format of successive figures, we show just those tasks which have not appeared in any of the preceding matrices (starting with Figure 6.26). For example, although the pilot must determine altitude during launch as well as climb, this task is

entered as a row heading only in the matrix devoted to launch.

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MONITOR AIRSPEED	}x	×			x			
SCAN EXTERNAL	x	x		[	X			
MONITOR ATTITUDE	×	×			X			
ATTAIN PROPER ATTITUDE	<u> </u>			ļ			<b></b>	·
MONITOR FORWARD ACCELERATION	x	×		1	x			

FIGURE 6.26 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING LAUNCH.

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	ELECT	TROENCEPHALOG ACTIVITY	RAPHIC	EVENT-RELATED POTENTIALS				
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					P300	DETECTION POTENTIAL	READINESS POTENTIAL	
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CHECK OXYGEN QUANTITY	x	×			×			
MONITOR VERTICAL VELOCITY	x	×			x			
RECEIVE COMMUNICATION	·		x					
COMMUNICATE			x					
UNLOCK SHOULDER HARNESS				<b>]</b>	••••••	}		
CLEAR VISUAL AIRSPACE WINGMAN POSITION	<b>∤</b>	ļ	ļ		• • • • • • • • • • • • • • • • • • • •	·····		
ATTAIN DESIRED THRUST	<b>}</b>	}		}	••••••			
MONITOR NAVIGATION	x	×			x			
MONITOH HEADING	×	X			x			

FIGURE 6.27 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING CLIMB.

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CHECK RADAR RANGE SCALE		×	 		×		
MONITOR AUTOMATIC DIRECTION FINDER BEARING		ļ	x		×		

FIGURE 6.28 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING RENDEZVOUS.

ORIES OF BIO	LOGICAL SIG	NALS				
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PILOT TASKS			₹ €	AD DE		- 5 <del>1</del>	<u> </u>	13	8 8
MON-TOR HADAR DISPLAY		×				ı			
CHECK FUEL STATUS		×		l	×				<del>                                     </del>
MONITOR ENGINE INSTRUMENTS	x	×			x		İ	1	
MAINTAIN VISUAL SURVEILLANCE			(	<b>{</b>				x	×
SET RADAR STORAGE (TARGET DISPLAY) TIME			x						
CHECK AIR TO AIR ARMAMENT STATUS		×			x		ĺ		
ERIFY SPARROW MISSILE SELECTED	X								ļ
SELECT AIR TO AIR MODE			x	1					
VERIFY MODE SELECT (PROGRAM)	x								1
CHECK JAMMER DISPENSE COUNTER	x	×			×				
CHECK FLARE DISPENSE COUNTER	x	×			×	1			1
CHECK CHAFF DISPENSE COUNTER	×	×			×		ļ		
VERIFY ALTOMATIC CHAFF DISPENSE ENABLE	x			<b>!</b>				×	
VERIFY CHAFF DISPENSE ENABLE	x								
ADJUST BRIGHTNESS OF THREAT DISPLAY	x		×	×					
SET RADIO VOLUMES		 	! x	×			1		
VERIFY ELECTRONIC COUNTERMEASURE SELECTED	x	×	I						1
VERIFY SAM RECEIVER ON							ļ		
THIRN RADAR ALTIMETER OFF			x						
TURN TACTICAL AIR NAVIGATION TO RECEIVE			x						
TURN IDENTIFY-FRIEND OR FOE TO STANDBY		l				 	1		[ ]
. ERIEF : ROSSING OF THE FORWARD EDGE OF RATTLE AREA		×							×

FIGURE 6.29 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING INGRESS.

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(	NALS				
			PERIPHERAL	ACTIVITY	
		PSYCHO- PHYSIOLOGICAL RESPONSES	,	CULAR ACTIVITY	
	CONTINGENT NEGATIVE VARIATION		EYE MOVEMENTS	5021. EA	E FION
\$					
	SS IN 1012 N. 12 C. I. C. 13 C. I. C. 13 C. I. C. 14 C. I. C	STREET AND THE PROPERTY OF THE STREET AND THE STREE	X ASSESSOR AT LINGMITUE STATUS	ACTIVATEAND DE CONTROL SOBSVATEMS	DETERMINE WHETHER GAZE GAS BEEN DIRECTED TO APPROPRIATE FOR ATION
			x x		x
-			×		x
ļ	x	x	×		×
				×	
			' 1		×
	<del></del>			×	× _
				^	×
			×		
			× ×		
1	×		<u> </u>		×
I					×
ı				×	
1					x
1	*****			×	^
١				×	
				×	
l		×	×		×

 $\mathcal{V}$ 

					CATEC	ORIES OF BIO	LOGICAL SIG	NALS	
		**********	BRAIN ELECT	RICAL ACTIVITY					
	ELECTI	ROENCEPHALOGE	ЗАРНІС	_	£ • EN	T RELATED POTEN	ITIALS		PSYCH PHYSIOLO RESPON
				EXOGENOUS COMPONENTS		ENDOG COMPO	ENOUS NENTS	CONTINGENT	•
		Į.			P300	DETECTION	READINESS POTENTIAL	VEGATIVE VARIATION	
				L		BIOCYBERNETIC	APPLICATIONS		<u> </u>
	ļ			<del>r</del> -	-	SIOCIBERIVETIC	AFFEICATIONS	v.	1
PILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PROCESS IN GRNATION AND MAKE APPHOPRIATE DELISIONS	ACTIVATE AND OH CONTROL SUBSYSTEMS	ADJUST DISPLAY PAHAMETEHS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	HE WE APONS	EVALUALI MOMENTARY FLUITUALIONS IN ALTENTIVENESS	ASSESS GENERAL PHYSICAL
CHECK NAVIGATION DATA		×	<del>                                     </del>	<b>†</b> – †	x				×
SET MASTER ARM TO SAFE	I		x	ł					[ ~
OBSERVE FLASHING BREAKAWAY CUE	Ì	Ì	<u> </u>	<u>l</u>	x				l
OBSERVE MISSILE TIME OF FLIGHT	l	×			x				i
VERIFY MISSILE LAUNCH			.,						1
DEPRESS TRIGGER			x				×		
OBSERVE IN RANGE CUE				<u> </u>	x			×	
OBSERVE LAUNCH LIMITS STEERING, ALLOWABLE STEERING ERROR.	Ī	×		<b>,</b>	x				1
FLY STEERING DOT COMMANDS									x
JETTISON FUEL TANKS			, x						
CHECK SPARROW MISSILE STATUS		×		<b>i</b>	×				1
SCAN EARLY WARNING DISPLAY TARGET STROBE BEARING	x	×			×				
NOTE AIR INTERCEPT WARNING LIGHT				<u>[</u>					×
OBSERVE RADAR TRACKING TARGET ASSESS DATA	×	×			x	×			† —
VERIFY LOCK ON		×			×				ļ
DEPRESS THPOTTLE DESIGNATOR CONTROL TO LOCK ON	1		x						[
POSITION ACQUISITION SYMBOL	 		x						1
HERE SEA KNOWLEDGMENT		ļ	ļ	ļ			•••••	x	
SET MASTER ARM TO ARM			×						<del> </del>
CASSERVE AZIMUTH I AN	x	×	[		×				
CRSSHIVE AST TOPE COVERAGE	x	×			x				
SET ANTENNA A CE VATION	,		x						ţ
SE EST VELOSITY STARSH MODE		<b></b>	×				i		
									L

FIGURE 6.30 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING MEDIUM RANGE INTERCEPT.

L SIG	NALS				
			PERIPHERAL	ACTIVITY	
		PSYCHO- PHYSIOLOGICAL RESPONSES		CULAR ACTIVITY	
ESS MAL	CONTINGENT NEGATIVE VARIATION		EYE MOVEMENTS	EY POSI	E TION
TIONS					
	EVALUATE MOMENTARY FLUCTUATIONS IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSISS GINERAL COGNITIVE STATUS	ACTIVATE AND OR CONTHOL SUBSYSTEMS	DI TEHMINE WHE THER GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION
		x		x	×
			×		×
	×		x x		x x
••••		x	x	×	x
			×		×
		×	×		x
••••	<b>x</b>			× ×	
				×	
				×	

					CATE	GORIES OF BIO	DLOGICA
			BRAIN ELECTR	ICAL ACTIVITY			
	ELECT	ROENCEPHALCGR ACTIVITY	APHIC		EVEN	T RELATED POTE	NTIALS
				EXOGENOUS COMPONENTS			SENOUS INENTS
					P300	DETECTION POTENTIAL	READINI POTENT
						BIOCYBERNETIC	APPLICAT
PILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PHOCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	ACTIVATE AND OH CONTROL SUBSYSTEMS	ADJUST DISPLAY PARAMETERS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	FIRE WEAPONS
PERFORM EVASIVE MANEUVERS	x	x	×		×	×	
ACTIVATE ELECTRONIC COUNTERMEASURES CHAFF.			x			<u> </u>	•
DETECT SAM LAUNCH ON THREAT LIGHT						×	<del>                                     </del>
MONITOR THREAT DISPLAY (SAM SITE AZIMUTH)		×			×		
DETECT ENEMY RADAR LOCK ON	x	×	·····		•	×	
CHECK DEFENSE ELECTRONIC COUNTERMEASURES	x	×					

FIGURE 6.31 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING SURFACE-TO-AIR MISSILE AVOIDANCE.

					PERIPHERAL	. ACTIVITY				
NT.	RELATED POTEN	TIALS		PSYCHO- PHYSIOLOGICAL RESPONSES	HYSIOLOGICAL OCULAR ACTIVITY					
	ENDOGENOUS COMPONENTS  DETECTION READINESS POTENTIAL POTENTIAL		CONTINGENT NEGATIVE VARIATION		EYE MOVEMENTS	EY POSI				
BI	OCYBERNETIC	APPLICATIONS	l							
	DE TEHMINE WHETHEH TARGET HAS BEEN DETECTED	FIHE WEAPONS	EVALUATE MOMENTARY FLUCTUATIONS IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND OR CONTROL SUBSYSTEMS	DETERMINE WHETHER GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION			
	×		x	x	x	x	×			
1			x	×	×		×			
						×				
Ţ	х						×			
	x						×			
1				×	×		×			
				_ x	x		×			

1 OCTOBER 1979

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		CATEGORIES OF BIOLOGICAL SIGNALS								
			BRAIN ELECT	RICAL ACTIVITY					Γ	
	+.+	THUE NEEPHALO			f • E	NT HELATED POTE	NTIALS		PSYCHO PHYSIOLOG RESPONS	
				EXTRUENOUS COMPONENTS			GENOUS ONENTS			
					F300	DETECTION POTENTIAL	READINESS POTENTIAL	CONTINGENT NEUATIVE VARIATION	ļ	
			<u> </u>	<u></u>		BIOCYBERNETIC	APPLICATIONS		<u> </u>	
								v.	Γ	
	ASSESS GERRRAL COGNETICE STATUS	EVALUATE MOMENTARY AREATS TO PHOLESS IN ORMATION AND MAKE APPROPRIATE DECISIONS	TVATE AND OFFERNITION	ADDIST DISPLAY VAHAMI 11 HS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MANE APPROPRIATE DECISIONS	(I) II RAMA WHETHER TARGET HAS BEEN DETECTED	i Waapuns	ALINAT WORLD STATE FOR LOATIONS	ASSESS GENERAL PHYSICAL STATUS	
PILOT TASKS	4	254	4 %	<del>-</del>	7 TO API	5 1	¥	2.2	8 £	
DETECT STRIKE FORCE RADAR	<b></b>			<b>.</b>		x				
SELECT NAVIGATION MODE	x	×	x	i .	×				1	
DETERMINE NEW HEADING TO STRIKE FORCE	}×	х								
CHECK ANGLE OF ATTACK	ļ	· · · · · · · · · · · · · · · · · · ·								
OBSERVE RETICLE OVER TARGET		×		ĺ	x			×		
OPTIMIZE RETICLE DYNAMICS	ļ	• • • • • • • • • • • • • • • • • • • •	x	×						
CHECK ROUNDS REMAINING	×	×	·						l	
OBSERVE GUN SYMBOLOGY	x	×			×					
SELECT GUN MODE	×	×	×			j [				
VISUALLY ASSESS MISSILE SUCCESS		·····		<b></b>		······			x	
VISUALLY MONITOR MISSILE FLIGHT		<del></del>	·			ļ			х	
VISUALLY VERIFY MISSILE LAUNCH				<u> </u>		<u> </u>			x	
OBSERVE IN RANGE CUE SHOOT LIGHT	x	×				×		x	×	
HEAR VISSILE TONE CHIRP		ļ	<b></b>			x				
WAINTAIN STEER DOT WITHIN ALLOWABLE STEERING ERROR	x	×			x				x	
DEPRESS UNCAGE SWITCH	• • • • • • • • • • • • • • • • • • • •	<del>{</del>	× │						ļ	
HEAR MISSILE TONE ABOVE THRESHOLD					x	×		ĺ		
OBSERVE TARGET DETECTION BOX SIDEWINDER CIRCLE ALIGNMENT		×		T					×	
PUSH RUDDER PEDALS		ł	1							
EXTEND RETRACT SPEEDBRAKE			x						Í	
OBSERVE TARGET DETECTION BOX OVER TARGET	- 1		ļ <b>.</b>			ļ	,		х	
SECECT VISUAL ACQUISITION MODE HACQ, VACQ, BORESIGHT			x					Į		
PASSERVE SIDEWINDER SYMBOLOGY STATUS		×			×					
SELECT SIDEWINDER MODE			x					[		
TRACK TARGET VISUALLY		••••••	ļ <b>.</b>						х	
HECK FILEL STATUS BINGO SETTING		×			×		ĺ	ĺ		

FIGURE 6.32 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING AIR COMBAT MANEUVERING.

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					_
L SIG	NALS				
			PERIPHERAL	ACTIVITY	
_		PSYCHO- PHYSIOLOGICAL RESPONSES		XULAR ACTIVITY	
iess Hal	CONTINGENT NEGATIVE , ARIATION		ÉYE MOVEMENTS	E Y POSI	E TION
rions			·		
	Example MIMELANCE CONTRACTORS IN A STRUCTURE OF STRUCTURE	ASSLSS of NEMAL PHYSICAL STATUS	ASSESS CENERAL COGNITIVE STATUS	ACTIVATI AND OH CONTHOL SUBSYSTEMS	DETERMINE WHETHER CAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION
				x	
			×		
••••	······	·····			x
	x				x x
					×
		x	x x	x	х х х
	х	× ×	×	-	^x
	 	x		×	*
		х	х		×
			]	×	
		х	×	x x	×
		х	x	x	х х х
		х	х		

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1					CATE	GORIES OF BIG	DLOGICAL SIG	GNA
			BRAIN ELECT	TRICAL ACTIVITY				
	£ L E C	TROENCEPHALOGE ACTIVITY	APHIC		EVEN	IT RELATED POTER	NTIALS	
1				EXOGENOUS COMPONENTS			GENOUS ONENTS	
					P300	DETECTION	READINESS POTENTIAL	0
					<del></del>	BIOCYBERNETIC	APPLICATIONS	
PILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	ACTIVATE AND ORICONTROL SUBSYSTEMS	ADJUST DISPLAY PAHAMETEHS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	FIRE WEAPONS	
OBSERVE FLASHING SYMBOLOGY  DEPRESS PICKLE BUTTON  SELECT HUD SYMBOLOGY  TRACK TARGET IN HUD FIELD OF VIEW	•••••			x	×		×	
OBSERVE DELIVERY MODE SYMBOL		х	×					
CHECK DELIVERY MODE		×						
SELECT AIR GROUND MASTER MODE			x					

FIGURE 6.33 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING AIR-TO-GROUND STRIKE.

DLOGICAL SIG	NALS						
			PERIPHERAL	ACTIVITY			
iTiALS		PSYCHO- PHYSIOLOGICAL RESPONSES	OCULAR ACTIVITY				
ENOUS NENTS READINESS POTENTIAL	CONTINGENT NEGATIVE VARIATION		EYE MOVEMENTS	E Y POSI			
APPLICATIONS					L		
FIRE WEAPUNS	EVALUATE MOMENTARY FLUCTUATIONS IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND OH CONTHOL SUBSYSTEMS	DETERMINE WHETHEN GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION		
		×	×		×		
Î	×	×	x		x x		
					х		
				×	U		
					×		
		х	×	х			
	ENOUS NENTS READINESS POTENTIAL APPLICATIONS	ENOUS NENTS CONTINGENT READINESS POTENTIAL  APPLICATIONS  APPLICATIONS  X  X  X	PSYCHO-PHYSIOLOGICAL RESPONSES  ENOUS NENTS  CONTINGENT NEGATIVE VARIATION  APPLICATIONS  APPLICATIONS  X  X  X  X  X  X  X  X  X  X  X  X  X	PERIPHERAL PSYCHO-PHYSIOLOGICAL RESPONSES CONTINGENT READINES POTENTIAL ASSESS GENERAL PHYSICAL STATUS  X  X  X  X  X  X  X  X  X  X  X  X  X	PERIPHERAL ACTIVITY  PSYCHO- PHYSIOLOGICAL RESPONSES  ENOUS NENTS CONTINGENT READINESS POTENTIAL  APPLICATIONS  APPLICATIONS  X  X  X  X  X  X  X  X  X  X  X  X  X		

					CATE	GORIES OF BI	OLOG <b>IC</b>
			BRAIN ELECT	RICAL ACTIVITY			
	FLECT	TROENCEPHALOG ACTIVITY	RAPHIC		EVEN	NT RELATED POTE	NTIALS
				EXOGENOUS COMPONENTS			GENOU <b>S</b> DNENTS
					P300	DETECTION POTENTIAL	READI POTEN
						BIOCYBERNETIC	APPLI <b>C</b>
PILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	ACTIVATE AND OR CONTROL SUBSYSTEMS	ADJUST DISPLAY PARAMETERS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	FIRE WEAPOWS
DISPENSE CHAFF  PERFORM SAM EVASIVE MANEUVER	x	×	x x		x	×	
SELECT PROGRAM OR SINGLE DISPENSE			x				
VERIFY ELECTRONIC COUNTERMEASURES SELECTED  VERIFY ELECTRONIC COUNTERMEASURES ON	x	x x		X	I		

FIGURE 6.34 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING EGRESS.

è	GORIES OF BIO	DLOGICAL SIG	INALS				
					PERIPHERAL	ACTIVITY	
Å	T-RELATED POTEN	ITIALS		PSYCHO- PHYSIOLOGICAL RESPONSES		·	
	ENDOGENOUS COMPONENTS  CONTINGENT  DETECTION READINESS NEGATIVE POTENTIAL POTENTIAL VARIATION				EYE MOVEMENTS	YE TION	
-	BIOCYBERNETIC	APPLICATIONS		L			
-	Joe Bennette	ATTENDATIONS	Ş				
	DETERMINE WHETHER TARGET PAS BEIN DETECTED	FIHE WEAPONS	EVALUATE MOMENTARY FLUCTUATIONS IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND OR CONTROL SUBSYSTEMS	DETERMINE WHETHER GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION
						×	
	X		x	x	x x	x	x x
						x	
						X	
	}						x
				×	×		×

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1 OCTOBER 1979

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1	CATEGORIES OF BIOLOGICAL SIGNALS									
			RRAIN SI SCTO	ICAL ACTIVITY	CATEC					
	FLECTE	ROENGEPHALDGF		CAL ACTIVITY	EVEN'	T RELATED POTER	ITIALS		PSYCH <b>O-</b> FritSIOL <b>OGIC</b>	
		ACTIVITY	r						HESPONSE	
				EXOGENOUS COMPONENTS		ENDOG COMPO		CONTINUES?		
					P300	DETECTION	READINESS POTENTIAL	NEGATIVE VARIATION		
									L	
			1	γ		IOCYBERNETIC	APPLICATIONS		1	
	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DE CISIONS	AI TIVATE AND OR CONTROL	ADJUST DISPLAY PAHAMITERS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DE TEHNING WHETHER TARGET	IH: WÉAPONS	ALIVATE WOMEN AND ALIVE TO A TO A	ASSESS GENERAL PHYSICAL STATUS	
PILOT TASKS	A S.	354	S & S	¥		ΞÍ		2 <u>z</u>	~ ~ ~	
CONFIRM POSITION OF ECHELON	×	×								
CHECK INDICATOR FOR CORRECT AMOUNT									-	
NOTE TANKER STANDBY GREEN OFF	1	ľ	ľ			×				
CHECK TANKER STANDBY GREEN ON	i					X				
OBSERVE TRANSFER LIGHT OFF			<del></del>			×		<u> </u>	<del>├</del> ──	
CHECK TRANSFER LIGHT ON			ł			^				
CHECK TANKER READY LIGHT ON		х			x	×				
CHECK DROGUE EXTENDED		ļ	x	\		_ ^	ŀ		1	
CHECK PROBE FULLY EXTENDED	1	×			×	×				
POSITION PROBE TO EXTEND		×	×				<del> </del>	<u> </u>	1 1	
SET EXTERNAL LIGHTS TO STANDBY BRIGHT			x							
DESELECT AUTOMATIC DIRECTION FINDER			1	į						
DEACTIVATE AUTOMATIC FLIGHT CONTROL SYSTEM	i e							İ		
FLY FORMATION THRUST		l	1			,		ļ	x	
FLY FORMATION ATTITUDE									х	
CHANGE TO REFUEL FREQUENCY			x							
CHECK MASTER ARM SAFF							ļ	ļ		
VERIFY ALTITUDE SETTING	ļ •	·····			•••••	<b></b>				
VERIFY LOCK ON							ļ	x		
COMMAND LOCK ON			×							
DESIGNATE TARGET FOR LOCK ON	<b></b>		×			×	1			
CHECK AERIAL REFUELING CONTROL POINT COORDINATES		×			×					
SELECT SECTION SCAN			х							
SELECT BAR SCAN	<del></del>		· · · · · · · · · · · · · · · · · · ·						1	
SELE/,T RADAH RAN/,E	1	×	×						1	
ENABLE WAYPOINT		ĺ							] ]	
SELECT RENDEZVOUS FREQUENCY		1			×		1			
SET IN AERIAL REFUELING CONTROL POINT ( )ORDINATES		×	×							
SELECT TARGET CLEARANCE ALTITUDE CHANNEL AIR TO AIR MODE.			}×	L			L	<u> </u>	×	

FIGURE 6.35 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING IN-FLIGHT REFUELING.

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OGICAL SIG	NALS	<del></del>			
			PERIPHERA	ACTIVITY	
MLS		PSYCHO- PHYSIOLOGICAL RESPONSES	(	OCULAR ACTIVITY	
NTS READINESS POTENTIAL	CONTINGENT NEGATIVE VARIATION		EYE MOVEMENTS	POSIT	E :10N
PLICATIONS					
I IFE WEAPONS	EVALUATE WOMENTARY TEOFTUATIONS IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND OR CONTROL SUBSYSTEMS	DE TEHMINE WHETHER GAZE HAS BEEN JIRECTED TO APPROPRIATE LOCATION
					×
<b></b>			х х х		x × ×
	x		×		×
			   	×	×
			_	x	x x x
		x	×	x	x
	x				x x x
		,		x x	
				×	
				x x x	
		×	×	x x	

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1 OCTOBER 1979

MDC E2046

					CATE	GORIES OF BI	
			BRAIN ELECT	RICAL ACTIVITY			
	ELEC	CTROENCEPHALOGE ACTIVITY	RAPHIC		EVE	NT-RELATED POTE	NTIA
				EXOGENOUS COMPONENTS			
			1		P300	DETECTION POTENTIAL	;
				<b>\</b>			
			1			BIOCYBERNETIC	API
ILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PHOCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	ACTIVATE AND OH CONTROL SUBSYSTEMS	ADJUST DISPLAY PARAMETERS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	
CHECK MULTIMODE DISPLAY FOR AUTOMATIC CARRIER LANDING STATUS	,	_ x			×		
VERIFY DATA LINK FREQUENCY							
SELECT COURSE	×	×	X		X		
ATTITUDE DIRECTOR INDICATOR			×				
VERIFY INSTRUMENT LANDING SYSTEM CHANNEL		×			<b>X</b>		
VERIFY BEACON CHANNEL	1	×		1	×		
SELECT AUTOMATIC CARRIER LANDING			x			<del> </del>	1_
VERIFY IDENTIFY FRIEND OR FOE CODE	×	×					
ENTER IDENTIFY FRIEND OR FOE DIGITS			×				-
SELECT IDENTIFY FRIEND OR FOE	- 1	×					
POSITION SPEEDBRAKES			×			1	
ADJUST CABIN AIR AS REQUIRED		}	x	1			Ì
SELECT RADAR OPERATE	х	×	×				
SELECT WAYPOINT	x	x	x				-
ACTIVATE RADAR ALTITUDE			x				
DEPRESS ENTER		1		1			
SELECT TACTICAL AIR NAVIGATION	1	x	×	1	x		1

FIGURE 6.36 BIO' SERNETIC APPLICATIONS AS A FUNCTION OF PILOT TAS. OURING MARSHAL.

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EG	ORIES OF BIC	LOGICAL SIG	INALS				
					PERIPHERAL	ACTIVITY	
EN1	RELATED POTEN	TIALS		PSYCHO- PHYSIOLOGICAL RESPONSES	(	OCULAR ACTIVITY	<i>'</i>
	ENDOG COMPO		. CONTINGENT		EYE MOVEMENTS	EYE POSITION	
_	DETECTION POTENTIAL	READINESS POTENTIAL	NEGATIVE VARIATION		į		
	IOCYBERNETIC	APPLICATIONS		<u> </u>			
	DETERMINE WHETHEN TARGET HAS BEEN DETECTED	FIRE WEAPONS	EVALUATE MOMENTARY FLUCTUATIONS IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND/OR CONTROL SUBSYSTEMS	DETERMINE WHETHER GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION
				x	X		x
7						x	
						×	
1					:	l .	×
				l I			×
_						X	
			×		×	×	×
ĺ						×	
				)		×	
1						×	
						×	
				] ]		×	
-				×	×	x	

1					CATE	GORIES OF BIC	OLOGICAL SIGN	
	<del></del>		BRAIN ELECTI	RICAL ACTIVITY		,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		
	ELECT	ROENCEPHALOGR ACTIVITY	1	EVENT-RELATED POTENTIALS				
						ENDOG COMPO	GENOUS ONENTS	
	1	. '			P300	DETECTION POTENTIAL	READINESS POTENTIAL	
•			·		£	BIOCYBERNETIC	: APPLICATIONS	
PILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	ACTIVATE AND OR CONTROL SUBSYSTEMS	ADJUST DISPLAY PAHAMETERS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	1 IHE WE APONS	
VERIFY RADAR ALTITUDE OPERATION		x x			x x			
RESET ALTIMETER	х	×	×					

FIGURE 6.37 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING PRELANDING.

LOGICAL SIG	NALS					
		PERIPHERAL ACTIVITY				
ATED POTENTIALS			PSYCHO- PHYSIOLOGICAL OCULAR ACTIV RESPONSES			
ENOUS NENTS READINESS POTENTIAL	CONTINGENT NEGATIVE VARIATION		EYE MOVEMENTS	EY POSI		
APPLICATIONS						
PIHE WEAPONS	EVALUATE MOMENTARY FLUCTUATION, 3 IN ATTENTIVENESS	ASSESS GENERAL PHYSICAL STATUS	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND/OR CONTROL SUBSYSTEMS	DETERMINE WHETHER GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION	
					x x	
		_ ×	X	x		
	ENOUS NENTS READINESS POTENTIAL	ENOUS NENTS  CONTINGENT NEGATIVE VARIATION  APPLICATIONS  CONTINGENT NEGATIVE VARIATION	PSYCHO-PHYSIOLOGICAL RESPONSES  ENOUS NENTS  CONTINGENT NEGATIVE VARIATION  APPLICATIONS  PSYCHO-PHYSIOLOGICAL RESPONSES  CONTINGENT NEGATIVE VARIATION	PERIPHERAL PSYCHO- PHYSIOLOGICAL RESPONSES  ENOUS NENTS CONTINGENT NEGATIVE POTENTIAL  APPLICATIONS  ASSESS GENERAL PHYSICAL  ASSESS GENERAL PHYSICAL  ASSESS GENERAL PHYSICAL  ASSESS GENERAL COGNITIVE STATUS	PERIPHERAL ACTIVITY PSYCHO- PHYSIOLOGICAL RESPONSES  ENOUS NENTS CONTINGENT READINESS POTENTIAL  APPLICATIONS  ASSESS GENERAL COGMITIVE STATUS  ACTIVATE AND/OR CONTROL  SUBSYSTEMS  ACTIVATE AND/OR CONTROL  ASSESS GENERAL COGMITIVE STATUS  ACTIVATE AND/OR CONTROL  ACTIVATE AND/OR CONTROL  AND/OR COMITIVE STATUS  ACTIVATE AND/OR CONTROL  AND/OR COMITIVE STATUS  ACTIVATE AND/OR CONTROL  AND/OR COMITIVE STATUS  ACTIVATE AND/OR CONTROL  AND/OR COMITIVE STATUS  ACTIVATE AND/OR CONTROL  AND/OR COMITIVE STATUS  A	

	CATEGORIES O						
	BRAIN ELECTRICAL ACTIVITY						
	E L E C	TROENCEPHALOGE ACTIVITY	RAPHIC		EVE	EVENT RELATED	
				EXOGENOUS COMPONENTS		8	
					P300	DETE <b>CT</b> POTE <b>NT</b>	
						BIOCYBERI	
PILOT TASKS	ASSESS GENERAL COGNITIVE STATUS	EVALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPHOPRIATE DECISIONS	AC IVATE AND OR CONTROL SUISYSTEMS	AIJJUST DISPLAY PAHAMETEHS	E , ALUATE MOMENTARY ABILITY TO PROCESS INFORMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	
COMMAND THRUST FOR BOLTER	`		×				
MONITOR MEATBALL  CROSS CHECK STEERING COMMANDS  ENGAGE COUPLER	×	×	x		x		
CHECK STANDBY ATTITUDE DIRECTOR INDICATOR NEEDLES	ì	×			×		
CROSS CHECK DATALINK/INSTRUMENT LANDING SYSTEM DATA	<del> </del>	×	·		×	<b> </b>	
CHECK APPROACH POWER COMPENSATOR ENGAGED	×	×			x		
ENGAGE APPROACH POWER COMPENSATOR	l .		x				
OPTIMIZE DISPLAY				x	<del></del>	ļ	
SELECT MINIMUM RADAR MAP RANGE		<b>x</b>	×				
VERIFY FLAPS POSITONED DOWN	ł	×					
POSITION FLAP SWITCH FULL	ŀ		×				
VERIFY LANDING GEAR DOWN		×	x				
FOSTTION LANDING GEAR DOWN							

FIGURE 6.38 BIOCYBERNETIC APPLICATIONS AS A FUNCTION OF PILOT TASKS DURING LANDING.

CATEG	CATEGORIES OF BIOLOGICAL SIGNALS								
			·	PERIPHERAL ACTIVITY					
EVENT RELATED POTENTIALS			PSYCHO- PHYSIOLOGICAL RESPONSES	OCULAR ACTIVITY					
ENDOGENOUS COMPONENTS . CONTINGENT				EYE MOVEMENTS		EYE POSITION			
P300	DETECTION POTENTIAL	READINESS POTENTIAL	NEGATIVE VARIATION						
В	BIOCYBERNETIC	APPLICATIONS	L			L			
E , ALUATE MOMENTARY ABILITY TO PROCESS INFOHMATION AND MAKE APPROPRIATE DECISIONS	DETERMINE WHETHER TARGET HAS BEEN DETECTED	FIHE WE APONS	VALUATE MOMENTARY FLUCTUATIONS IN ATTENTIVENESS	SSESS GENERAL PHYSICAL	ASSESS GENERAL COGNITIVE STATUS	ACTIVATE AND:OR CONTROL SUBSYSTEMS	DETERMINE WHETHER GAZE HAS BEEN DIRECTED TO APPROPRIATE LOCATION		
				x x		х	×		
×				x x	×	×			
×				X	×		×		
×			×	×	x x		×		
X				×	^	×	^		
				) ^		×			
				×					
				×	×	×			
			×	×	×		×		
				×		x			
				×	x		×		
				×		×			

### 7.0 CONCLUSIONS AND RECOMMENDATIONS

In proposing to add a communication channel from the pilot to the central computer, we have shown that electrophysiological measures are relatable to pilot status and, ultimately, to performance. Further, we have made the distinction between the long term changes in cognitive function that occur across the mission, and the more transient aspects of information processing and decision-making which are associated with specific pilot tasks. As McCallum (in press) notes, however, it has not been determined whether a single physiological measure can serve as a reliable index of operator status, or, rather, if several response systems must be monitored to reveal operator states which may threaten performance. This issue of single vs. multiple inputs also must be resolved in deciding upon the most effective means of augmenting conventional methods of control activation.

We did not rank the various mission segments when suggesting biocybernetic applications in the last subsection. A ranking could have been made dependent upon the relative importance of each mission segment, either with respect to the overall mission objectives or with regard to survivability. Clearly, if we were forced to be selective (perhaps due to cost/benefit considerations) in applying biocybernetic techniques, we would choose those mission segments or requirements for which the pilot is especially vulnerable and both the accuracy and speed of performance are critical. In reviewing the in-flight mission requirements presented in Section 4, we find that the following, which occur during air-to-air or air-to-ground engagements, place the greatest demands on the pilot:

- o penetration
- o threat warning

- o detection
- o location
- o identification
- o decision
- o execution
- o assessment

The principal tasks which must be accomplished within these mission requirements may collectively be termed fire control functions. That is, we are concerned with the pilot's ability to establish correct range, azimuth, elevation, and/or FOV coordinates for the sensors used during the target acquisition process. Moreover, we are concerned with his ability to interpret multisensor imagery and to select laser/EO designators and, if necessary, countermeasures. Then, if we assume that appropriate range and/or velocity considerations have been taken into account in choosing the weapon, that arming has been accomplished prior to the engagement, and that the actual time of release is computed automatically, we are concerned with the pilot's ability to maneuver the aircraft so that the target is positioned within the missile launch or gun envelope. Superimposed upon these responsibilities is a general accountability for flight control, navigation, communications, subsystems monitoring, and, especially, threat detection/evasion.

As the pilot participates in fire control functions not managed directly by the computer, our principal information needs are related to momentary fluctuations in the pilot's capacity to process information and reach decisions. If real-time measures of pilot status are available, computer graphic techniques can be called upon to create pictorial and symbolic displays with sufficient detail to lead the pilot through changing tactical situations and

**1 OCTOBER 1979** 

weapons procedures. These decision aids will enable the pilot to adopt strategies for action which are in accordance with the current posture of the mission.

Most of the electrophysiological studies we cited earlier took place in university laboratories, and the task demands imposed upon the subjects were somewhat constrained from an operational point of view. It is apparent that the progression of a program which considers electrophysiological signals as input to adaptive military systems will require that investigations be extended to appropriate part-task and full-mission simulations. Thus, we presently are conducting part-task flight simulations in which we have challenged the pilot's ability to make multiple decisions within short periods of time. Our intent in these investigations is to further define the features of brain electrical activity and eye behavior which, when analyzed on-line during the simulation, may forewarm imminent deteriorations in pilot performance.

Finally, we should note that biocybernetic applications are, in fact, being addressed by the military laboratories (cf. 0'Donnell and Hartman, in press). Reising (in press), of the Air Force Flight Dynamics Laboratory, has predicted that techniques which yield pilot status information will be an essential component of adaptive crew stations in the future. Moreover, pattern analysis of "thought"-related EEG activity for control purposes has been discussed recently by scientists from the Air Force Aerospace Medical Research Laboratory (Aviation Week and Space Technology, Vol. 110, Number 5, 1979, pp. 239-243). It would seem that the high-risk program sponsored by DARPA over the past several years may significantly influence R&D activities within the military and aerospace industries.

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#### **1 OCTOBER 1979**

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